BOEING VERTOL CO PHILADELPHIA PA
INTERACTIONAL AERODYNAMICS OF THE SINGLE ROTOR HELICOPTER CONFI--ETC(U)
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DAAJ02-77-C-0020 AD-A062 639 UNCLASSIFIED USARTL-TR-78-23G-V-7A NL 1 OF 3 062639 111 C · Ninglin USARTL-TR-78-23G





INTERACTIONAL AERODYNAMICS OF THE SINGLE ROTOR HELICOPTER CONFIGURATION

VOLUME VII-A - Frequency Analyses of Wake Split-Film Data,
Buildup to Baseline

Philip F. Sheridan

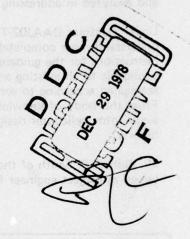
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September 1978

Final Report for Period March 1977 - February 1978

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Prepared for

APPLIED TECHNOLOGY LABORATORY

U. S. ARMY RESEARCH AND TECHNOLOGY LABORATORIES (AVRADCOM)
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78 12 28 031

APPLIED TECHNOLOGY LABORATORY POSITION STATEMENT

In 1975 a wind tunnel test program was conducted in the Boeing-Vertol 20-foot V/STOL Wind Tunnel on a 1/5th-scale UTTAS model to investigate and find solutions for several aerodynamic problems encountered during the UTTAS flight-testing. Specifically, these tests focused upon (a) the structure of the hub/rotor wake in the vicinity of the empennage, (b) the formulation of the ground vortex and its relation to hub loads and fuselage loads during transition, and (c) the occurrence of vibratory air pressures from the blade passing over the fuselage. Only portions of the above-mentioned wind tunnel test data were reduced and analyzed in addressing the flight-test problems of the UTTAS aircraft.

Under Contract DAAJ02-77-C-0020, Boeing-Vertol completed analyses on the data to understand more completely the aerodynamic interactions that are involved and to formulate instructions for the guidance of designers in these respects. The results of these studies are applicable to all existing and future single-rotor/tail rotor helicopters. The data have been segregated according to aerodynamic interactions and associated phenomena/problem areas. From this body of knowledge, a generalized set of design guidelines meaningful to the single-rotor helicopter design concept formulation were developed and are included in these reports.

Mr. Robert P. Smith of the Aeronautical Technology Division, Aeromechanics Technical Area, served as project engineer for this effort.

DISCLAIMERS

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

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SI III	INGLE ROTOR HELICO	TIONAL AERODYNAMICS OF THE TER CONFIGURATION.	FINAL REPORT
Vo	lume VIIA Frequenc	y Analyses of Wake Split- Buildup to Baseline.	15 Mar 1077 - 13 Feb 2078
Z. AUTI	HOR(a)	C - HOUZ 640	8. CONTRACT OR GRANT NUMBER(*)
Pr	nilip F./Shertdan	7 (1	DAAJØ2-77-C-0020
	rorming organization noting Vertol Compar		10. PROGRAM ELEMENT, PROJECT, TASK
P.	0. Box 16858 miladelphia, Pa.		62209A 1L262209AH76
11. CON	TROLLING OFFICE NAME A	ND ADDRESS	IN REPORT DATE
		aboratory, US Army Research poratories (AVRADCOM)	16. human of shoot 210
14. MOI	HITORING AGENCY NAME &	ADDRESS(II different from Controlling Office)	
		(12)2120	Unalasatetad
		VIII.	Unclassified 15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DIS	TRIBUTION STATEMENT (of	this Report)	
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	TRIBUTION STATEMENT (of	the abetract entered in Block 20, if different h	rom Report)
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D - Open Hubcaps

E - Air Ejectors

F - Air Ejectors With Hubcaps; Wings

G - Fairings and Surface Devices

Volume VIII, Frequency Analyses of Wake Single Film Data

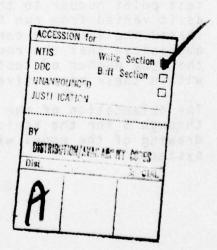
A - Buildup to Baseline

B - Basic Configuration Wake Exploration

C - Hubcaps and Air Ejectors

TABLE OF CONTENTS

ali I ogniti i delle e di mangone e di e e e e e e e e e e e e e e e e e						PAGE
INTRODUCTION		•	·		•	6
OUTLINE OF WAKE INVESTIGATIONS (TABLE 1).	•	7 10				7
LIST OF TEST RUNS (TABLE 2)	•	•		-11 -11		11
INDEX TO RAKE POSITIONS (TABLE 3)	•		•	•		18
RAKE ORIENTATION DIAGRAM (FIGURE 1)	•	•		•	•	24
HOT FILM RAKE LOCATIONS (FIGURE 2-6)		•		•	•	25
UTTAS 1/4.85 - SCALE MODEL GEOMETRY AND PRESSURE TRANSDUCER LOCATIONS (FIGURE 7))	•		•		30
SPLIT-FILM SPECTROGRAMS OF WAKE						31



INTRODUCTION

Volume VII presents an array of machine plotted graphs of wake angle and velocity versus frequency in the band from 4 to 240 Hz derived from the split film transducers. This encompasses data in the spectrum through 10 times rotor speed which is 1433 RPM or 23.88 Hz.

The graphs showing wake frequency spectra are sequenced in the same order as the Outline of Wake Investigations (Table I). These graphs are dustributed among Volumes VII-A through VII-G by the major categories of Table I in the following arrangement:

Volume $_{
m VII-A}$ - Build-up to Baseline Volume $_{
m VII-B}$ - Basic Configuration Volume $_{
m VII-C}$ - Effect of Hub Caps Sect

Volume VII-C - Effect of Hub Caps Sections 1 & 2 Volume VII-D - Effect of Hub Caps Sections 3 & 4 Volume VII-E - Effect of Hub Caps Section 5 and

Volume VII-F - Effect of Air Ejectors

Air Ejectors with Open Hub Caps and
Effect of Wings and Misc. Section 1

Volume VII-G - Effect of Wings and Misc. Sections 2 & 3

The Table I outline and other material is included for reference and as context to the work of each sub-volume. Table 2, the List of Test Runs, arranges the runs in numerical order and gives pertinent text parameters.

The Index of Rake Positions, Table 3, lists the hot film transducer rake positions in the model coordinate system for each run and its test points. The main feature of Table 3 is the indexing of the test point number to the model water line station and butt line as it varied from run to run. The table groups the runs as they shared the indexing correspondence of point with position. It is emphasized that the runs in a group do not necessarily all share the same number of test points but they do have same correspondence within their respective ranges of test points.

The orientation of the rake is shown pictorially in Figures 1 through 6 for the various test runs. Figure 7 presents a scaled drawing of the model with reference to the three-axis coordinate system.

	TABLE 1	a Maki		
	OUTLINE OF WAKE IN	VESTIGATIONS		
4.	Description	Configuration Code		Base- line
Buil	ld-up to Baseline	1 00000 21	20.00	af hadis
1.	Nacelles removed	K ₁₃ +H ₁ -N	149	150
2.	Blades off, rotating hub	K ₁₃ -M+H _{1.0}	160	156
3.	" , non-rotating hub	K ₁₃ -M+H _{1.0}	158	156
4.	" , hub off	K ₁₃ -M-H _{1.0}	159	156
Basi	c Configuration			
1. W	Make Explorations near Empennage	Specific Section		
((a) 15" Long. + traverse at T/R C.L.	K ₁₁	111	
	(b) 9" Vert. + " above T/R "		112	
1	d) 8" " (continue 112)		113	
((e) 13" " behind stab.	"	115	
(f) Lateral traverse, left stab. (One T.P. only)		116	
	g) Same continued	"	117	
	(h) Same continued (One T.P. only) (i) Lateral traverse right stab.	"	118	
ì	j) T/R effect on wake	$K_{11}+T_2^0$	121	115
2. 0	Climb/Descent Studies	ose Mest. I conte l		
((a) Climb 900 FPM	K _{1 1}	135	
((b) Descent 800 FPM		136	
Eff	ect Of Hub Caps	Ale		
1	. Solid Caps on Canister			
	(a) 7.6" diam. 2.17" ht. soft Pitch Arms	K ₁₁ -H _{1.0} +H _{1.2}	137	136
		K ₁₃ +H _{1.2}	153	156
	(b) 7.6" diam. 2.45" ht. flt. test config.	K ₁₃ +H _{1.2.1} +I ₁ +E _{1.0}	207	188

TABLE 1 (CONTINUED)

OUTLINE OF WAKE INVESTIGATIONS

Description	Configuration Code*	Run No.	Base- line
Effect of Hub Caps (Continued)	900 900		2002.1
2. Solid Caps Raised Above Canister	2-955E/18		200
(a) 7.6" diam. 2.45" ht. 70" depth, .55 gap	H _{1.2.2} +I ₁ +E _{1.0}	208	188
(b) 10.0" diam. 3.25" ht. 1.55" depth, .50" gap	H _{1.8.1} +I ₁ +E _{1.0}	189	188
(c) 10.0" diam. 4.125" ht. 2.05" depth, .875" gap	H _{1.8.2} +I ₁ +E _{1.0}	190	188
(d) Repeat of 189		210	188
3. Open Caps Without Underbody	500,000		
(a) 10.0" diam. 1.25" gap,blades (b) " " gap, no blades	H _{1.0.1} +I ₁ +E _{1.0}	193 166	188/166 158
(c) " 2.05" gap,blades	H _{1.14.1} +I ₁ +E _{1.0} H _{1.0.1} -M	211 165	188 158
(e) " " 1.87" gap,blades (f) 16" diam. 2.00" gap,blades	H _{1.0.3} +I ₁ +E _{1.0}	191	188 156/167
(g) " " gap, no blades	H ₁ .7.1-M	167	158
	H _{1.7.2}	169	156
4. Open Caps with Underbody	expv no tos	to s	2.17
(a) 7.6" diam. 1.25" gap	H1.11.1+I2+E1.0	194	188
(b) " " " " center	H _{1.11.1} +I ₂ +E _{4.0} H _{1.11.2} +I ₂	198	188
post	11.11.2+12	202	194
(d) 10.0" diam5" gap, no blades	H _{1.5.1} -M	164	158
(e) " " 1.25" gap, no blades	H _{1.5.2} -M	161	158
	H _{1.5.4} -M	163	158
	H _{1.5.3} -M	162	158
	H _{1.5.2}	154	156/161

TABLE 1 (CONTINUED)

OUTLINE OF WAKE INVESTIGATIONS

Description	Configuration Code*	Run No.	Base- line
5. Miscellaneous Hub Covers (a) Hub fairing 16" diam. (b) Wham-O-Frisbee 10" diam. (c) Fab. glass Frisbee 16" diam.	H _{1.3} H _{1.9.0} +E _{1.2} H _{1.9.1} +E _{1.2}	151 182 183	150 181 181
Effect of Air Ejectors	ey Trugga Wille G Pari (gayangan) xikil 1	E CES.	£ (0)
1. Basic system no blowing 2. " " 40 psi 3. " " 150 psi 4. Wide chord shroud 40 psi 5. Wide " " 150 psi 6. W/C shroud w. lip 40 psi 7. Same Contoured Parallel 150 psi 8. Bifurcated duct 0 psi 9. " 40 psi 10. " 150 psi	H _{1.0} +E _{1.0} H _{1.0} +E _{2.5.1} H _{1.0} +E _{3.5.2} H _{1.0} +E _{3.5.4} 1.0+E _{3.5}	174 175 176 184 187 203	156 156/172 156/173 156/174 156/173 156/174 156 156/203 156/203
Air Ejectors with Open Hub Caps with Underbodies			
1. 7.6" diam. 1.25" gap, 0 psi 2. " " " 20 psi 3. " " " 40 psi 4. " " " 150 psi 5. " " " 40 psi 6. " " " 40 psi 7. " " 150 psi 8. Same with center post 9. 10.0" diam. 2.0" gap wide ch'd shroud (150 psi)	H _{1.11.1} +1 ₂ +E _{1.0} """ H _{1.11.1} +I ₂ +E _{4.0} """ H _{1.11.2} +I ₂ +E _{4.6} H _{1.5.4} +E _{2.5.1}	195 196 197 198 199 200 201	188/172 188 188/173 188/174 188/194 188/196 188/196 188/200 156/176
Effect of Wings and Misc.			
1. Wings (a) Nacelle-mounted stub wing (b) Single slotted flapped wing (c) Dougle slotted flapped wing (d) Boom-mounted stub wing	H _{1.0} +W _{1.0} +E _{1.1} H _{1.0} +W _{3.0} +E _{1.0} H _{1.0} +W _{2.0} +E _{1.0} H _{1.0} +W _{4.0}	178 180 179 186	181 181 181 156

TABLE 1 (CONTINUED)

OUTLINE OF WAKE INVESTIGATIONS

	Desc	ription	Configuration Code*	Run No.	Base- line
2.	(b) Round (c) Extended (d) Flat to	ings op behind shaft top behind shaft ed flat top fairing op + 16" cap, 4" gap d fairing/nacelle fairing	K ₁₁ +D ₁ K ₁₁ +D ₂ H ₁ +D ₄ H _{1·7·2} +D ₄ P _{1·0}	140 141 170 171 152	138 138 156 170 156
3.	(b) Guideva (c) Longita	rices generators ane between nacelles adinal strakes cosity spoiler	K ₁₁ +VG _{2.1} K ₁₁ +FV ₁ H _{1•5.3} +S ₄ K ₁₁ +X ₁	139 142 155 143	138 138 156 138
			A TOUR CHANGE AND SECURE		
				H H H	
			res * * neog perneo neow yes *5.5 m they ob	5 (10 kg (12) (13) (13)	
			20 48 DOM 094		

*Basic Code is Kl3 unless noted otherwise.

-2.0 MODEL 0 = = = = = 0.9 . = = = = DISK LDG. 8 = = = = = BASIC INVESTIGATIONS OF THE HUB WAKE RPM MR/TR 1433/0 = = = = = TABLE 2. LIST OF TEST RUNS Vrun 80 = = = Vert traverse through MR at Vert. traverse below TR CONFIGURATION/CONDITION /13" Vert. traverse behind K₁₁/15" Long. wake traverse above TR center line - left Vert. wake traverse TR center line /Lateral traverse center line stabilizer vortex /5" .8/ "6/

TAIL ROTOR

h/d

Off

8

=

=

=

=

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=

=

=

=

=

=

=

/116 continued

=

117

stabilizer

o

=

=

=

=

=

=

=

- right

" /Lateral traverse

119

121

stabilizer

/116 continued

118

1433/ 4500

K₁₁+T₂/Effect of tail rotor flow

on wake

K₁₁/Wake in 900 fpm climb

135

=

=

=

=

=

=

=

Off

=

4.5

0.9

=

=

=

=

=

-2.0

0.9

=

=

=

descent

fpm

800

" /Wake in

136

-	_	
1	1	

115

116

=

114

=

113

111

NO.

112

TABLE 2(CONTINUED) LIST OF TEST RUNS EVALUATION OF WAKE-ALTERING DEVICES

								-
RUN	CONFIGURATION/CONDITION	Vrun	RPM	DISK	MODEL	MODEL	MR HT.	TAIL
NO.		KNOTS	MR/TR	pst.	9 5	o A	p/q	ROTOR
137	Kll-Hl.0+Hl.2/Effect of 7.6 inch diam. solid hub cap	80	1433/0	8	9	-3.8	8) jjo
138	K ₁₁ /Repeat of base run			=	-		=	
139	Kll+VG2.1/Effect of vortex gener- ators on aft crown	=		E	=	=	=	
140	<pre>K₁₁+D₁/Flat-topped "doghouse" fair- ing on aft crown</pre>	н	=	=	=	=	=	
141	$\kappa_{11}^{+D_2/\text{Rounded-top}}$ fairing	=	=	=	=	=	=	•
142	K ₁₁ +FV ₁ /Deflection vane on crown between nacelles	=	=	=	=	=	=	-
143	$K_{11}^{+X_1}$ /Variable porosity spoiler			E	2		=	=
149	$K_{13}^{+}H_1^{-}N_1$ /Effect of nacelles off also acd stiff pitch arms (K_{13})	60	1075/0	4.5	=	=	=	=
150	K ₁₃ +H ₁ /60 knot baseline		•				=	Ε
151	K ₁₃ +H _{1.3} /16 inch diam. helmet fair- ing	=				•	-	=
152	K ₁₃ +P _{1.0} /Pylon and intake fairings	80	1433/0	8	=	=	=	
153	$K_{13}^{+H}_{1.2}/Repeat$ 137 with K_{13} pitch arms			=	=	=	=	=

ROTOR Off MR. HT. 8 = = = = = = = = -3.8 ANGLES = = = = = MODEL 9 = = = = = DISK LDG. psf œ = = = = = = = = EVALUATION OF WAKE-ALTERING DEVICES RPM MR/TR 1433/0 1433/0 0/0 0/0 = = = = = TABLE 2 (CONTINUED) LIST OF TEST RUNS KNOTS VTUN 80 = = = = = = = = = = 1.8 K₁₃-M-H_{1.0}/Wake studies with hub of K₁₃-M+H_{1.0}/Wake studies with blades K₁₃-M+H_{1.0.1}/10" open hub cap, no underbody, same cap vert, position as Run 154 strakes on aft crown K₁₃-M+H_{1.5.2}/Repeat of 154 without blades 4 5 K₁₃-M+H_{1.0}/Same as 158 except hub K13-M+H1.0.2/Same as 165 with cap K₁₃+H_{1.5.2}+S₄/Same as 154 except K₁₃+H_{1.0}/Baseline with K₁₃,i.e., off, hub not rotating K₁₃-M+H_{1.5.3}/Same as 161 except except K₁₃+H_{1.5.2}/10" open hub cap, 7" underbody, 1.25" gap K₁₃-M+H_{1.5.1}/Same as 161 except 0.5" gap CONFIGURATION/CONDITION lowered by 0.5" stiff pitch arms as 161 rotating K13-M+H1.5.4/Same gap 155 156 160 154 158 159 161 164 162 163 165 166 NO.

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	IN OF WAKE-ALTERING DEVICES
RUNS	RING
TEST	-ALTE
OF	VAKE
LIST	OF V
TABLE 2 (CONTINUED) LIST OF TEST RU	EVALUATION
7	-
TABLE	

						-		
RUN	CONFIGURATION/CONDITION	VTUN	RPM	DISK	MODEL	MODEL	HT.	TAIL
NO.		KNOTS	MR/TR	pst.	8	0 4	h/d	ROTOR
167	K ₁₃ -M+H _{1.7.1} /16" open cap, no underbody, 2" gap	80	0/0	æ	9	-3.8	8	Off
168	K ₁₃ +H _{1.7.1} /Blades on, same cap config. as 167	•	1433/0	=	=			
169	Kl3 ^{+H} l.7.2/16" open cap, no under- body, 4" gap	=	=	=	•		=	=
170	K ₁₃ +H _{1.0} +D _{4.0} /Extended flat top fairing on aft crown		=	=	•	2		:
171	K ₁₃ +H _{1.7.2} +D _{4.0} /Same fairing as 170, same cap as 169		=	•		=	=	:
172	Kl3+H1.0+E1.0(Opsi)/Basic air ejector zero blowing baseline	=	=	=	=			1
173	$^{\mathrm{K}_{13}}$ $^{+\mathrm{H}_{1.0}}$ $^{+\mathrm{E}_{1.0}}$ $^{(40\ \mathrm{psi})}$ /Same as 172 with 40 psi supply	=	-	=	=	=	=	
174	K ₁₃ +H _{1.0} +E _{1.0} (150 psi)/Same as 172 with 150 psi supply		=	=		•		
	Kl3+Hl 0+E2.5.1 (40 psi)/Ejector with wide chord shroud at 40 psi			=	•	=		•
9/1	(150 psi th 150 p	•	-	-		•		
177	$K_{13}^{+H_1}$. $f_{14}^{+E_2}$ $f_{11}^{+E_2}$ $f_{110}^{+E_1}$ cap like 163			=	=		:	
178	K ₁₃ +H _{1.0} +W _{1.0} +E _{1.1} (0 psi)/Nacelle mounted wing		=	=	:	=	=	

	TABLE 2 (CONTINUED) LIST OF TEST RUNS	TEST RU	NS					
	EVALUATION OF WAKE-ALTERING DEVICES	ALTERIN	G DEVICE	Ø				
	CONFIGURATION/CONDITION	VTUN	RPM	DISK	ANC	MODEL	MR HT.	TAIL
		KNOTS	MR/TR	pst.	°g	•	h/d	ROTOR
	K13+H1.0+W2.0+E1.0(0 psi)/Double slotted flapped wing	80	1433/0	8	9	-3.8	8	Off
	K ₁₃ +H _{1.0} +W _{3.0} +E _{1.0} (0 psi)/Single slotted flapped wing	•	-			•	=	=
STATE OF STATE OF	K ₁₃ +H _{1.0} +E _{1.2} (0 psi)/Baseline with ejector tube moved aft	=	=	•	=			•
	$^{\mathrm{K}_{13}+\mathrm{H}_{1.9.0}+\mathrm{E}_{1.2}}$ (0 psi)/Standard 10" frishee	=	=		=		=	=
	K ₁₃ +H _{1.9.1} +E _{1.2} (0 psi)/16" fabri- cated frisbee	•	=		5	=	=	=
	K13+H1.0+E3.5.2 (40 psi)/Wide chord with lip at 40 psi		=		=	=	=	
	K ₁₃ ^{+H} 1.0 ^{+E} 3.5.2 (150 psi)/Same as 184 with 150 psi air				=	=	=	=
	K ₁₃ +H _{1.0} +W _{4.0} /Boom mounted stub wing	=			=	=	=	-
	K ₁₃ +H _{1.0} +E _{3.5.4} (150 psi)/Like 185 with modified shroud					=	=	
Section 1	Kl3+H1.0+I1+E1.0(0 psi)/Baseline with I; instr. ring	=	=			=		=
10-1-11-1	K ₁₃ +H _{1.8.1} +I ₁ +E _{1.0} (0 psi)/Solid cap, 10" diam. 3.25" height	•			=	=	=	=
	K ₁₃ +H _{1.8.2} +I ₁ +E _{1.0} (0 psi)/Same as 190 except + 4.12" height	=	=			•	=	=

TABLE 2 (CONTINUED) LIST OF TEST RUNS

EVALUATION OF WAKE-ALTERING DEVICES

RUN	CONFIGURATION/CONDITION	VTUN	RPM	DISK	MODEL	MODEL	MR.	TAIL
NO.		KNOTS	MR/TR	pst.	8 8	0 1	h/d	ROTOR
191	Kl3+H1.0.2+I1+E1.0 (0 psi)/10" cap, no underbody, 1.87" gap	80	1433/0	8	9	-3.8	8	Off
193		=	•	=				:
194	Kl3+H1.11.1+I2+E1.0(0 psi)/7.6" cap, underbody, 1.25" gap	•	=	=	=		-	
195	K13+H1.11.1+I2+E1.0(20 psi)/Same as 194 with 20 psi air				=	=	=	=
196	K13+H1.11.1+I2+E1.0(40 psi)/Same as 194 with 40 psi air			•	•		=	
197	K13+H1.11.1+I2+E1.0(150 psi)/Same as 194 with 150 psi air		=	•	•	=	=	:
198	K ₁₃ +H ₁ .11.1+I ₂ +E _{4.0} (0 psi)/Same as 194 except blowing tube 2" aft					=		
199	K13+H1.11.1+I2+E4.0 (40 psi)/Same as 198 with 40 psi air	=						
200	K13+H1,11,1+I2+E4,0 (150 psi)/Same as 198 with 150 psi air		10 N 8 N	•			•	
201	K ₁₃ +H _{1.11.2} +I ₂ +E _{4.0} (150 psi)/Same as 200 except center support cap			=	TKOS	•	8 10 8 17	
202	K ₁₃ +H _{1.11.2} +I ₂ /Baseline with I ₂ and no blowing tube	TIA TRAK	SOUTH PORT		=			=
203	K ₁₃ +H _{1.0} +E _{5.0} (0 psi)/Bifurcated air duct baseline				=	=		

	DEVICES
RUNS	
TEST	OF WAKE-ALTERING
OF	AKE-
LIST	OF W
2(CONTINUED)	EVALUATION
TABLE	

ES HT. TAIL	v h/d ROTOR	-3.8 ° Off		100 017		88		uera Te	16 C			
MODEL	• 8	9	•	=		=			3 S S S S S S S S S S S S S S S S S S S			
	pst.	8		=	=	=	-		3.0			
RPM	MR/TR	1433/0	:	=		-	8 A A A A A A A A A A A A A A A A A A A		S		11.1	
Vrun	KNOTS	80		= *O		•	• 3					
CONFIGURATION		K13+H1.0+E5.0 duct with 15	Kl3+Hl.0+E5.0 (40 psi)/Same as 204 with 40 psi air	207 K ₁₃ +H _{1.2.1} +I ₁ +E _{1.0} (0 psi)/7.6" solid cap, no gap	Kl3+H1.2.2+I1+E1.0 (0 psi)/Same as 207 except 0.55" gap	Kl3+H1.15.1+I ₁ +E _{1.0} (0 psi)/Repeat of 189	K13+H1					
RUN	0	204	205	207	208	210	211		6.0 6.0 6.0			

TABLE 3

INDEX TO RAKE POSITIONS

RUN NUMBER	POINT	WATER LINE	STATION	BUTT	LOCATION
111	20 21	53.5	103.1	-7.25	1
	22		105.0	"	i di al
	24 26		107.0		12.8
	28		111.0		
	30	н	112.9		
	32	"	114.9	" "	No.
	36	10	118.9	5	
112	2	48.9	107.3	-7.25	1
	4	50.8	103.3		
	2 4 6 8	54.5	103.3	"	100
	10	56.2		"	1 5 9
	12	57.2			1 多位
113	2	51.7	103.3	-3.25	1
	6 8	52.3 52.8			1 1 2 2
		53.3			
	10	53.9		"	1 5 5
	11	53.3	L "		
114	2	44.5	103.0	-3.25	1
	6	46.4			
	8	50.0			
	10	51.9			
115	3	52.9 52.0	124.7	-3.25	1
	4	52.0	"		
	3 4 6 9 10 12	48.0			
	10	46.0			
	12	44.1			
	14	42.1 53.0			
	18	54.0			
	20	55.0			

TABLE 3 (CONTINUED) INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER	MODEL STATION	BUTT	LOCATION FIGURE
116	7	36.9	100.5	-17.5	1
117	2 4 6 8 10	37.6 37.3	100.5	-16.0 -14.0 -12.0 -10.0 - 8.0	1
118	2	37.6	100.5	- 6.0	1
119	2 5 8 9 14 16 20 25	37.3 "" "" 51.5 52.3	99.6 "" "" 102.5 101.7	+ 6.0 8 10 14 16 17.5 -17.5	1
121	3 4 6 8 10	62.9 53.5 50.1 46.0 42.1	129.0	+ 5.7	2
135	2 4 6 8 10 12 14	56.9 54.5 52.5 50.5 48.5 46.5 44.5	106.3	- 5.7	3
136	2 4 6 8 10 12 14 17 18 19	56.5 54.5 52.5 50.6 48.5 46.5 44.5 37.1 39.0 41.0	104.0	- 8.0	4

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST	WATER LINE	MODEL STATION	BUTT	LOCATION FIGURE
137	3	38.7 39.9	98.4	- 8.0	5
1	5 7 9 11	42.0 44.0 46.0	100.5	" "	
	13 15 17	48.0 50.0 52.0	"		
	19	54.0	• . 16	"	813
138-41, 143	2 3 4 5 6 7 8 9	38.8 40.0 42.0 44.0 46.0	98.4	- 8.0	5
•	7 8 9	48.0 50.0 52.0 54.0	103.6	n n n	
142	7 8 9	37.8 40.2 42.0	98.4	- 8.0	5
.E	11 12 13 14 15 16	44.0 46.0 48.0 50.0 52.0 54.0 56.8	103.6	# # # # # # # # # # # # # # # # # # #	301
3 3	O _ Q - ~ N	u-80	6.35 6.36 8.36 8.46 6.06	1 2 0	236.
			2.96 2.70 3.44 2.73 3.75 3.90 3.00 3.00 3.00	2.5 2.7 2.4 2.5 2.5 2.5	

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER	MODEL STATION	BUTT	LOCATION FIGURE
149-151	2 3 4 5 6 7 8	38.8 40.0 42.0 44.0 46.0 48.0 50.0	98.5 100.6 103.5	- 8.0	5
	10	52.0 54.0		•	2.13
152-6, 158 161-4, 166 167, 169-71 175, 177-9 180,182,184 186-8, 190	2 3 4 5 6 7	42.9 44.9 46.9 48.9 50.9 52.9	97.9 100.6 104.6	0.0	6
191,193,194 196,198,201 204,207,208 211	8	54.9 56.9	9 H 38 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	•	173,174,174 195,200,200 196,200,200
159	1 2 3 4 5	54.9 52.9 50.7 48.6 46.7	104.6	0.0	6
160,203	5 6 7 8 9 10	42.9 44.9 46.9 48.9 50.9 52.9 54.9	97.9 100.6 104.6	0.0	6
165	3 4 5 6 7 8	44.9 42.9 46.9 48.9 50.9 52.9	97.9 100.6 104.6	0.0	6

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT	LOCATION FIGURE
168, 183	4	42.9	97.9	0.0	6
	5	44.9	200	"	1000
	5 6 7 8 9	46.9	100.6	"	
	7	48.9	3°5	"	
	8	50.9	104.6	"	
	9	52.9	"	"	
	10	54.9	7	".	
172	3	42.9	97.9	0.0	6
	4	44.9	0.05		
	6	44.9		"	
	7	46.9	100.6		1000
	6 7 8 9	48.9		"	
	9	50.9	104.6	"	
	10 11	52.9	11	"	101
	11	54.9			
173,174,176	1	42.9	97.9	0.0	6
185,195,197	2	44.9		"	A VALUE OF A
199,200,205	3	46.9	100.6		STATE AND STATE
210	4	48.9	104.6		
	5	50.9	104.0		0.85
	1 2 3 4 5 6 7	54.9	"		
181	2	42.9	97.9	0.0	6
	3	44.9	46.43	"	
	2 3 4 5 6 7	46.9	100.6		
	5	48.9	4219		805.08
	6	50.9	104.6	•	
	7	52.9	F. 6.6	•	
	9	54.9	8 1.85	" 6	
	10	1 3	8 "06	" e	
	11		1 2 2	"01	
	12		"	11	
	13	42.9	97.9		
			5.52		
		1.00 0.01	8 24		
			6.08		
			8213		

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT	LOCATION FIGURE
189	29 30 31 32 33 34 35 36 37 38 39	42.9 44.9 46.9 48.9 50.9 50.9 52.9 54.9	97.9 100.6 " 104.6 100.6 104.6	0.0	6
202	3 4 5 6 7	43.4 44.9 46.9 48.9 50.9	97.9 100.6 104.6	0.0	6

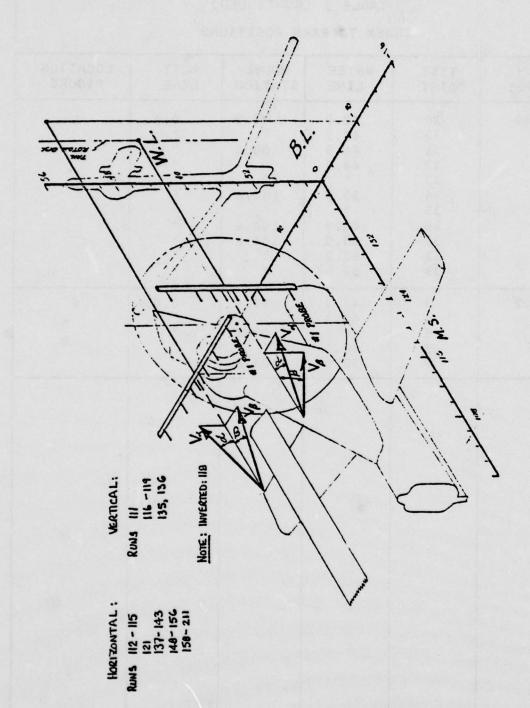


FIGURE 1 - RAKE ORIENTATION DIAGRAM

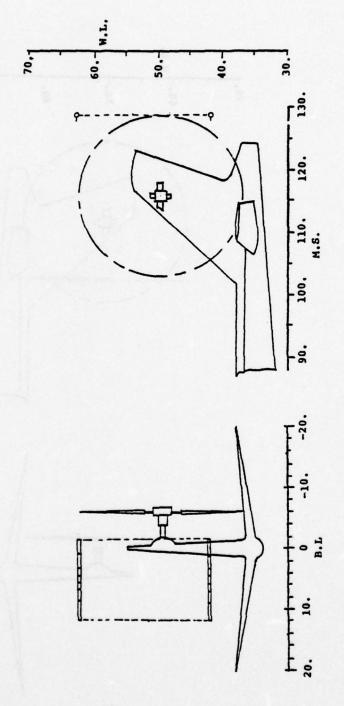


FIGURE 2 -HOT FILM RAKE LOCATIONS

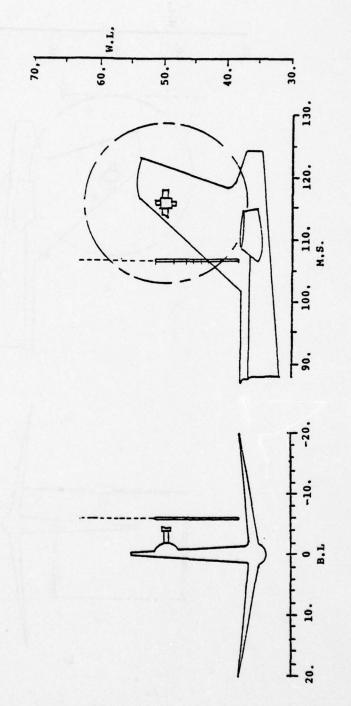


FIGURE 3 -HOT FILM RAKE LOCATIONS

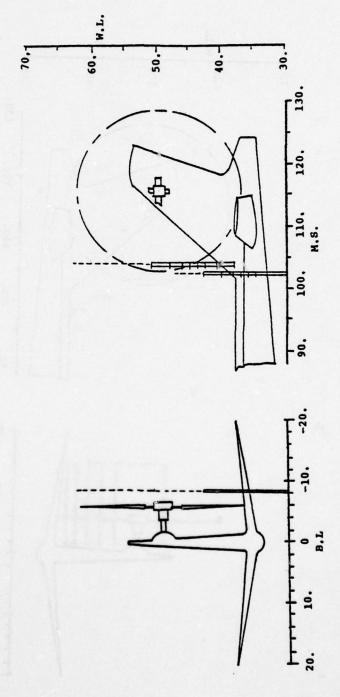


FIGURE 4 -HOT FILM RAKE LOCATIONS

RUN 137, 138, 139, 140, 141, 142, 143, 148, 149, 150, 151

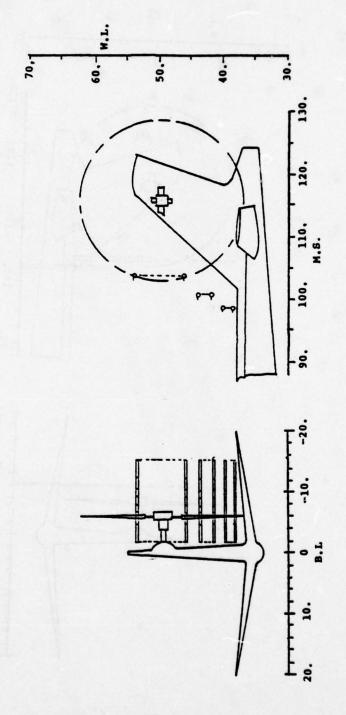


FIGURE 5 -HOT FILM RAKE LOCATIONS

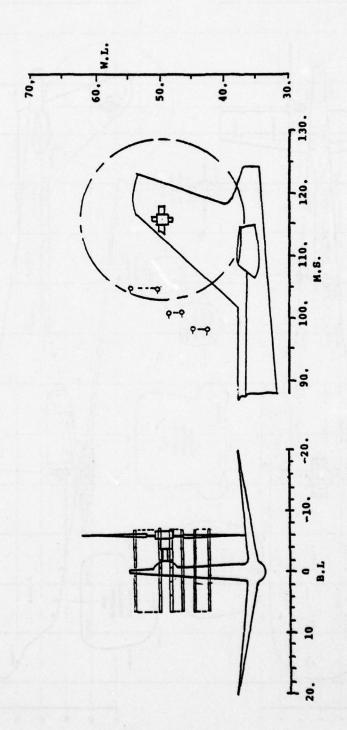
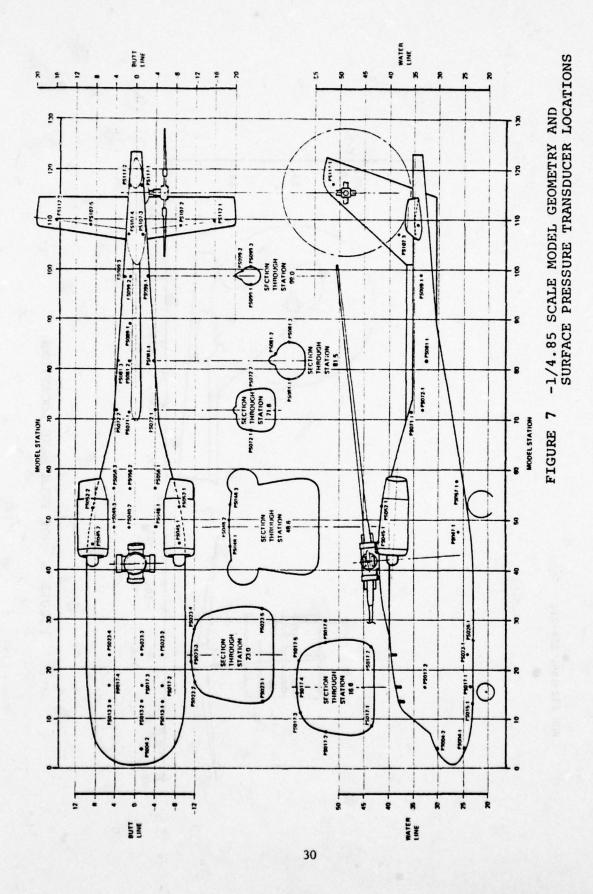
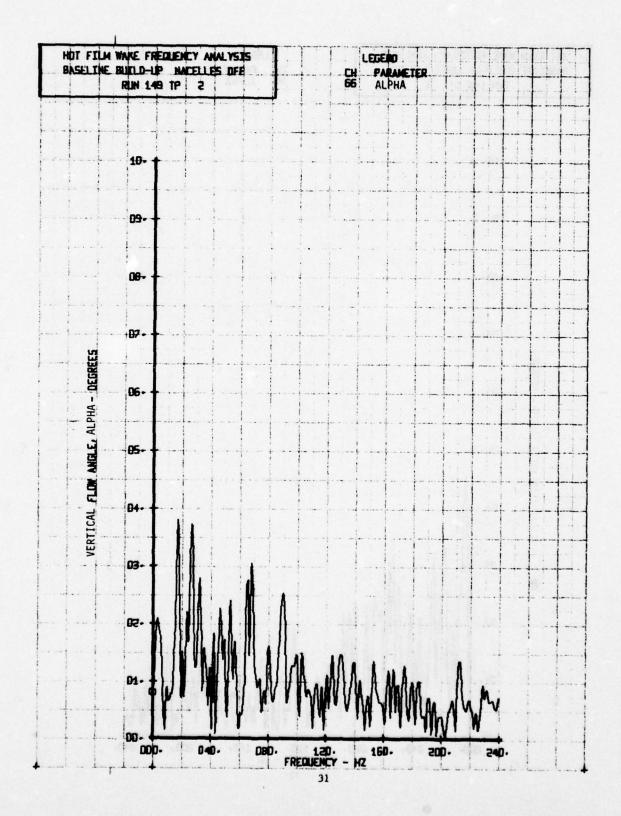
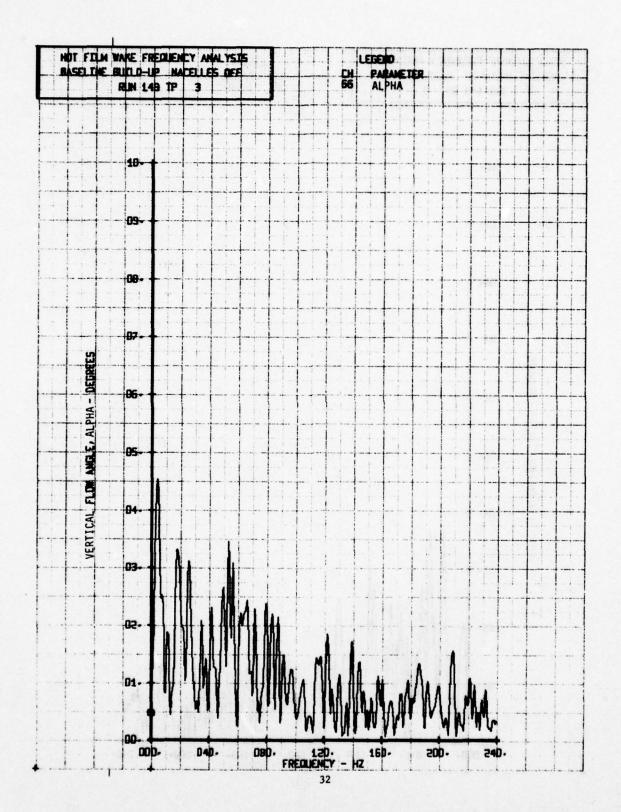
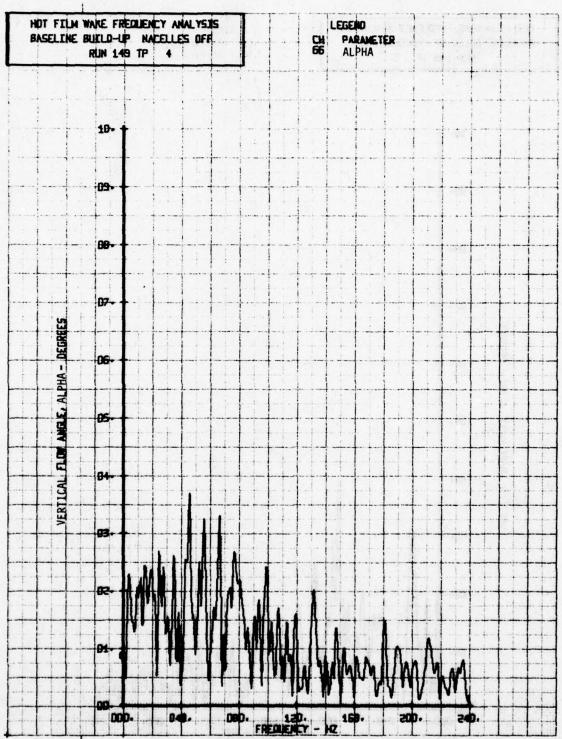


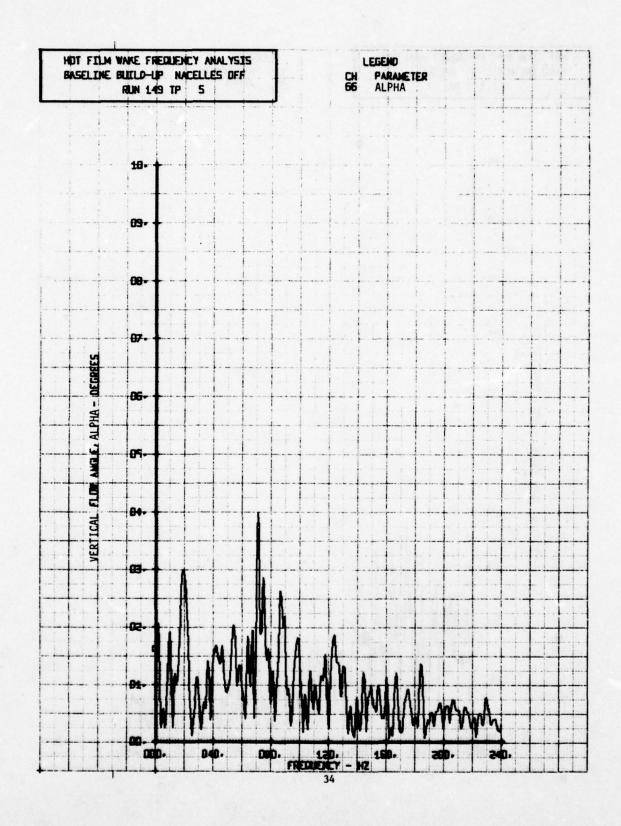
FIGURE 6 -HOT FILM RAKE LOCATIONS

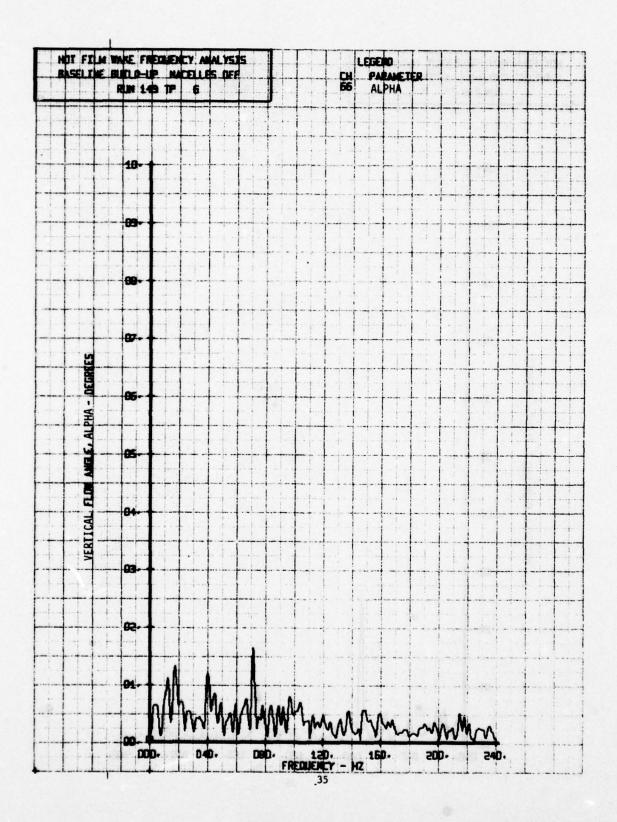


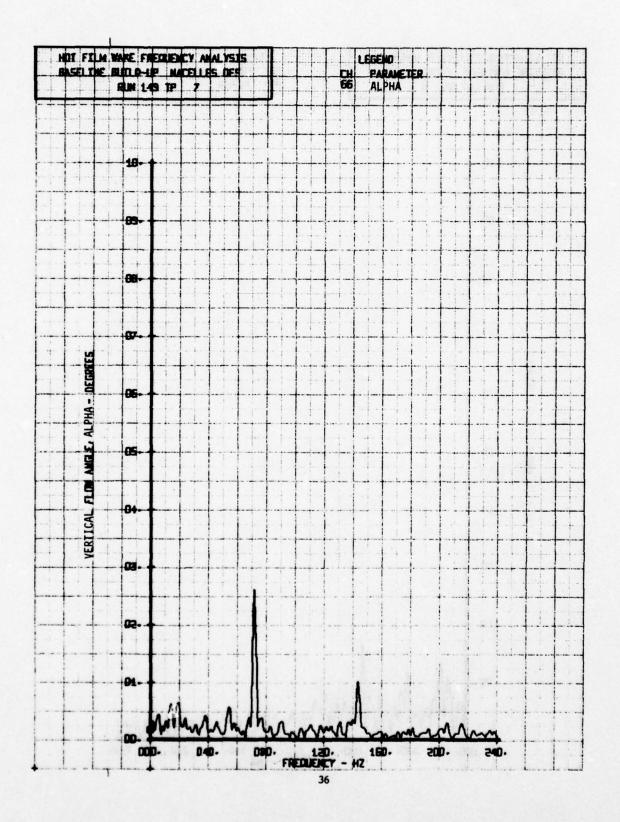


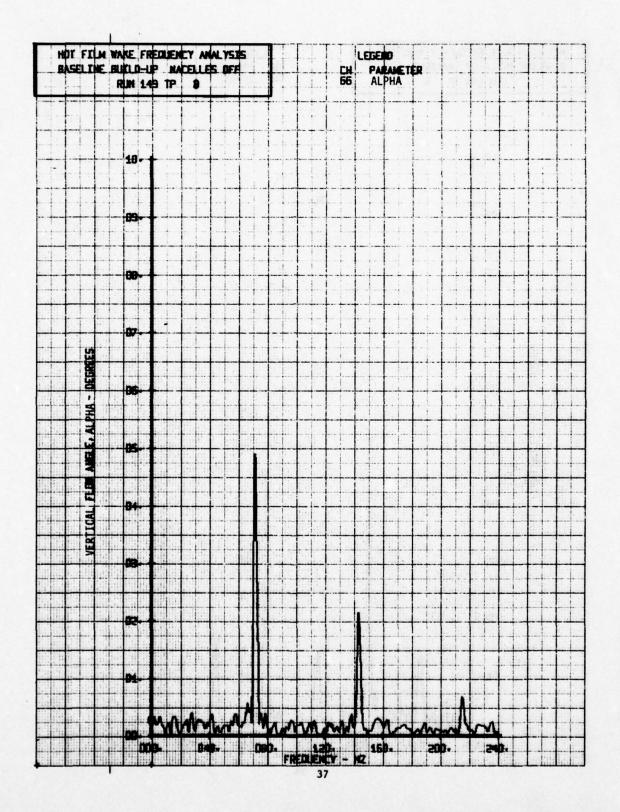


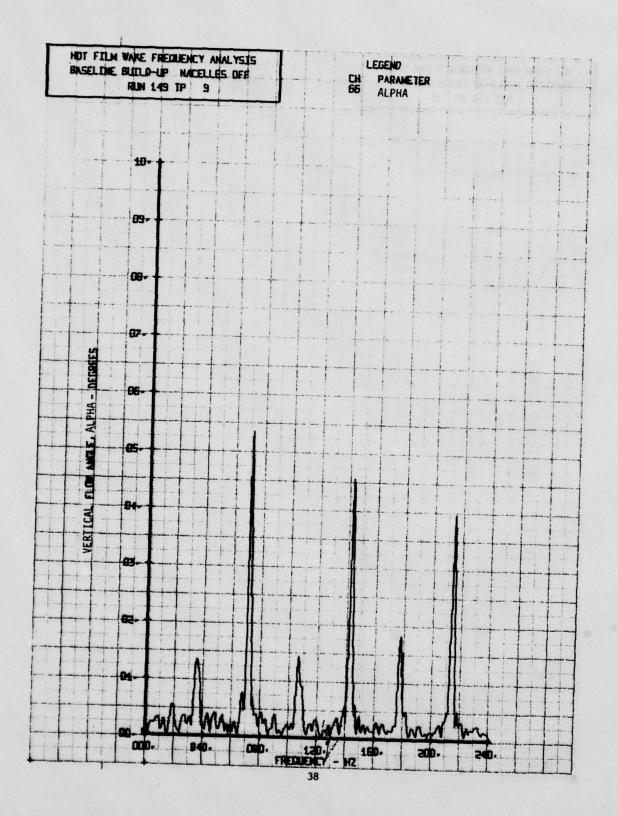


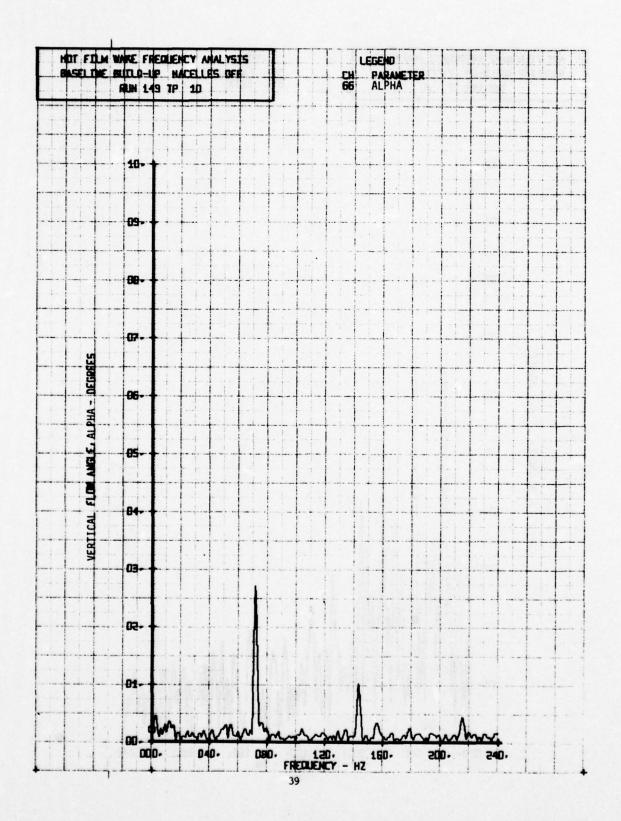


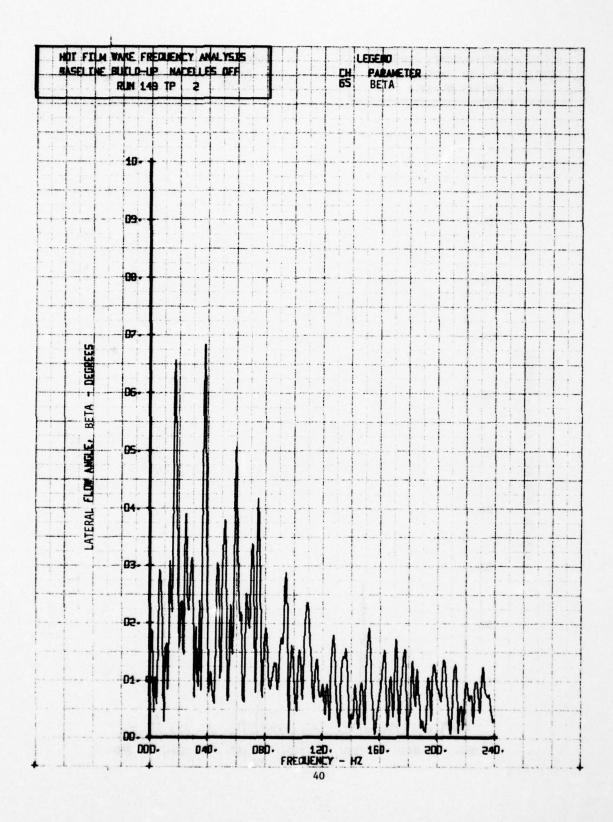


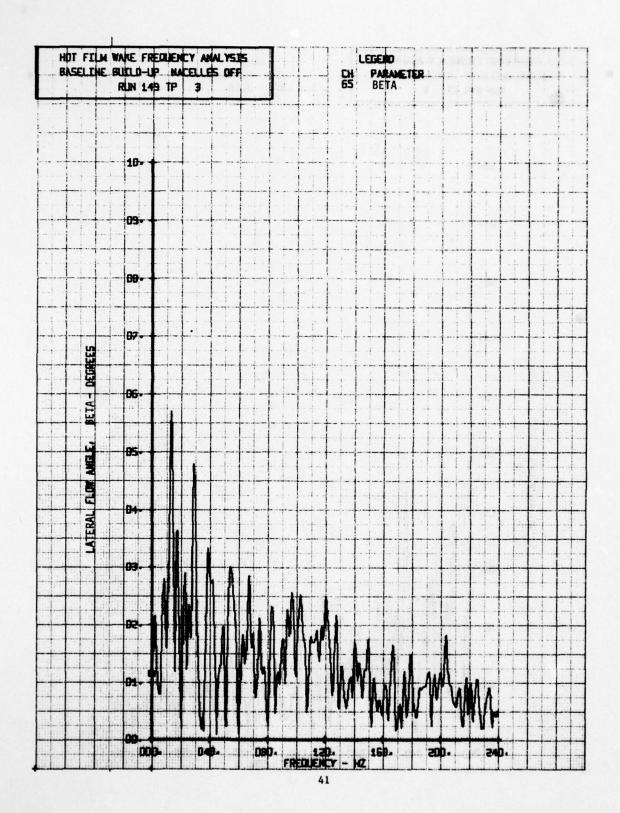


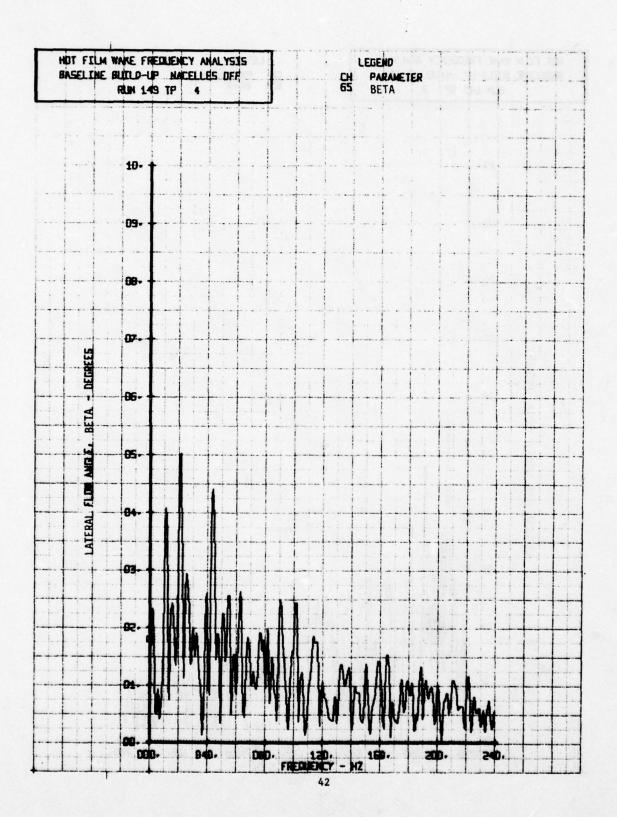


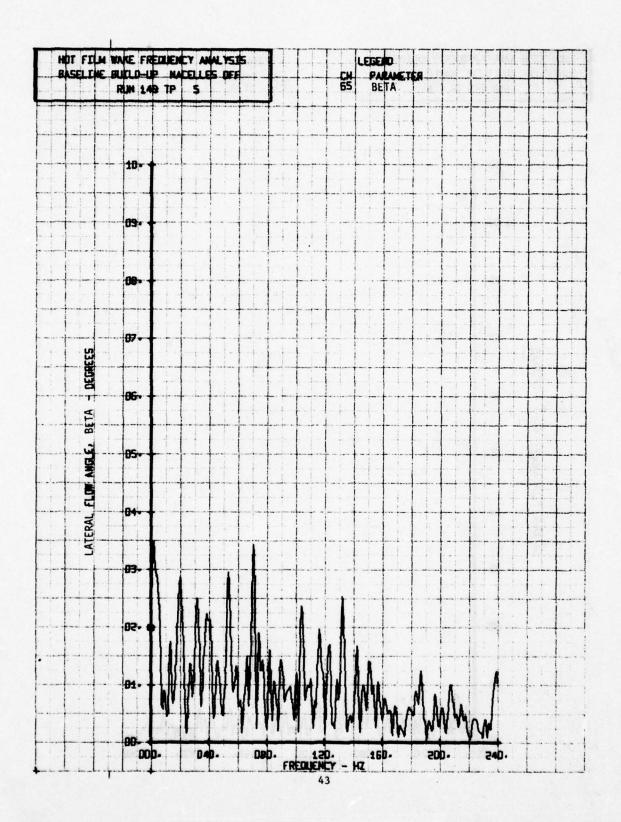


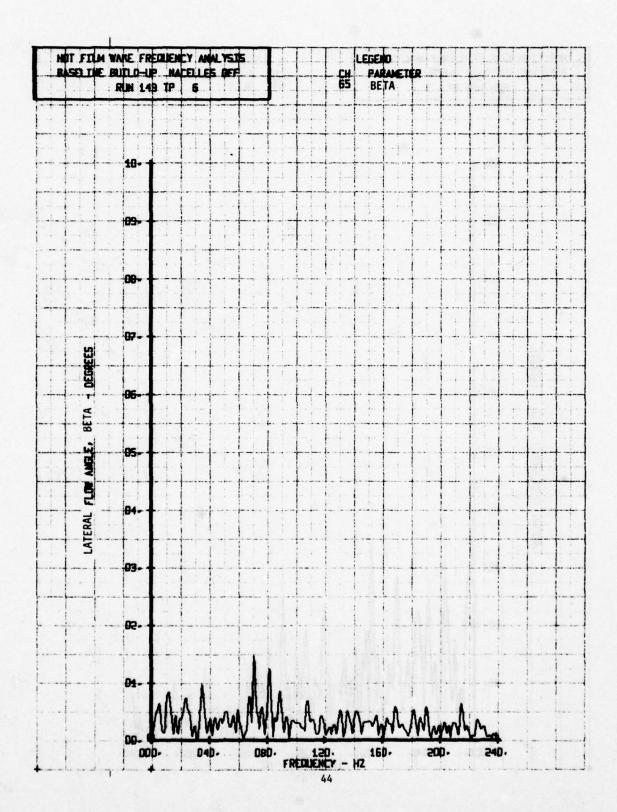


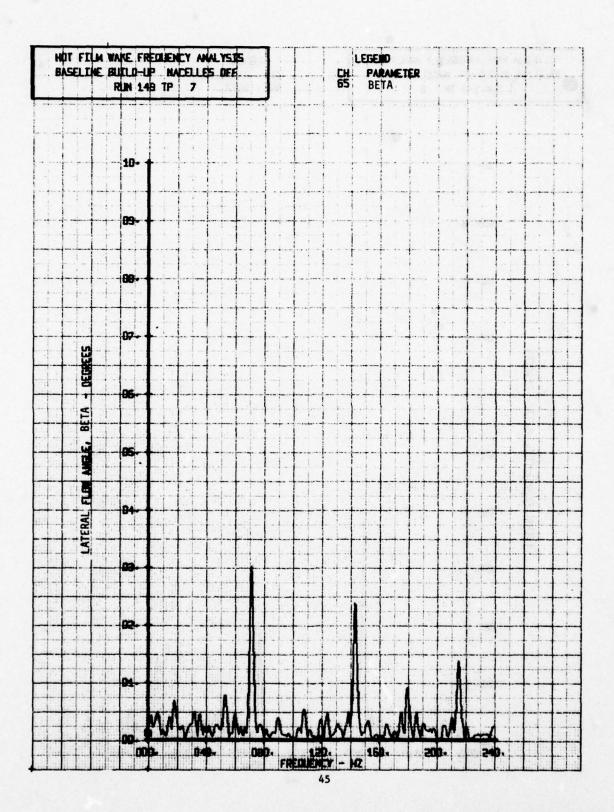


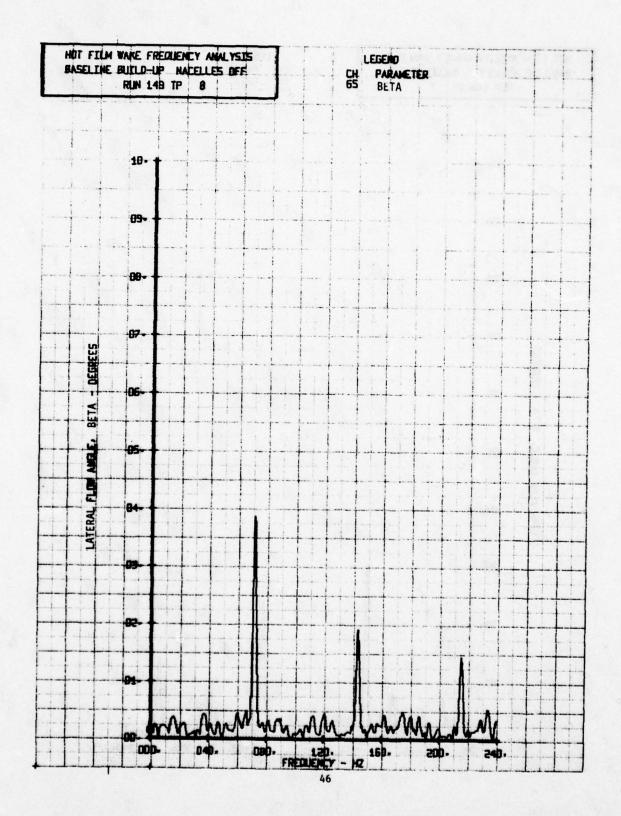


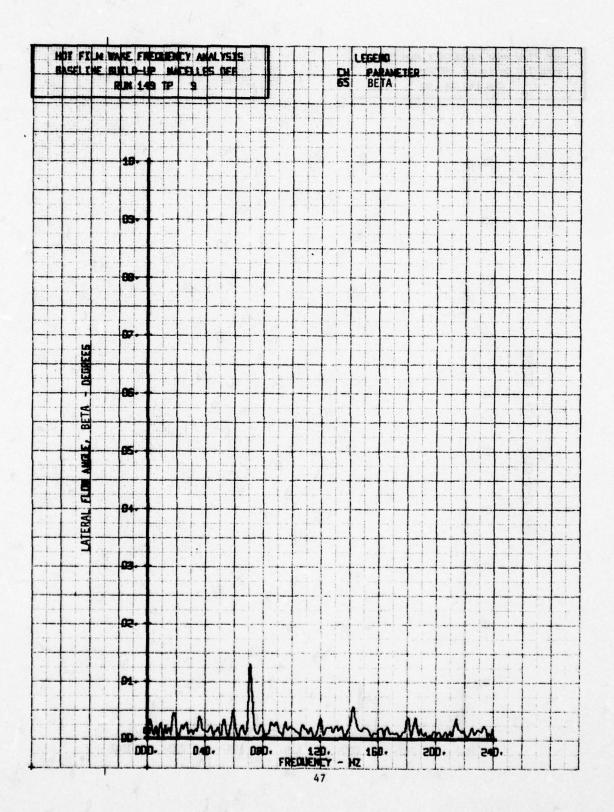


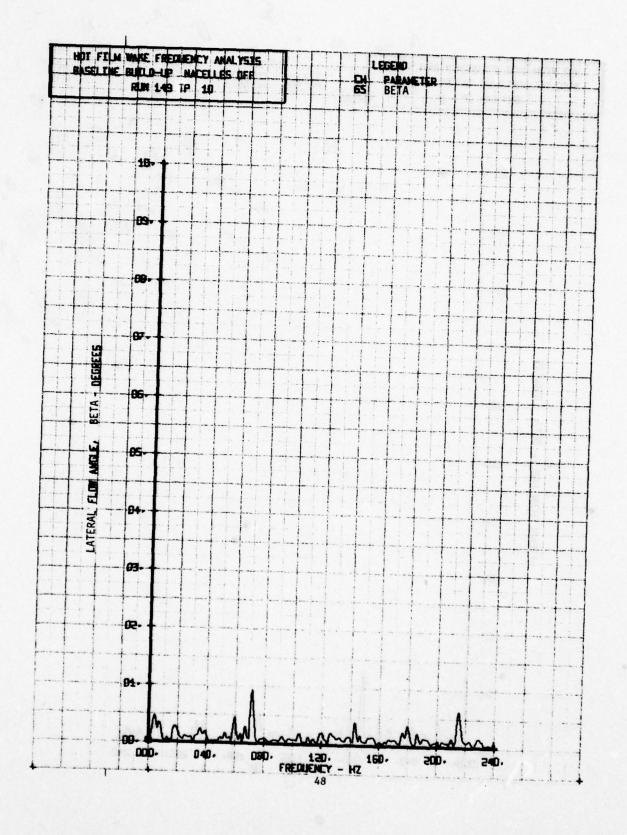


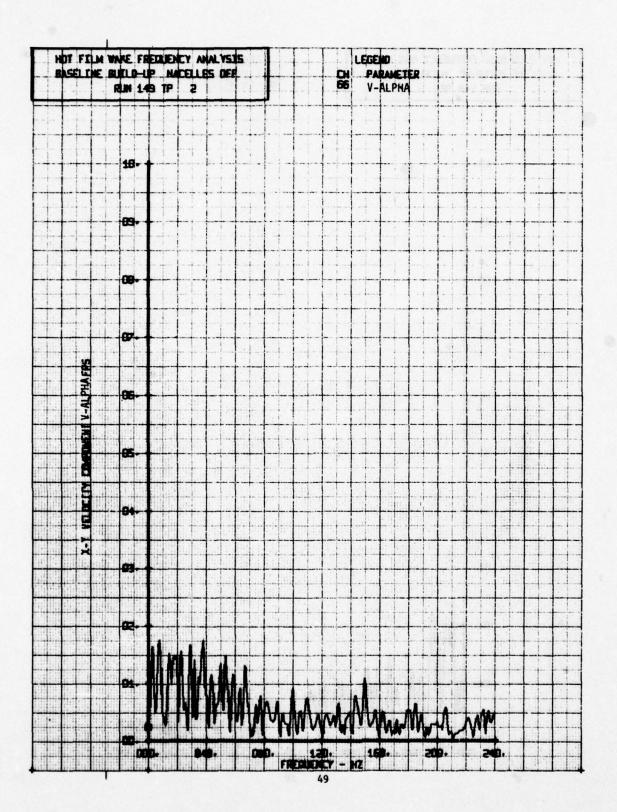


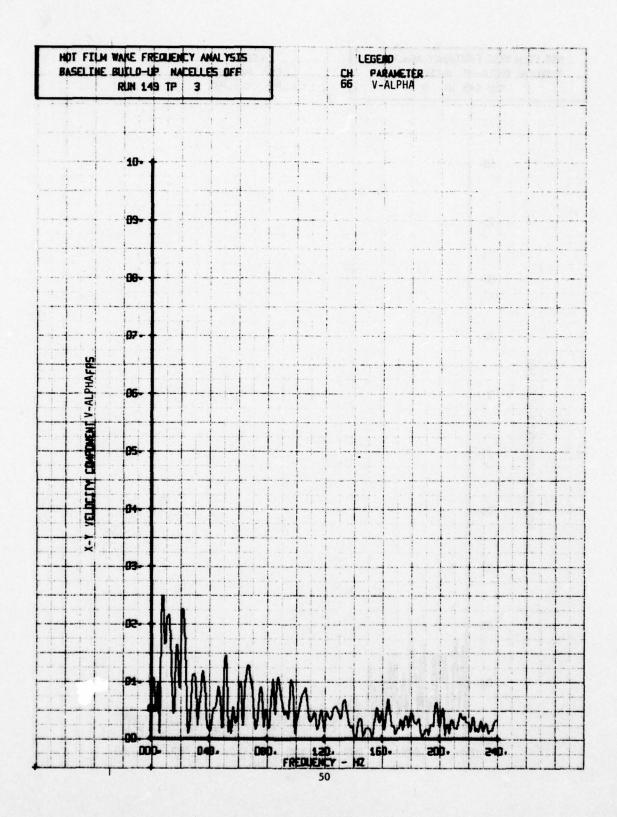


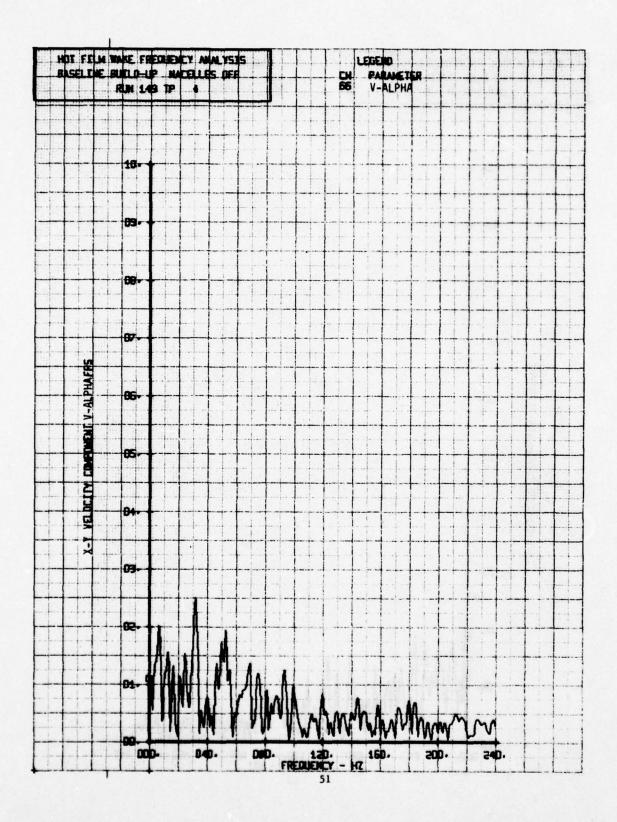


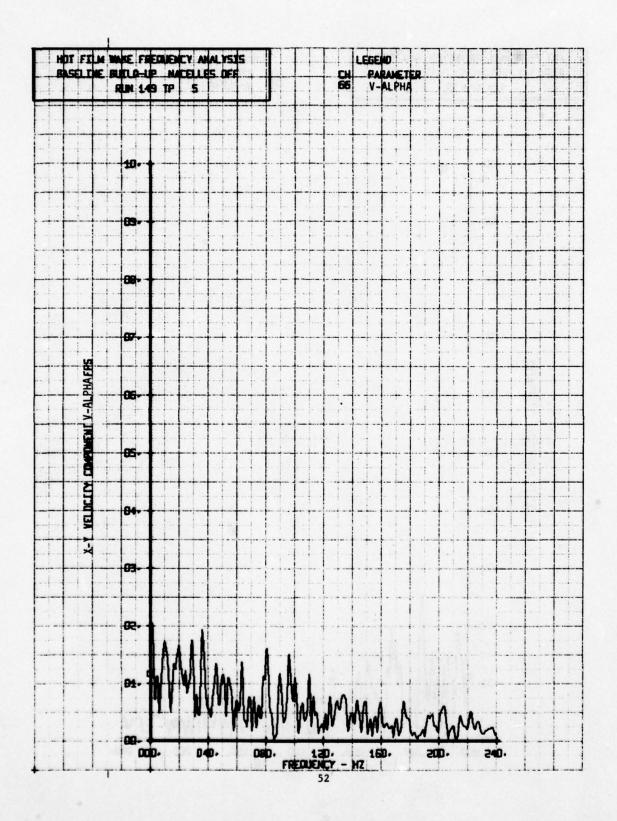




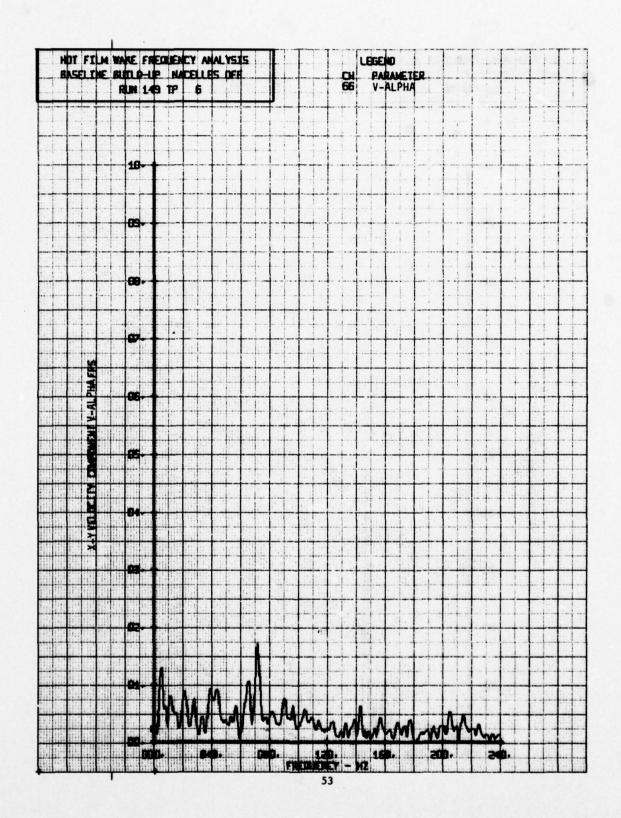


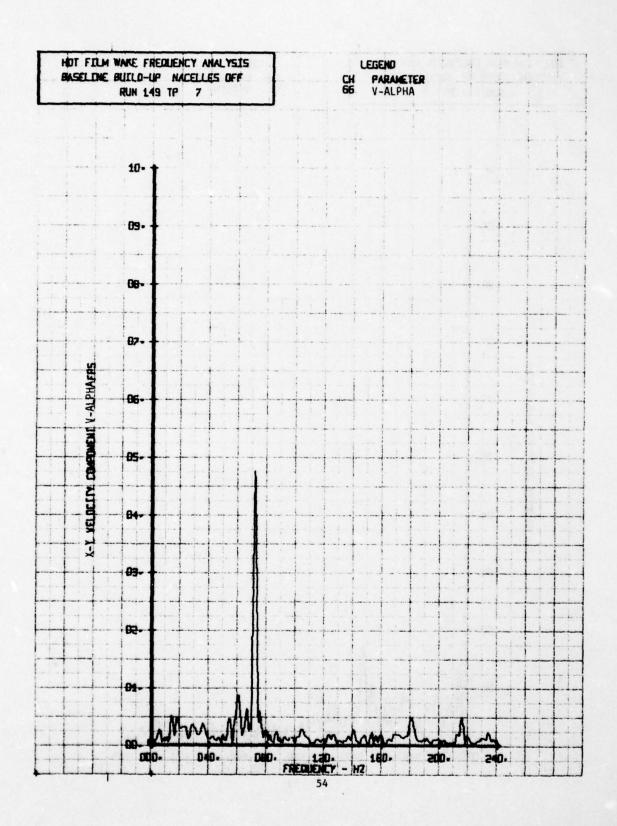


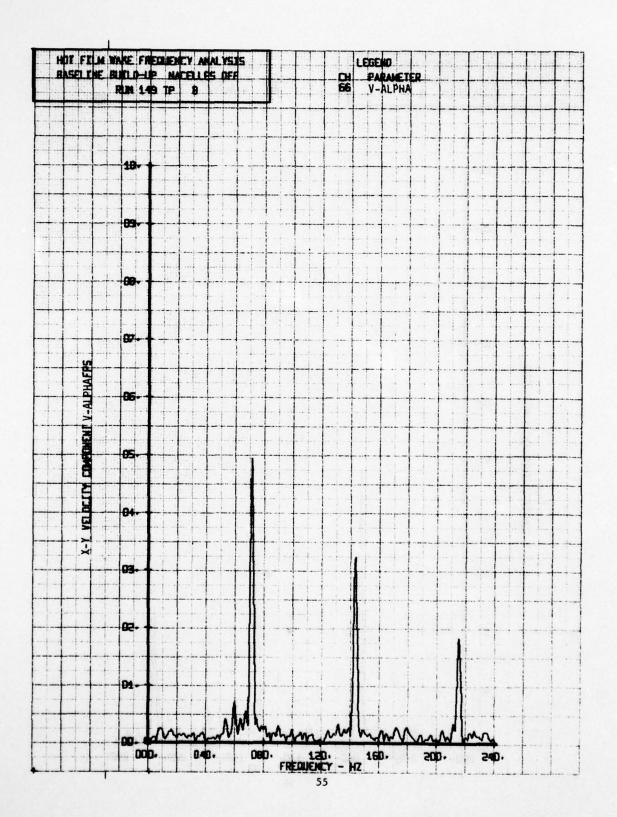


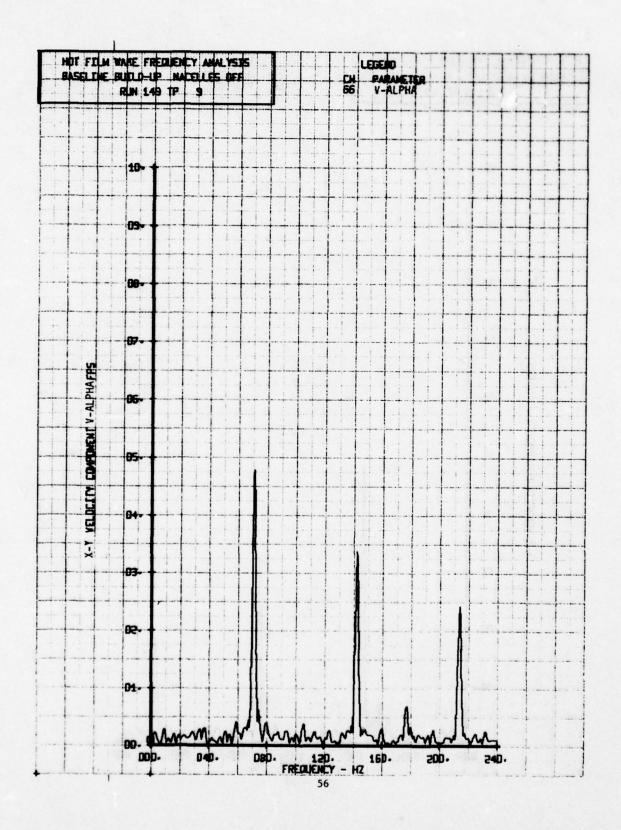


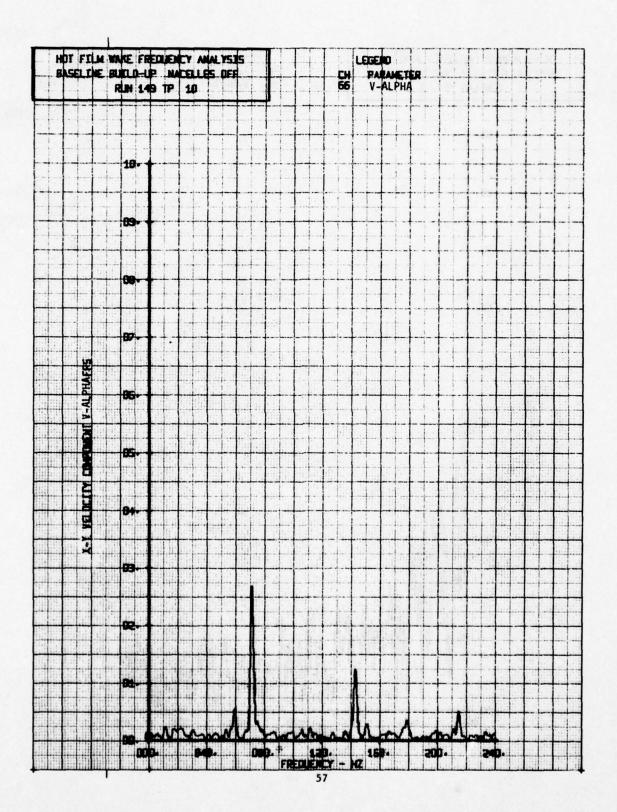
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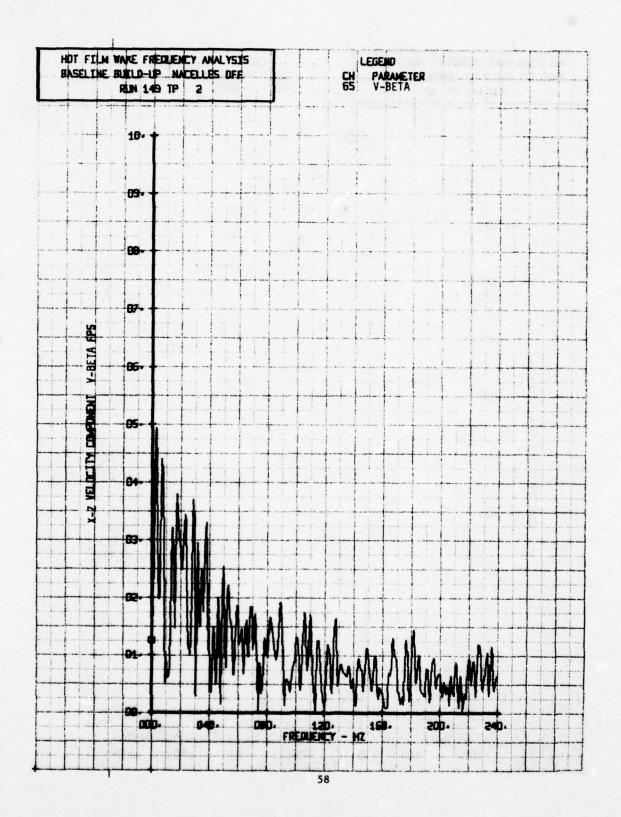


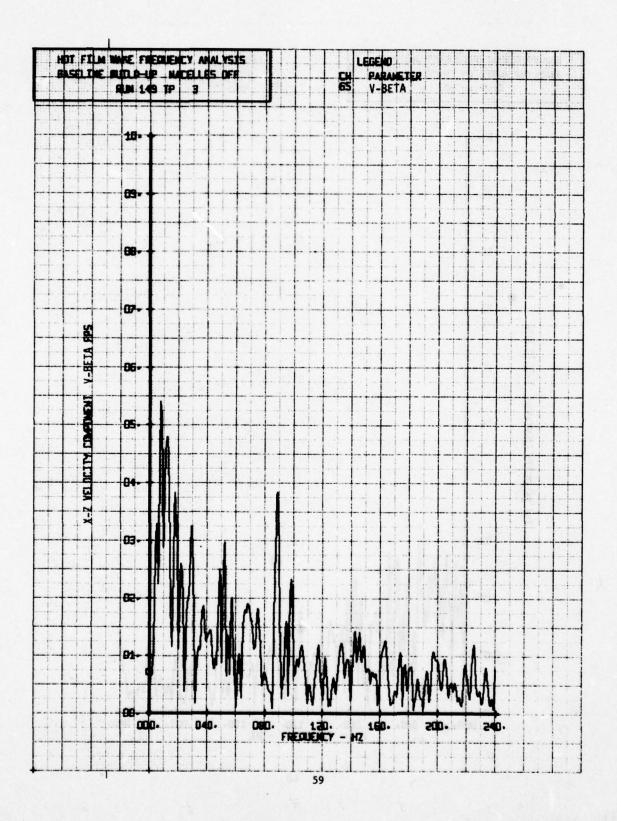


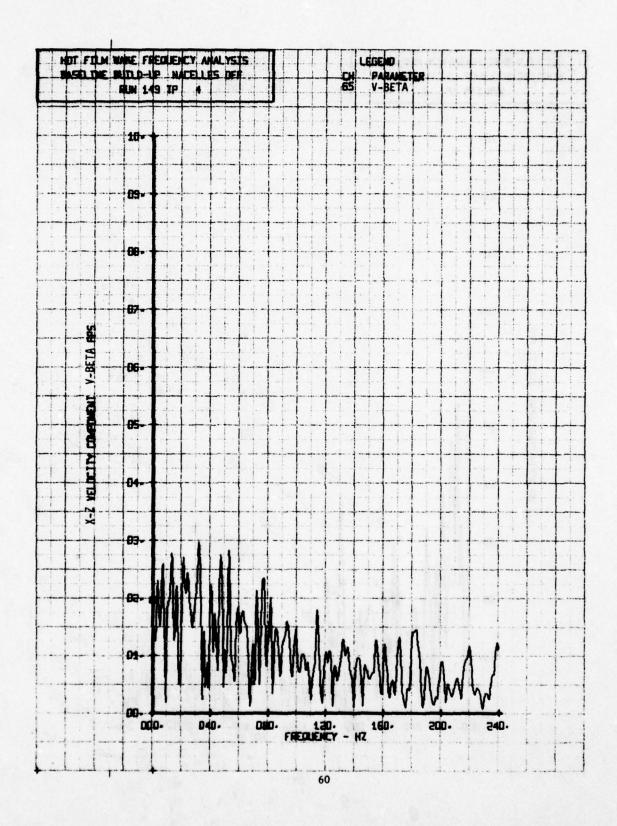


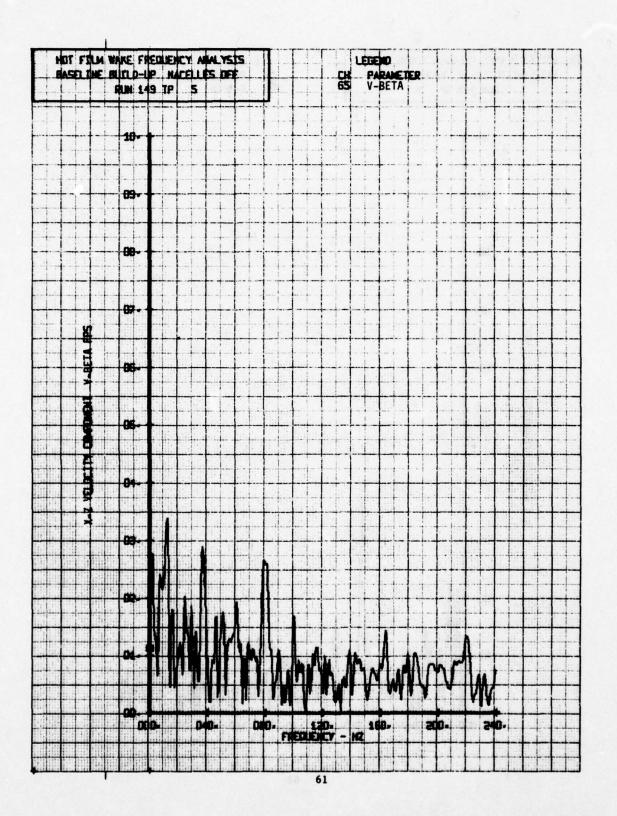


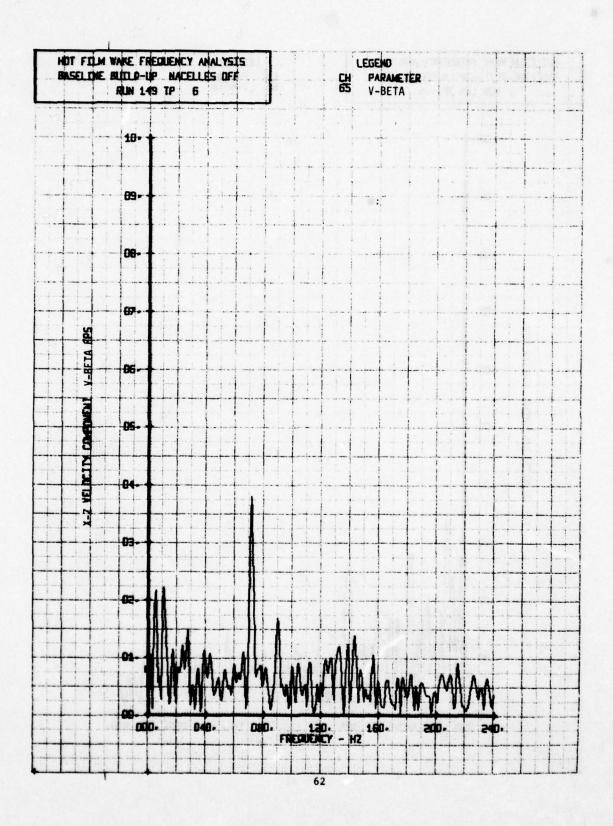


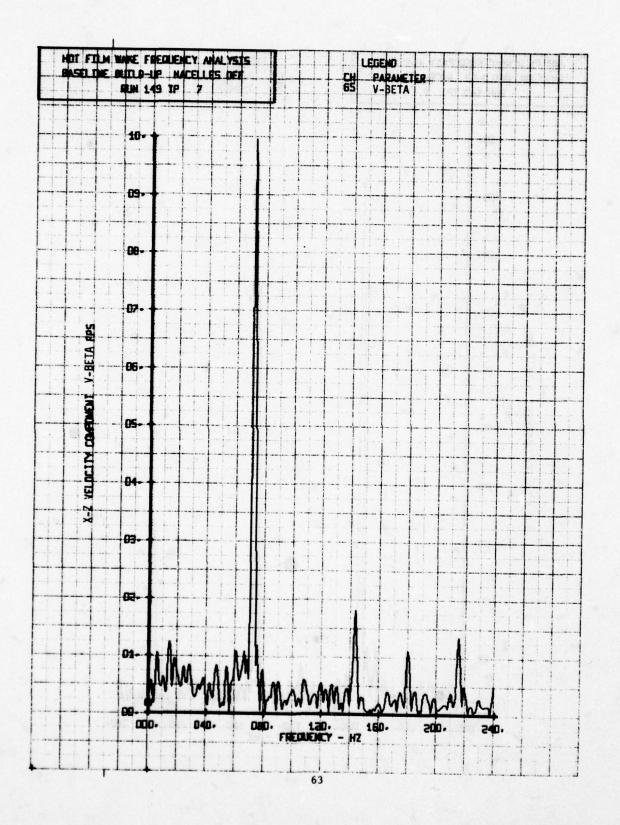


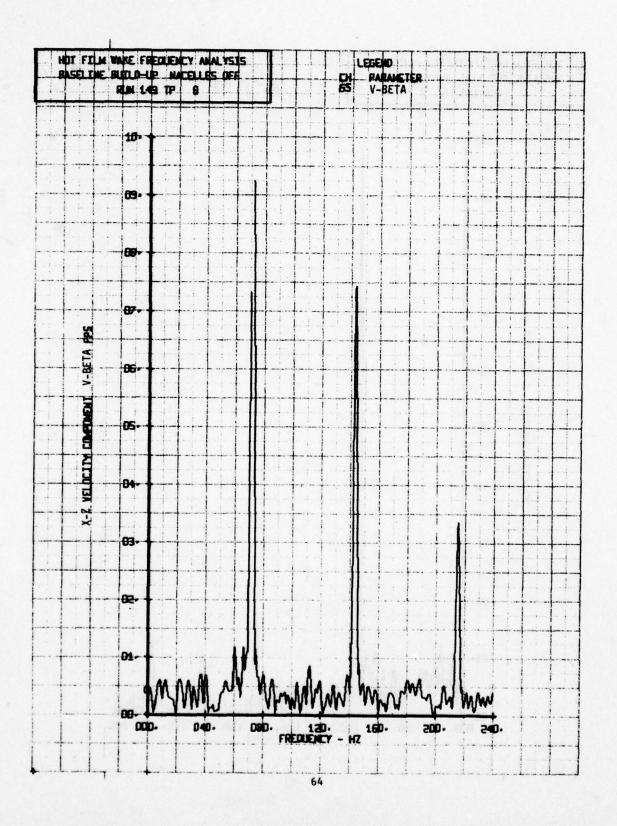


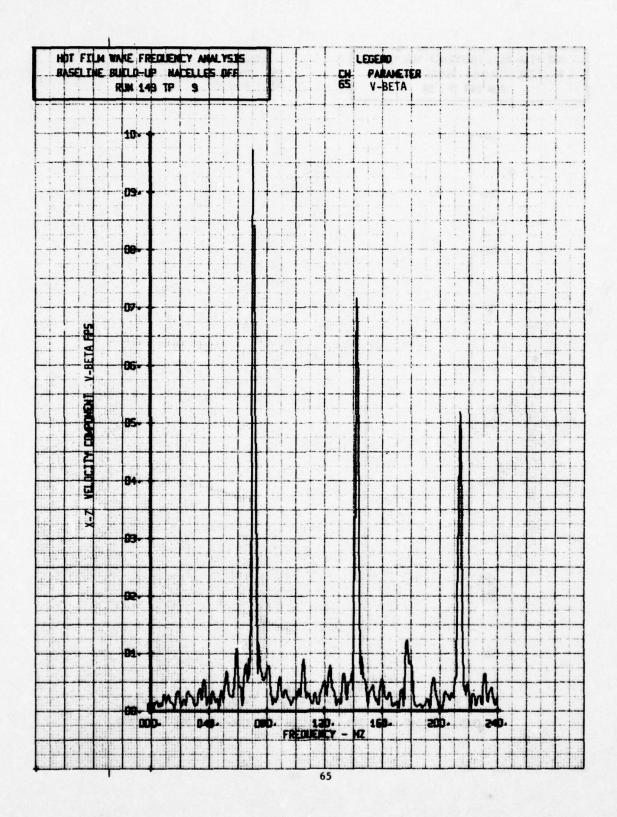


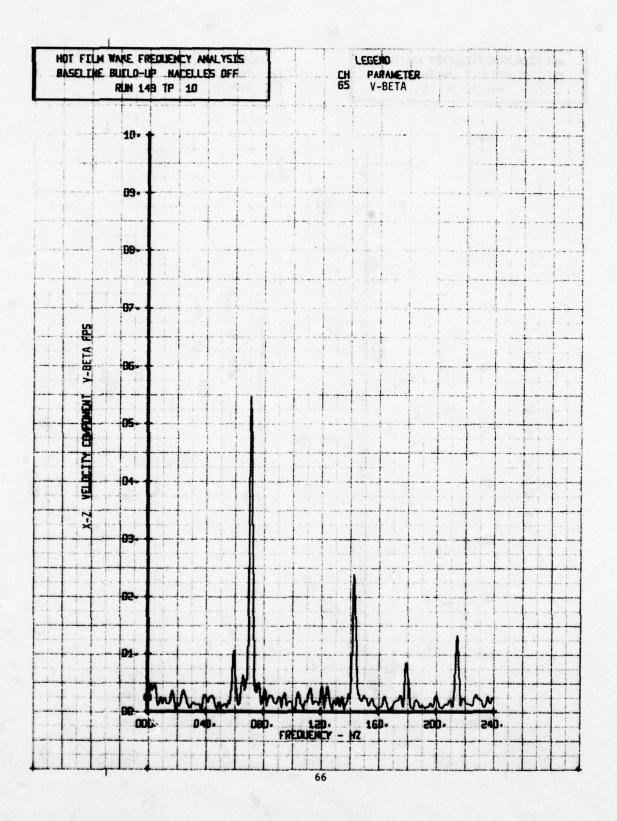




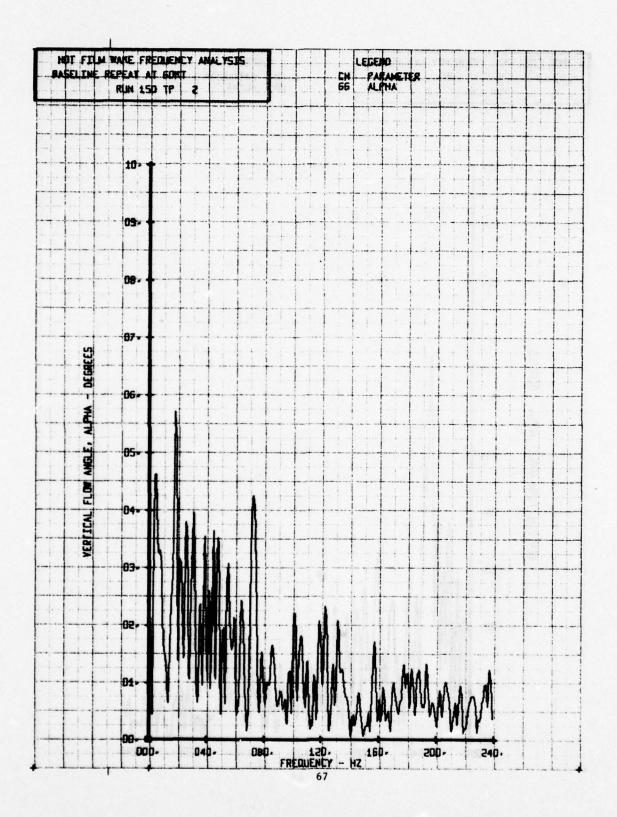


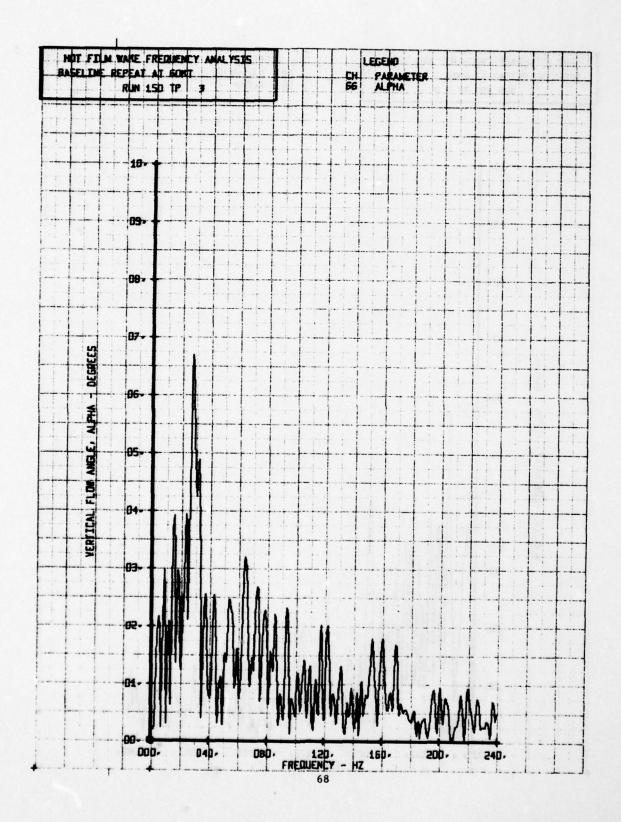


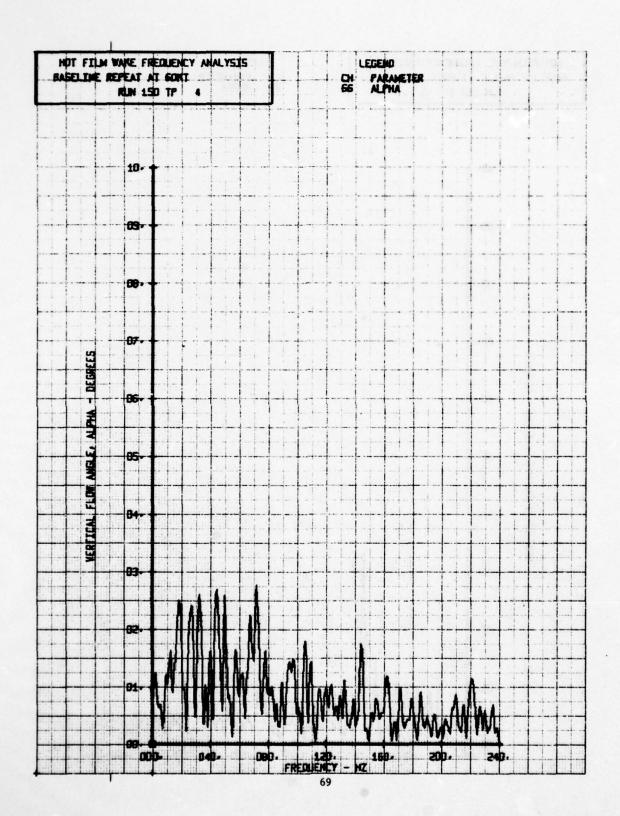


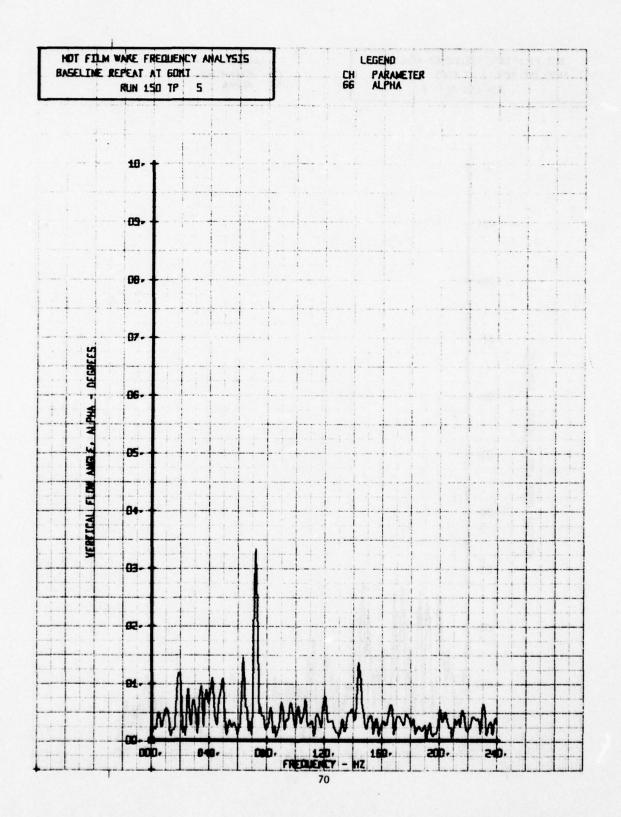


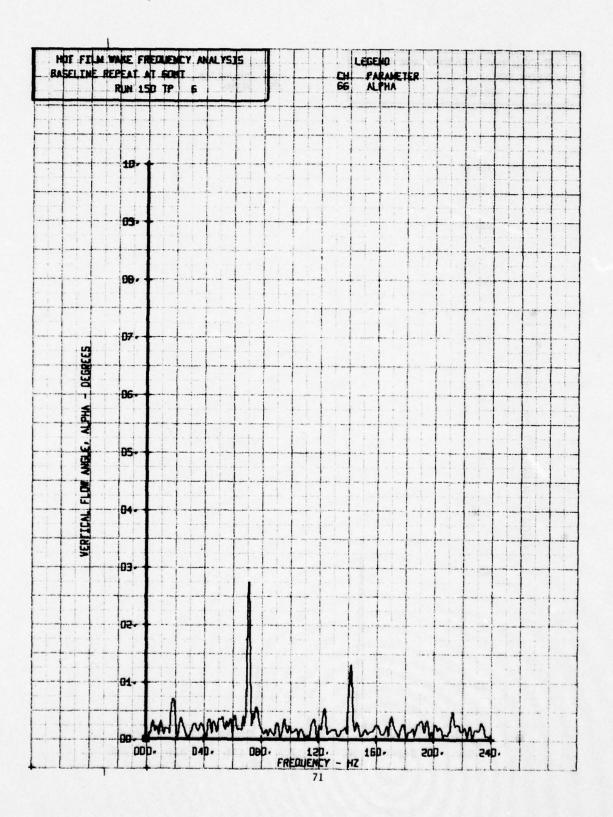
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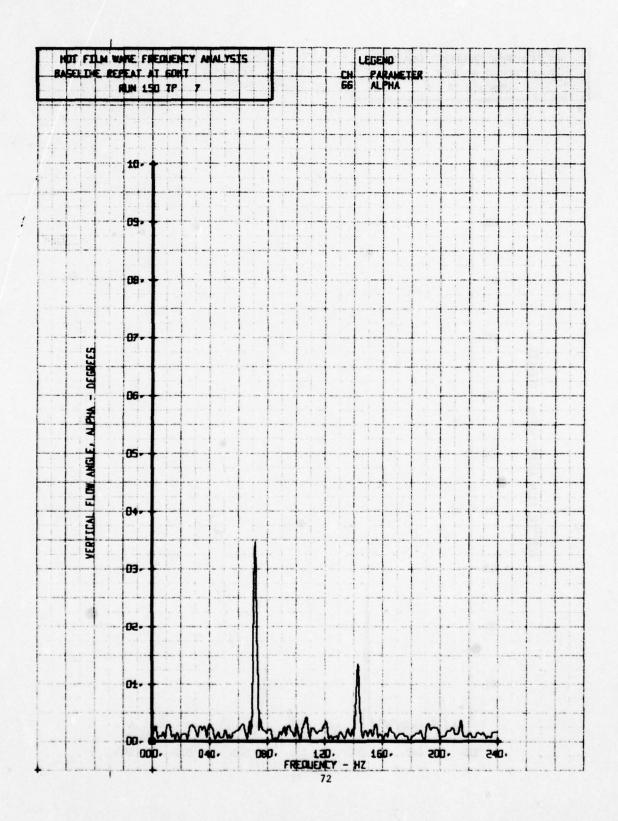


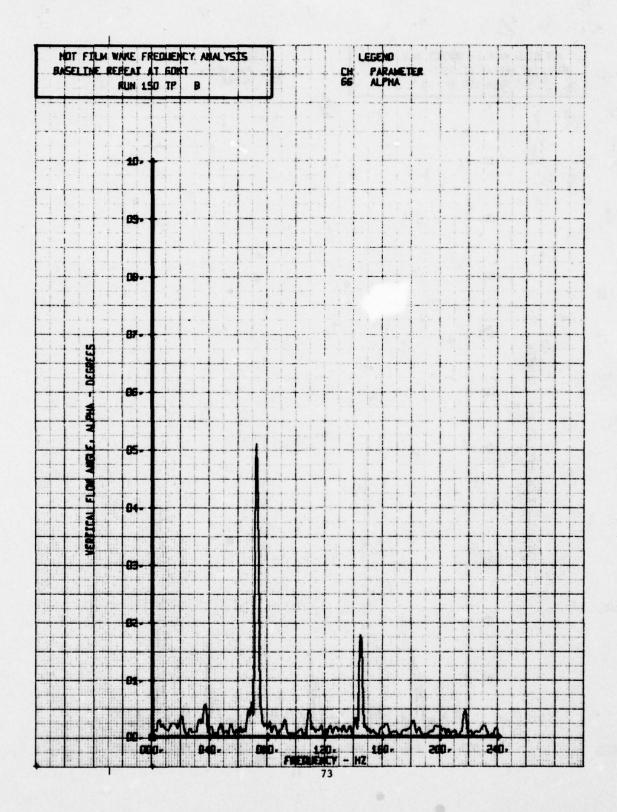


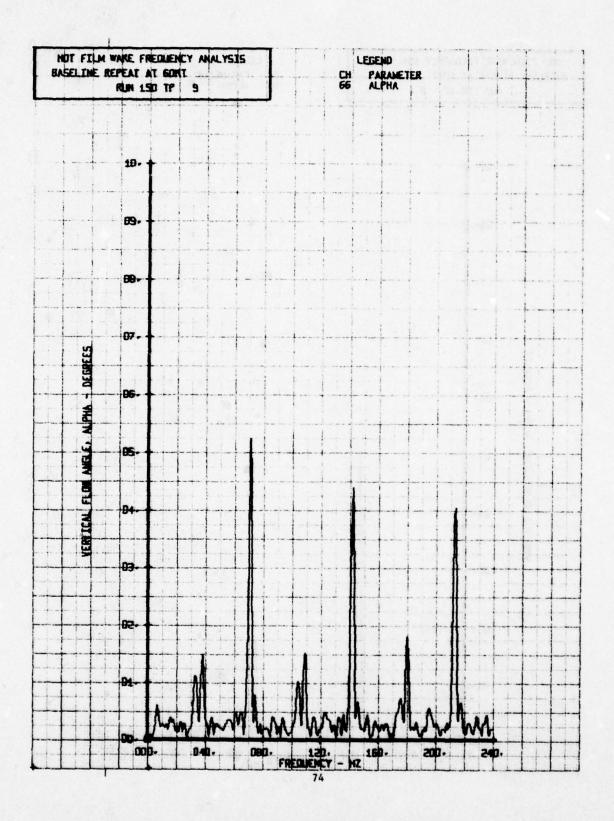


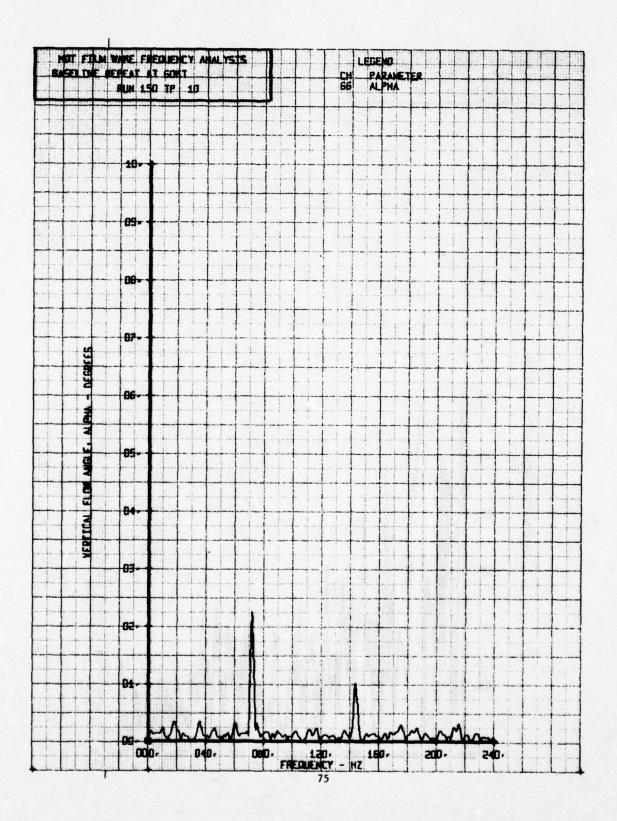


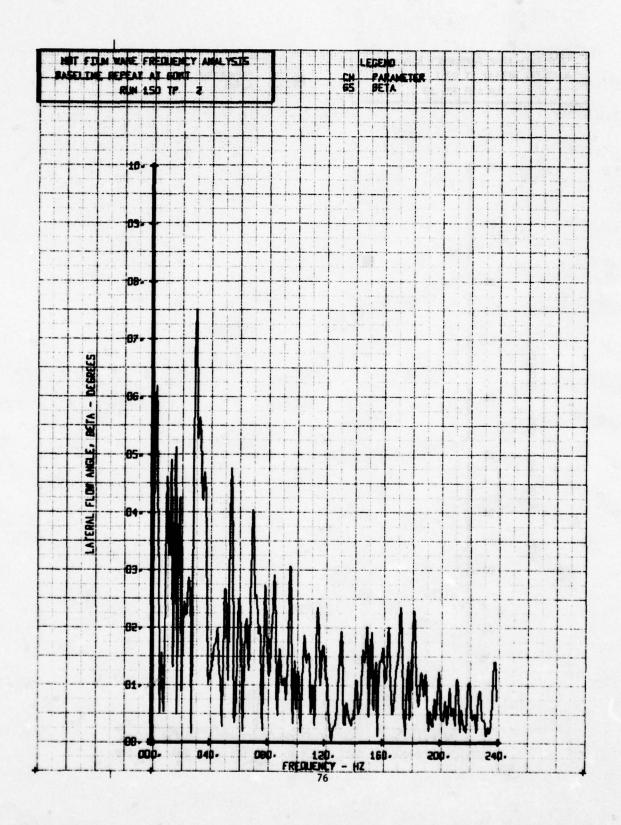




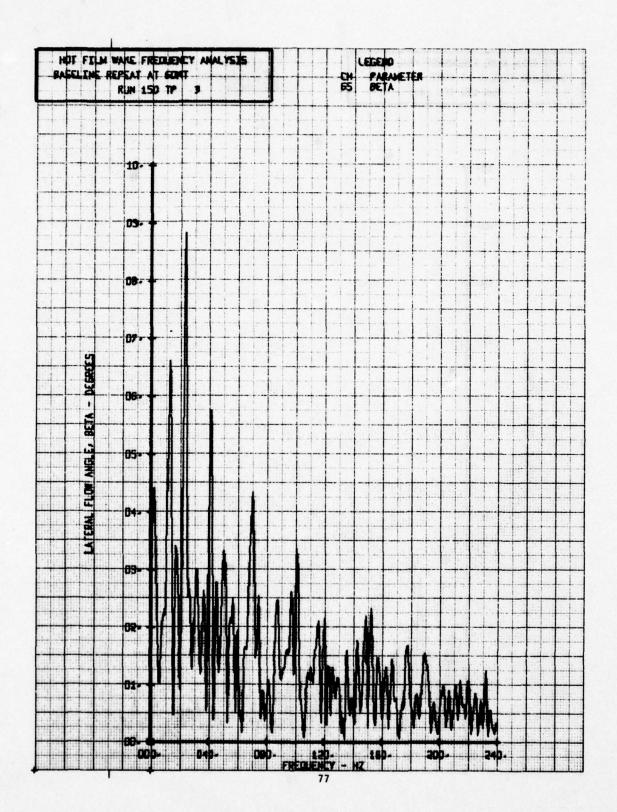


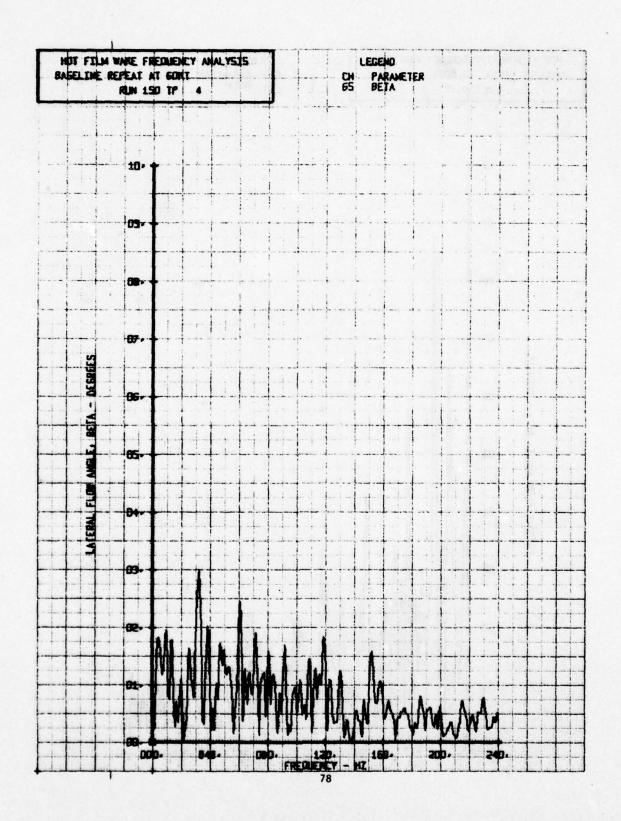


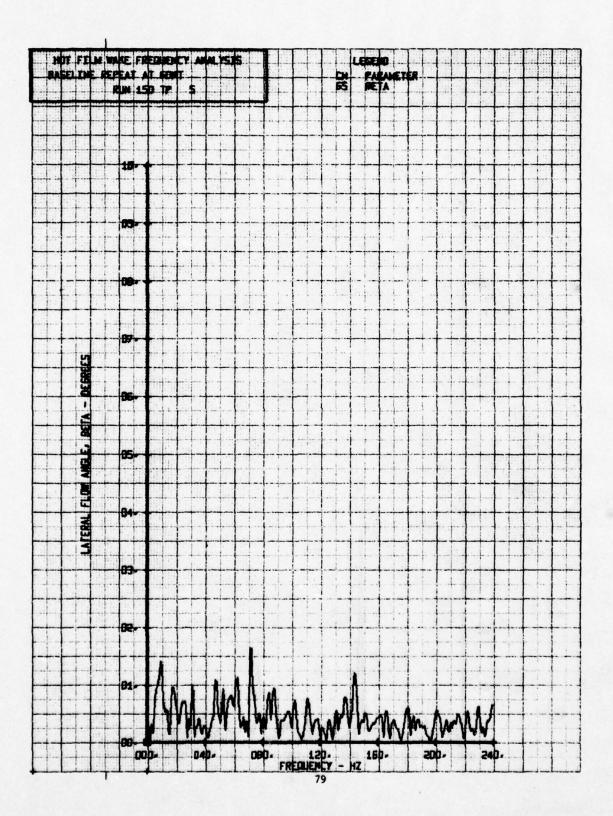


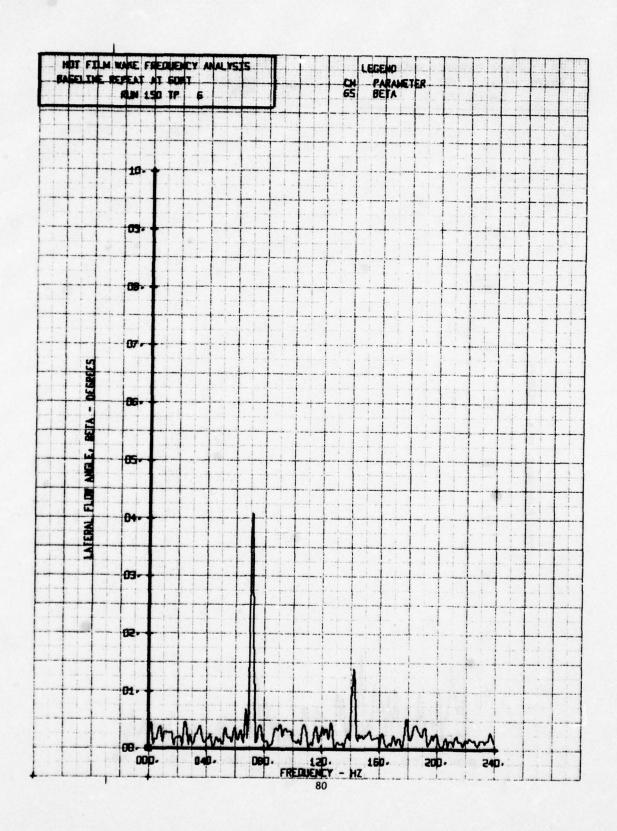


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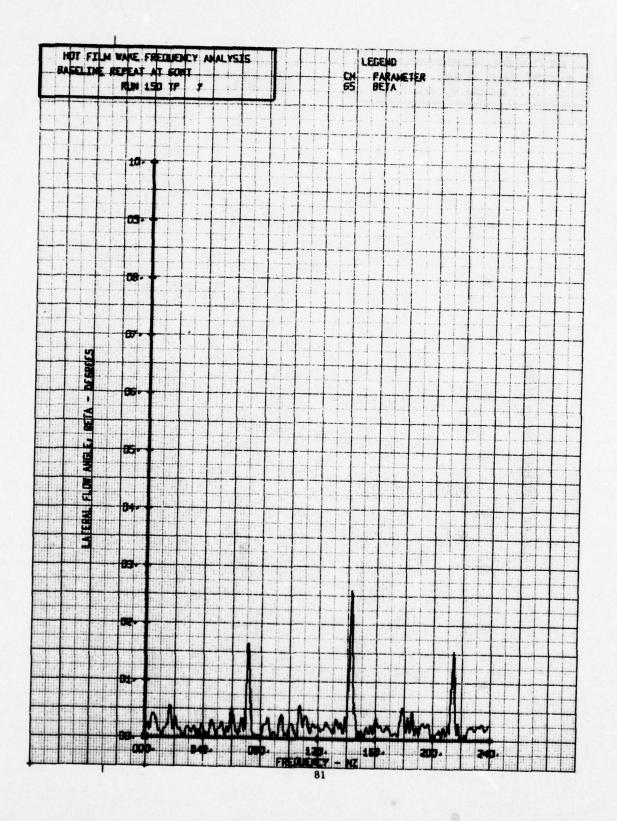


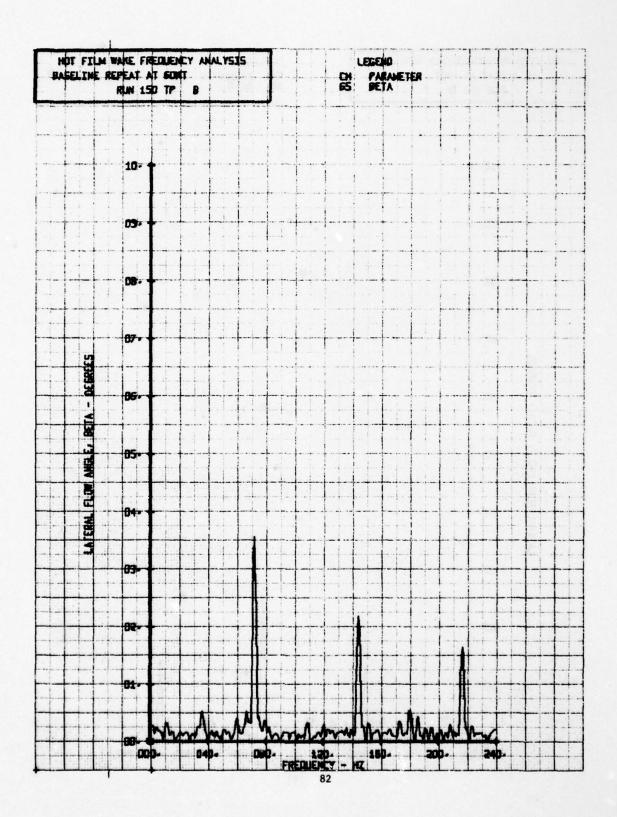


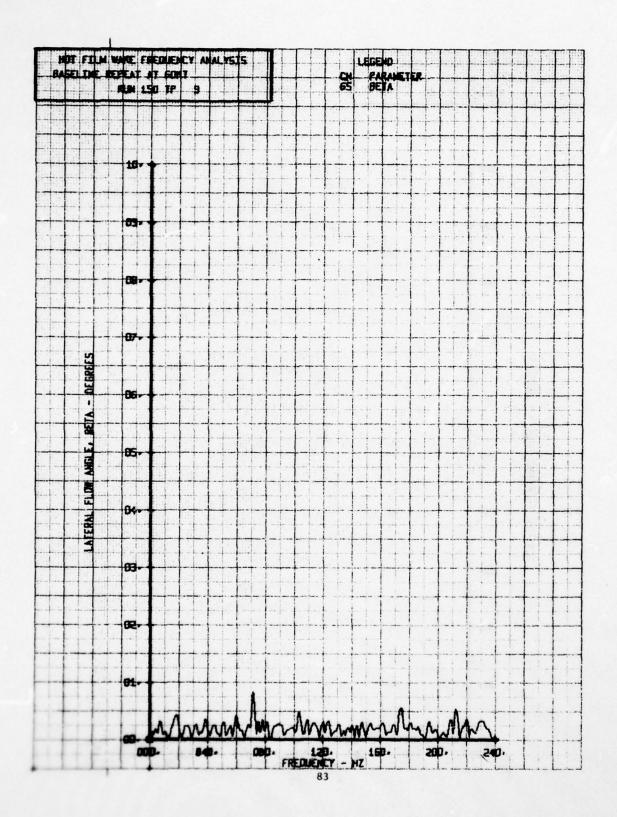


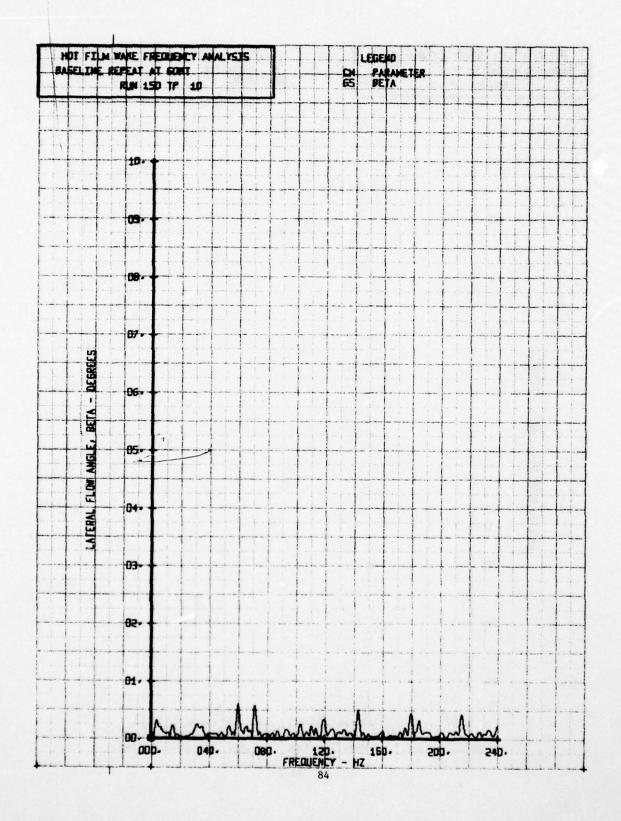


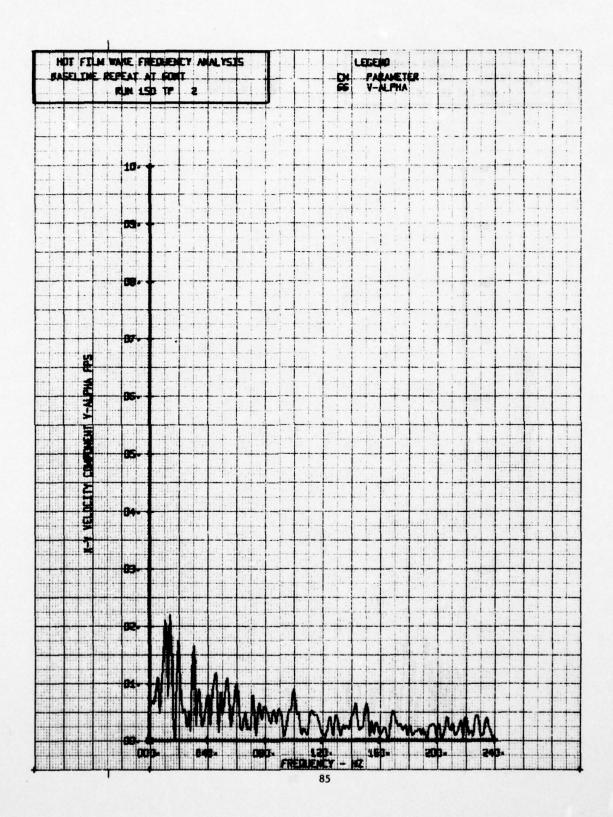
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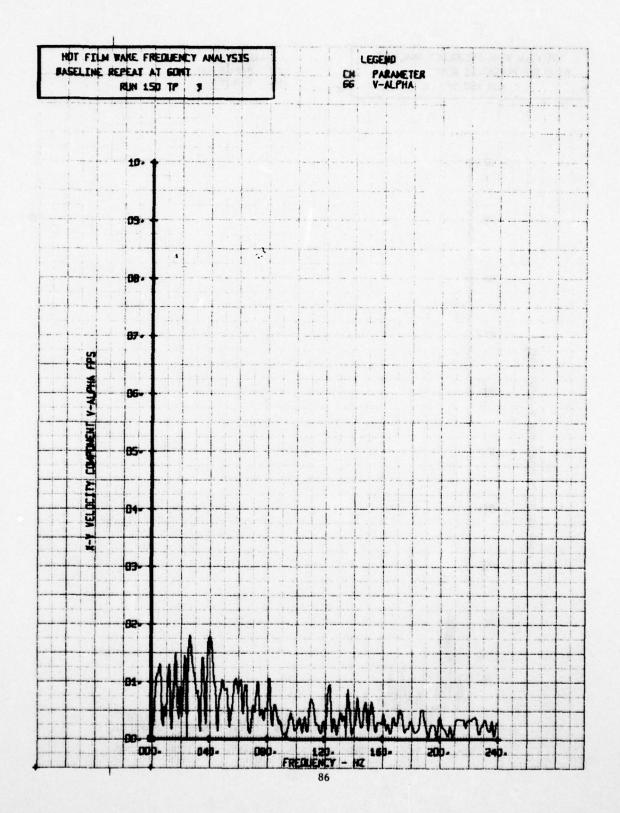


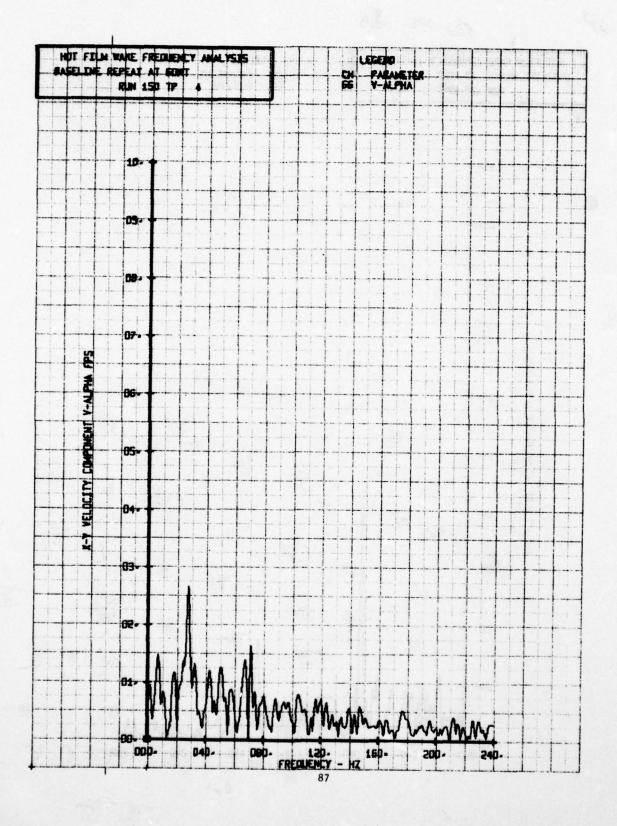


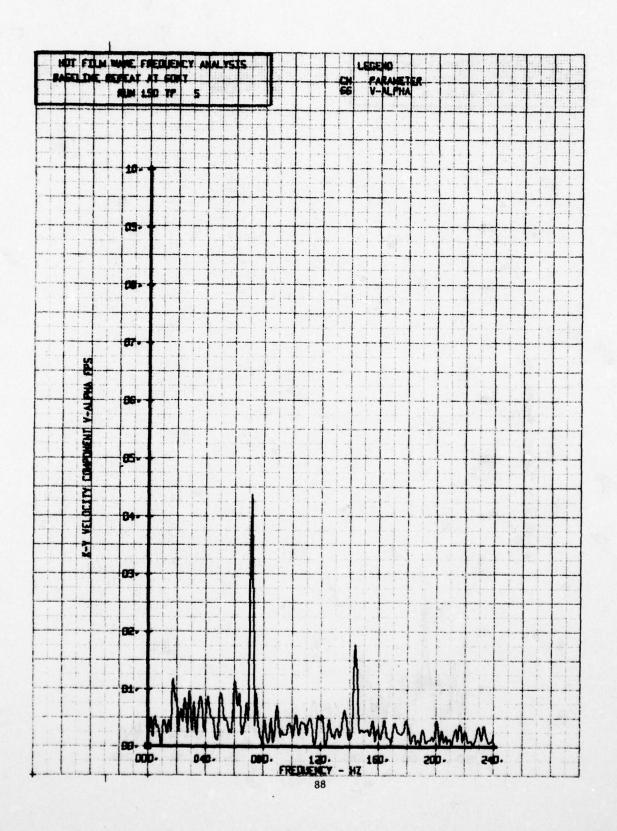


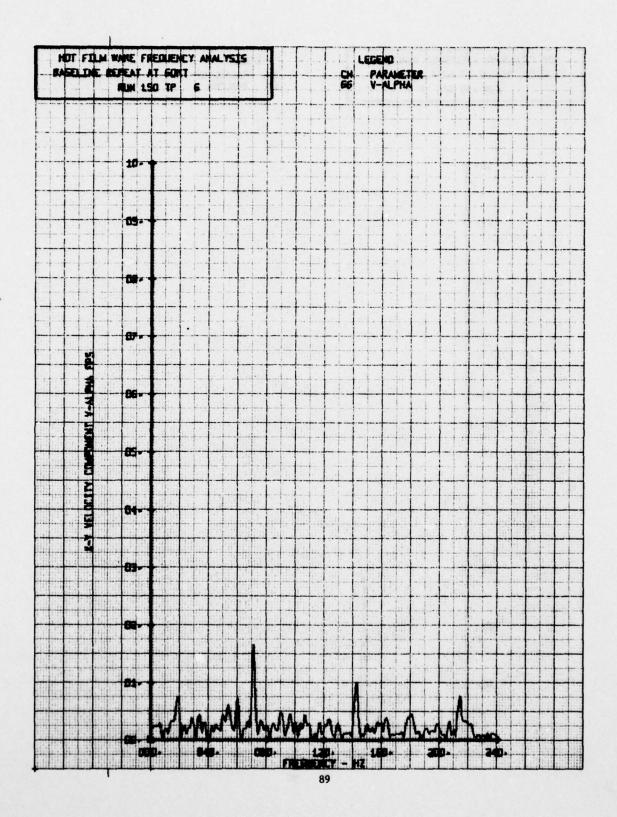


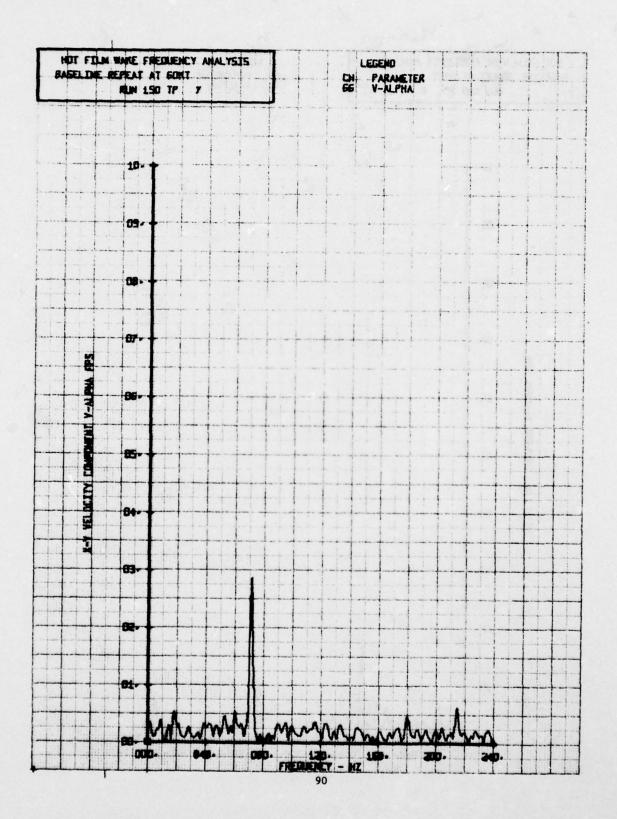


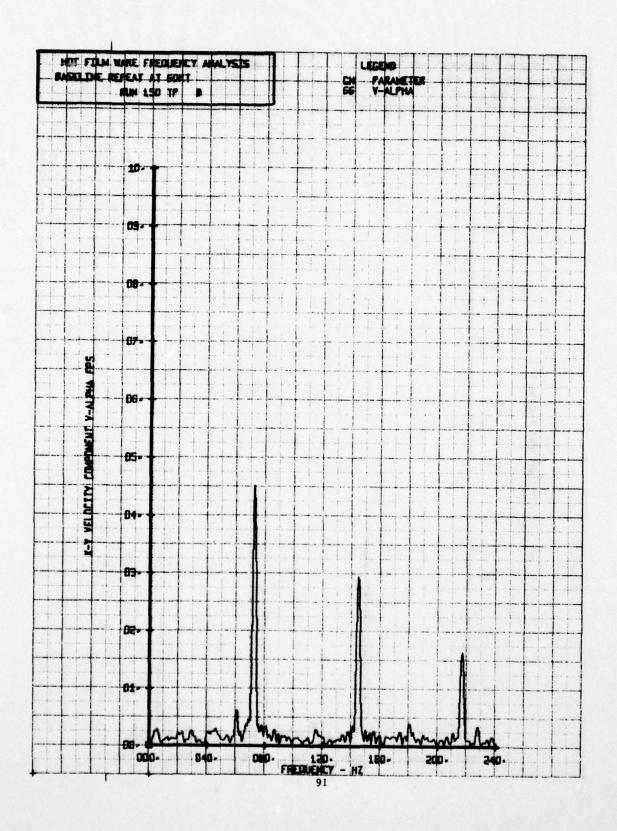




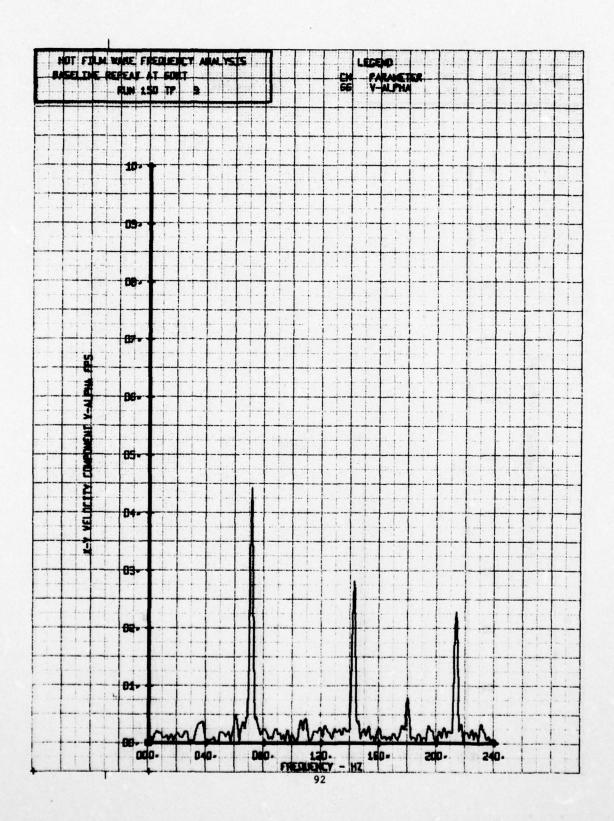




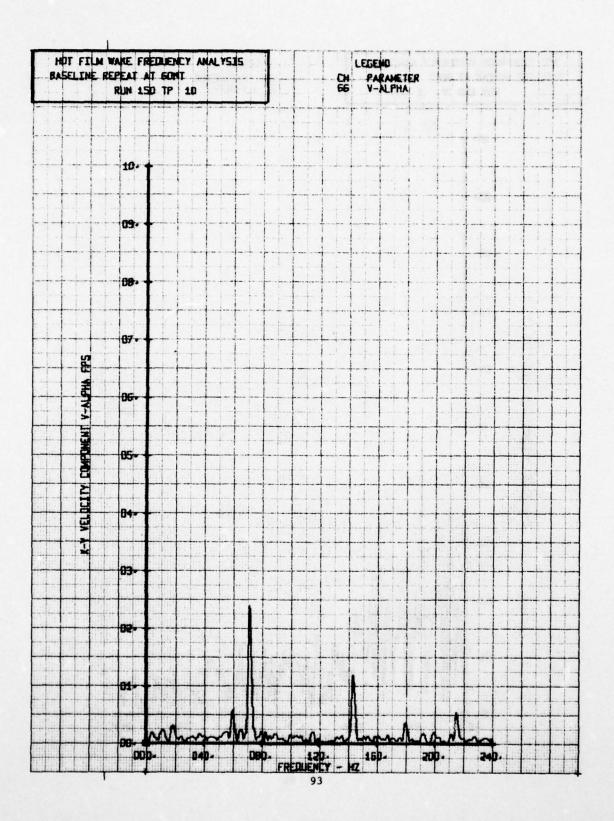


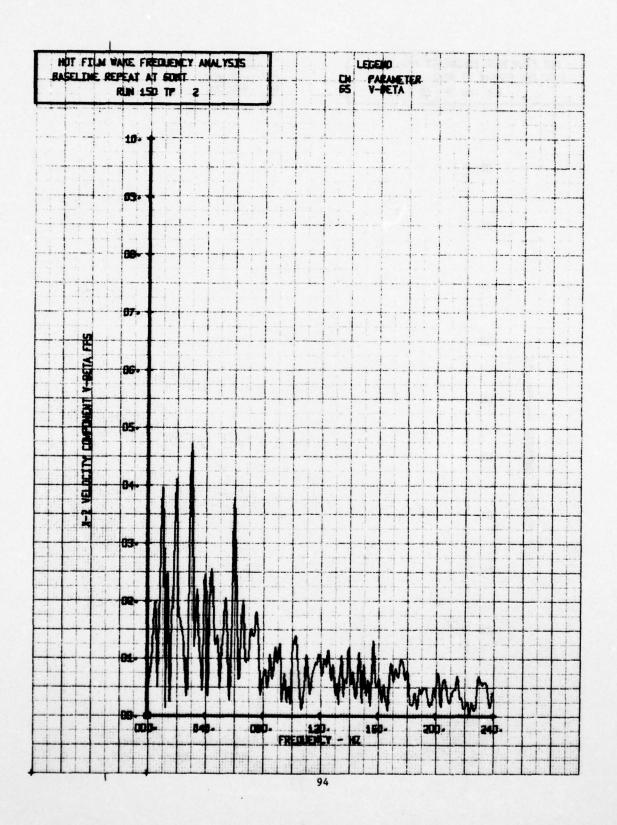


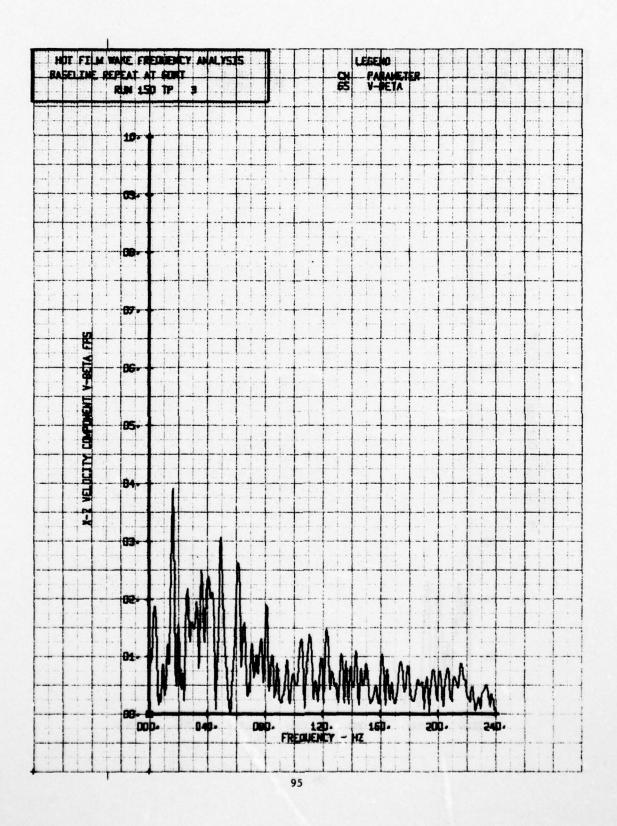
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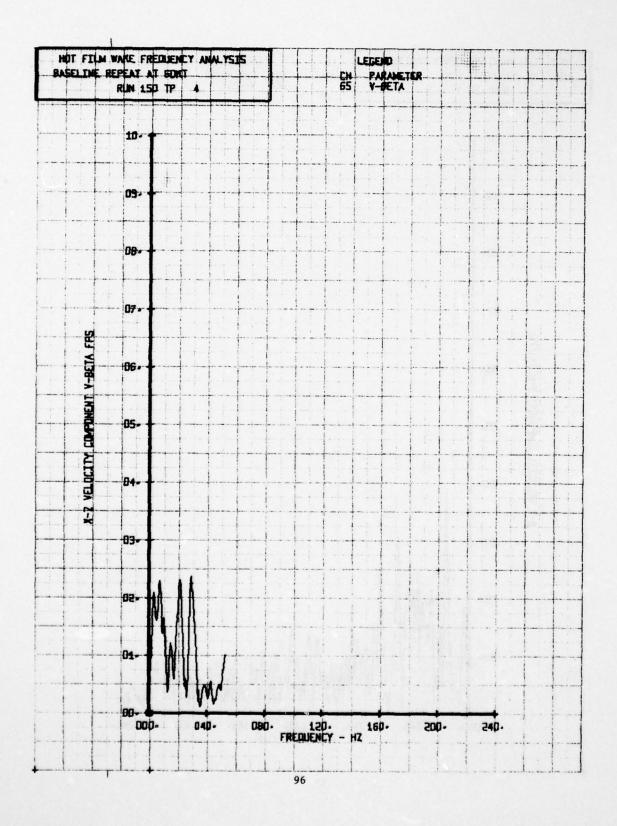


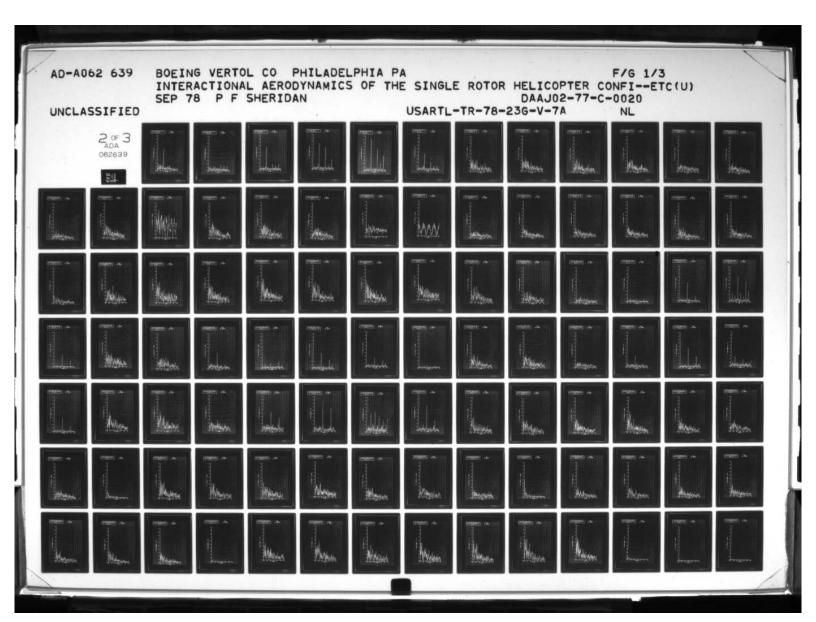
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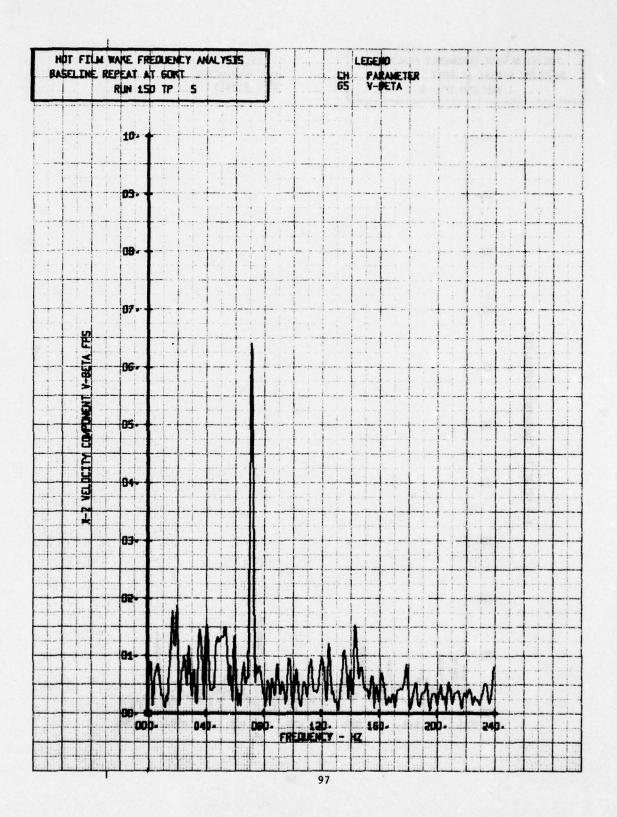




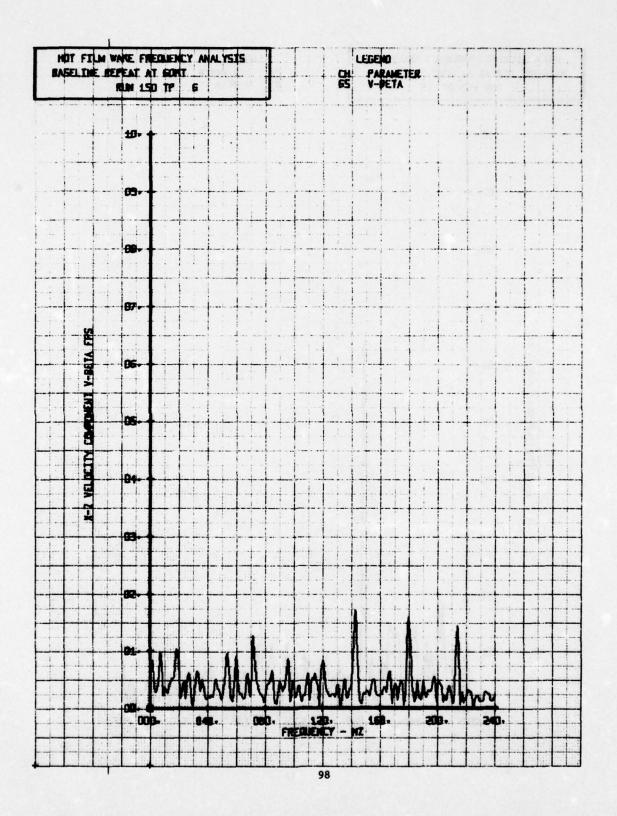


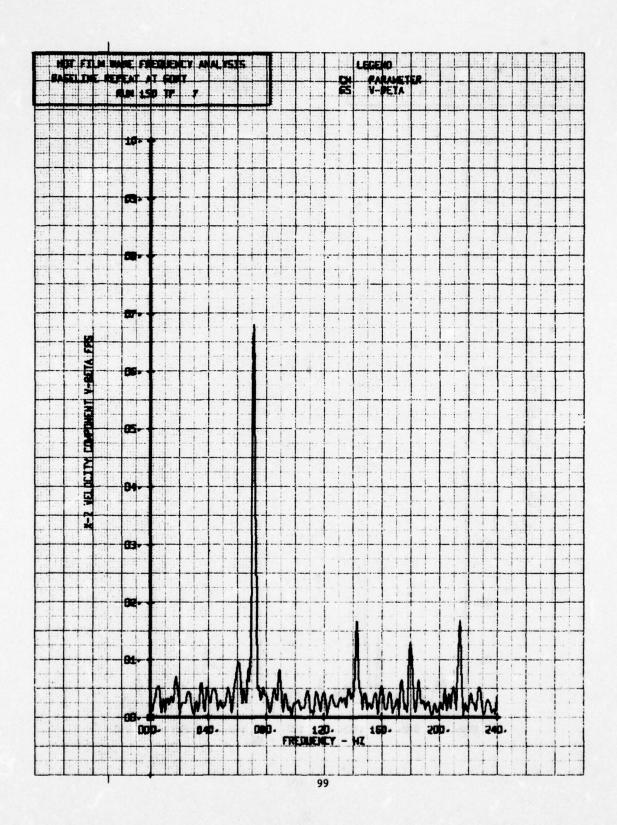


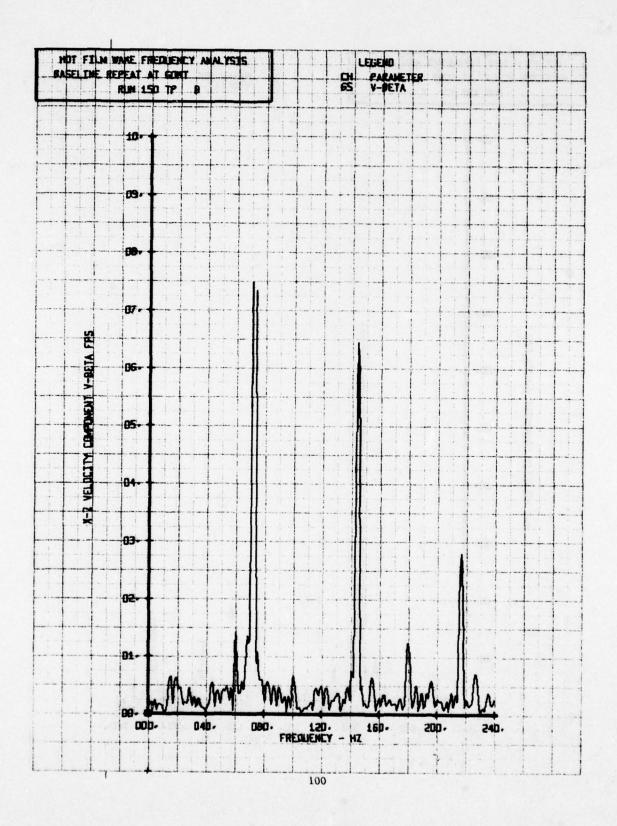




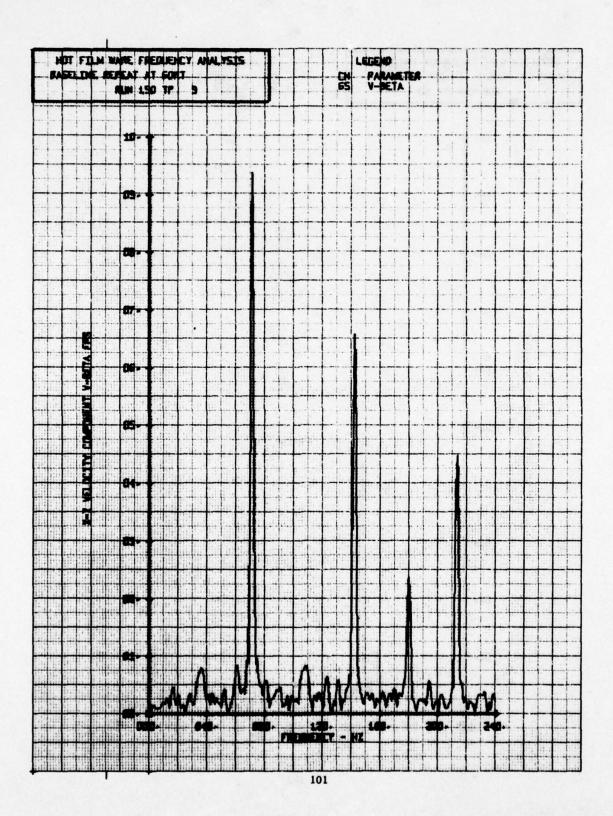
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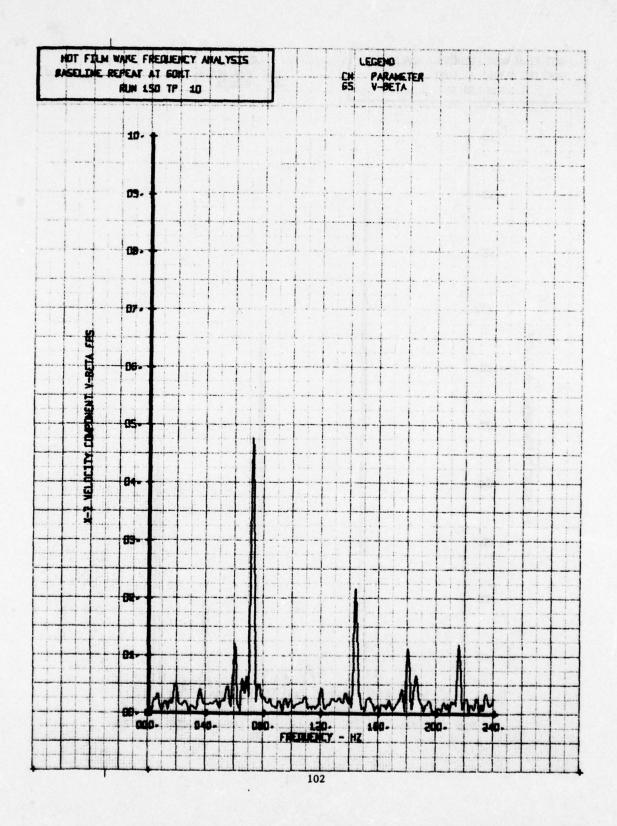


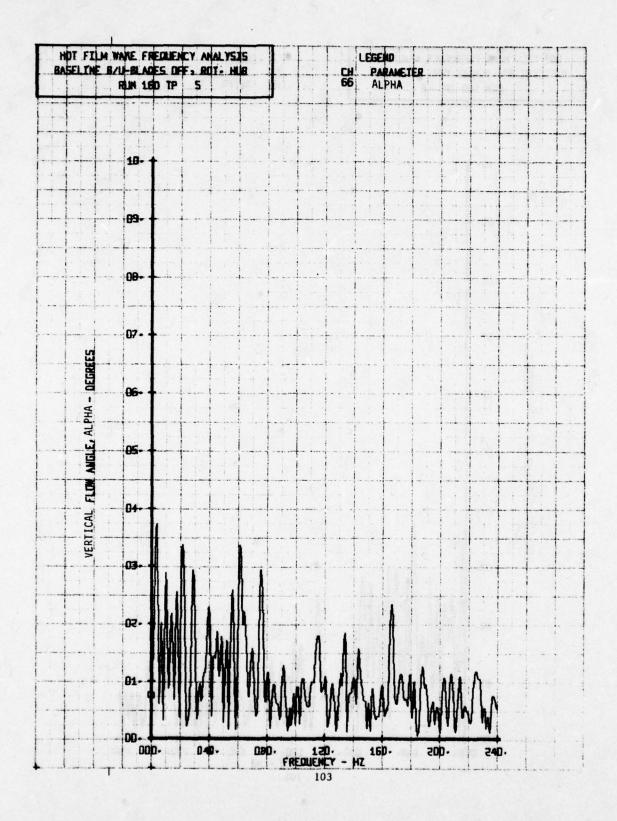


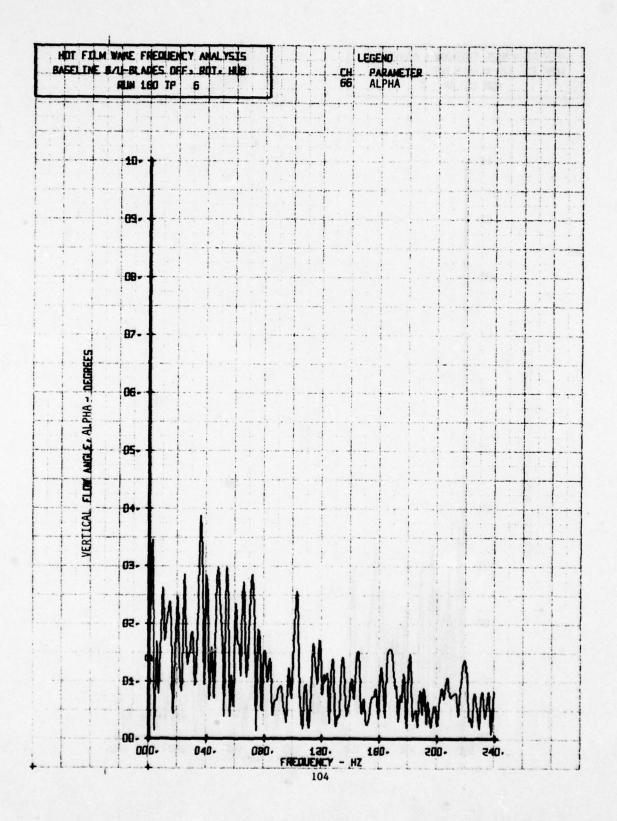


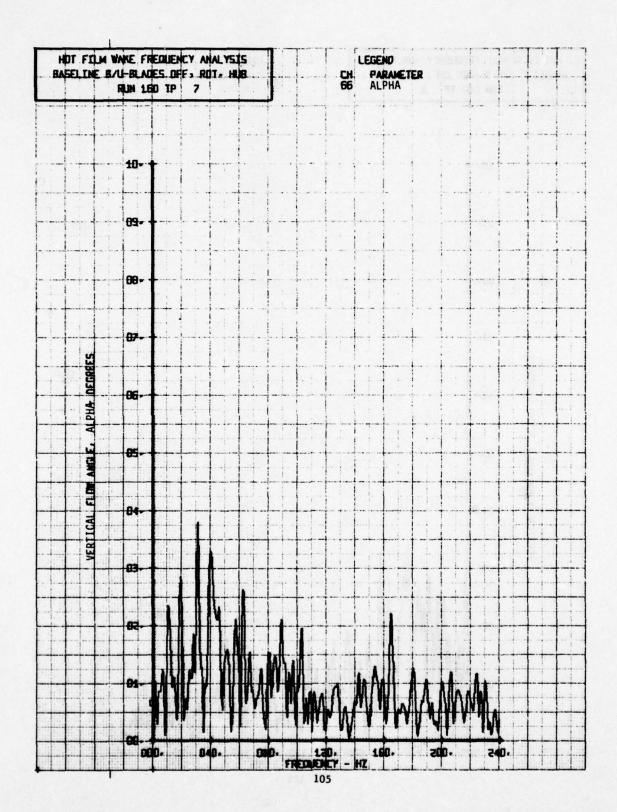
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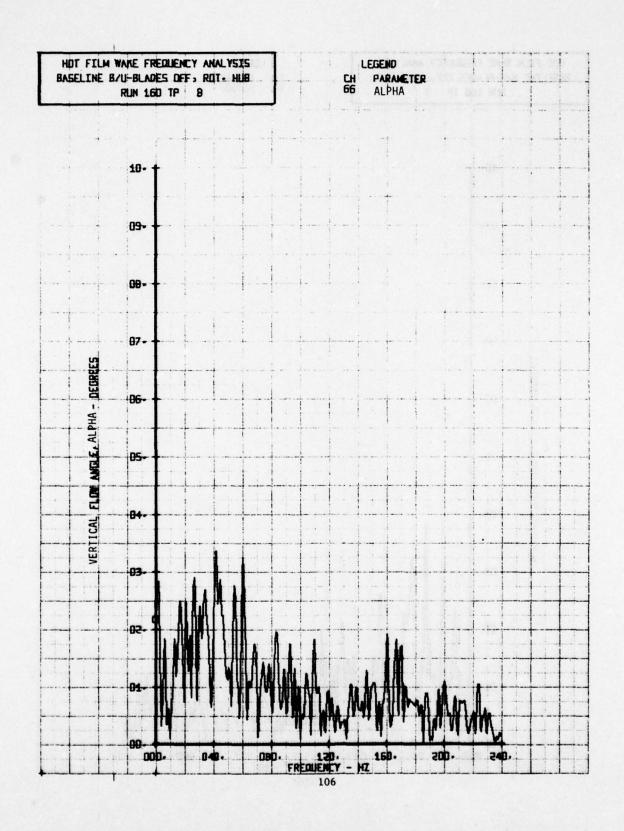


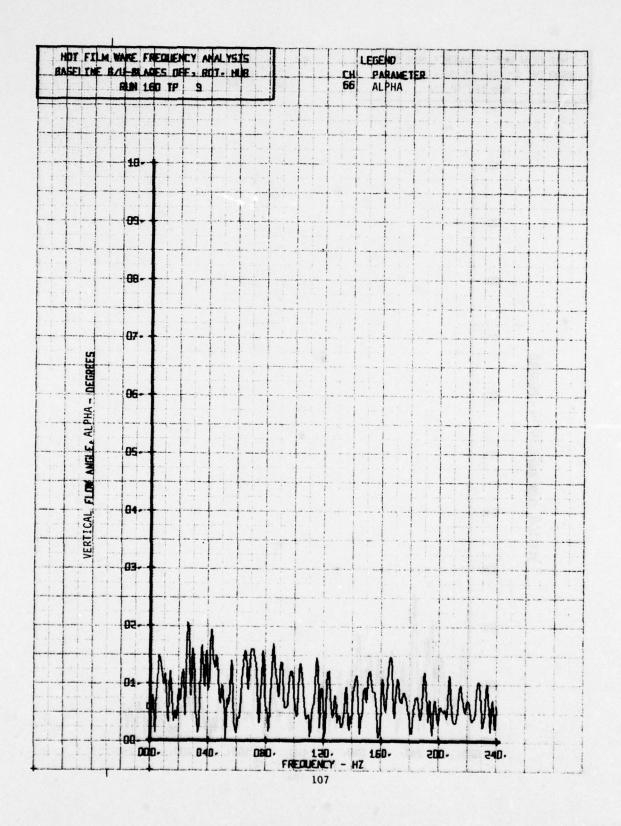


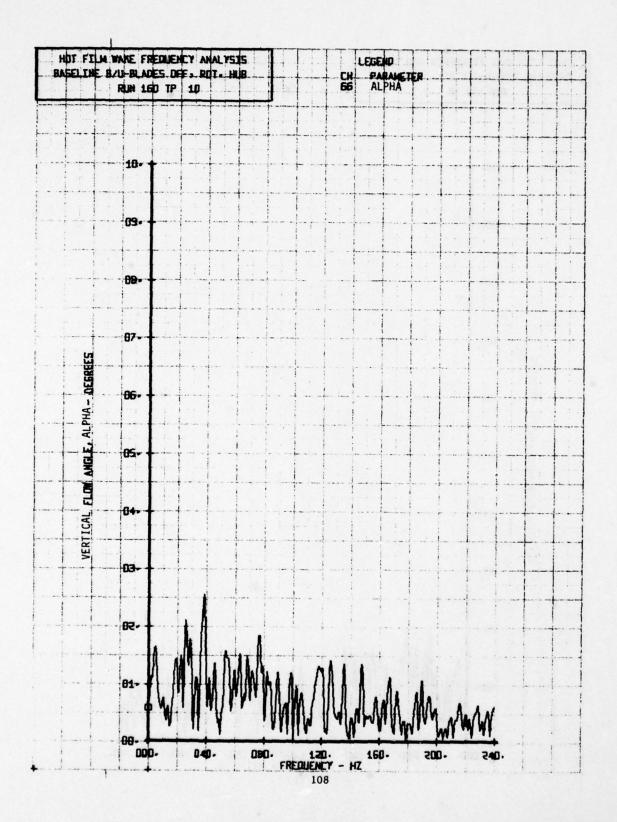


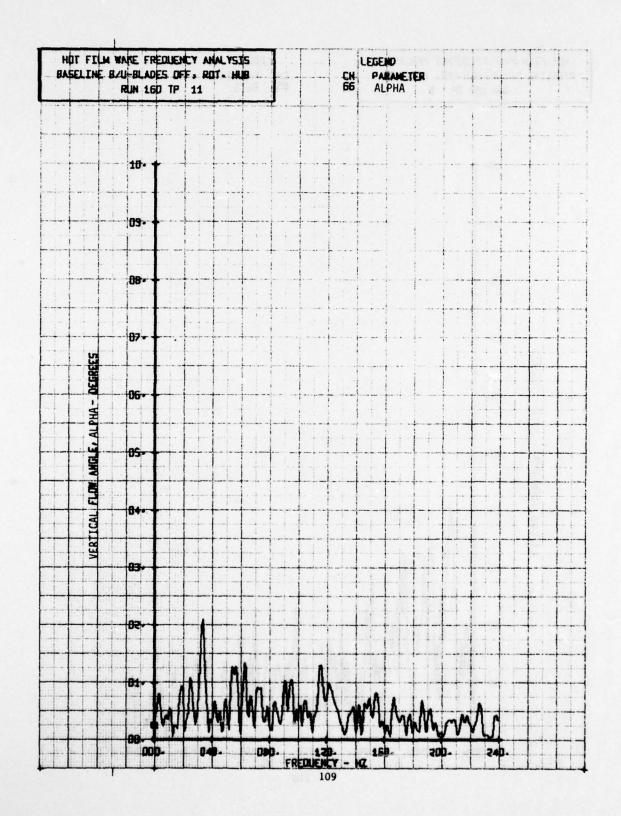


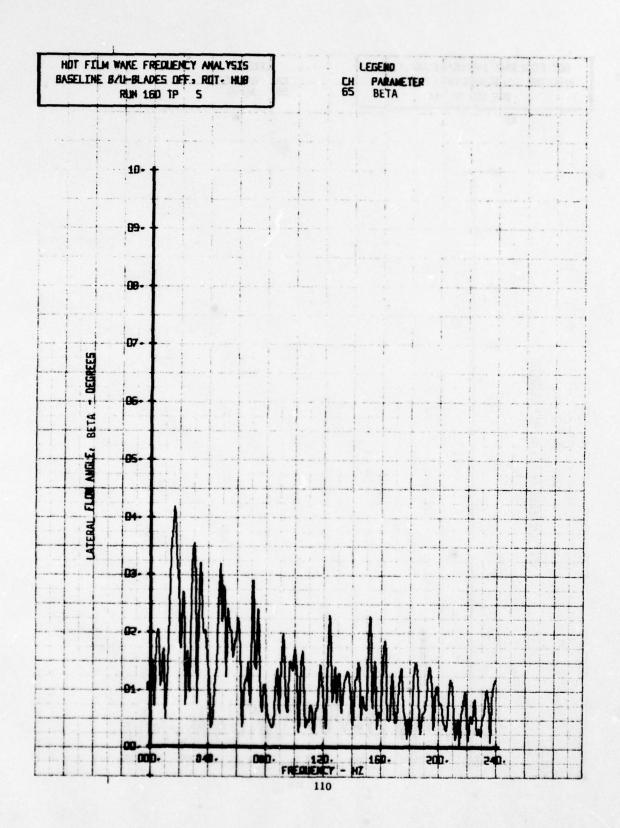


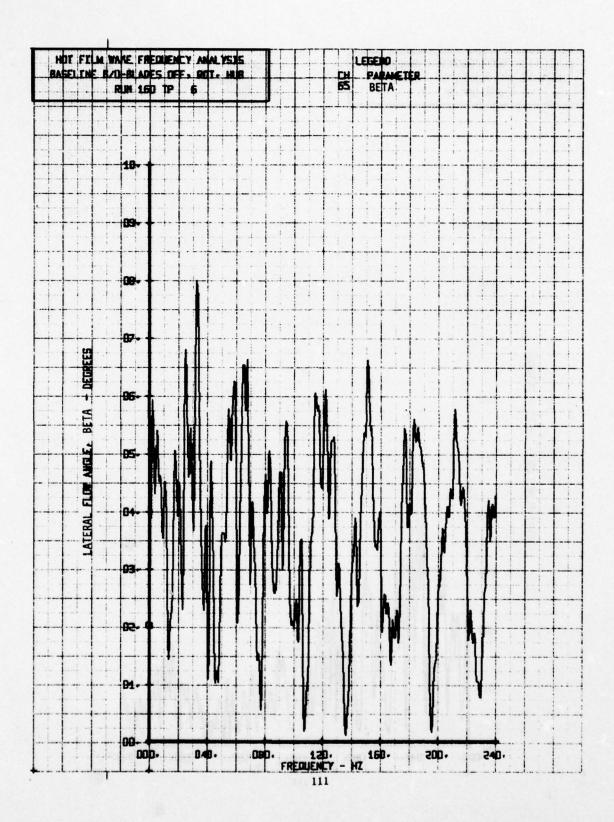


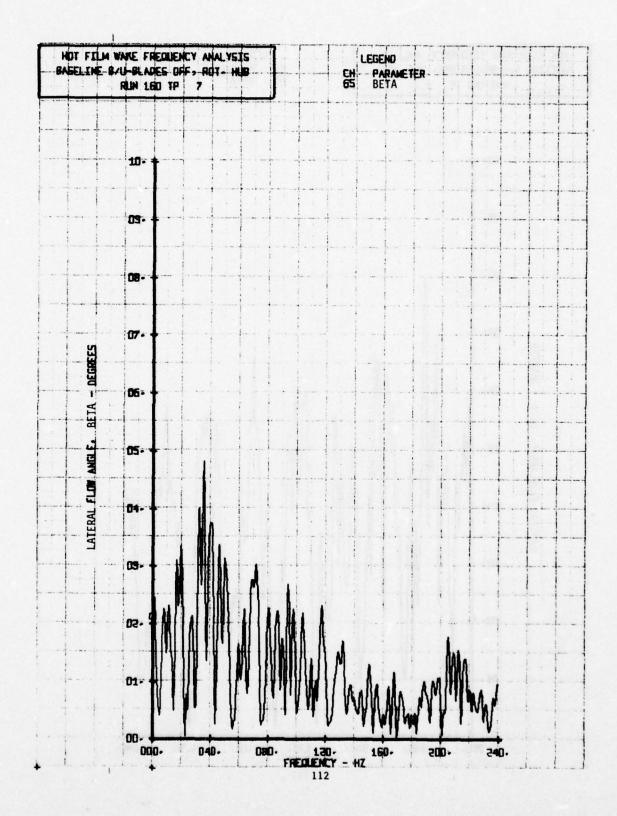


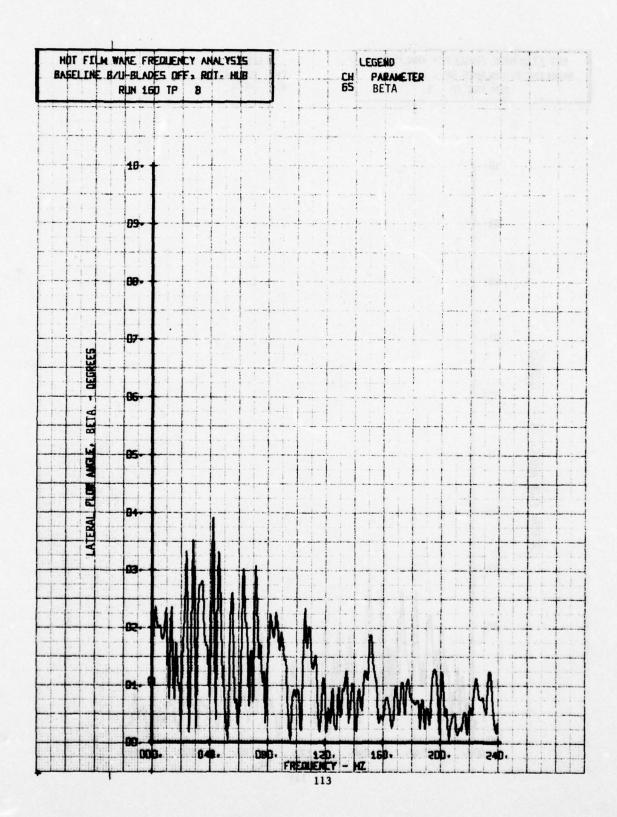


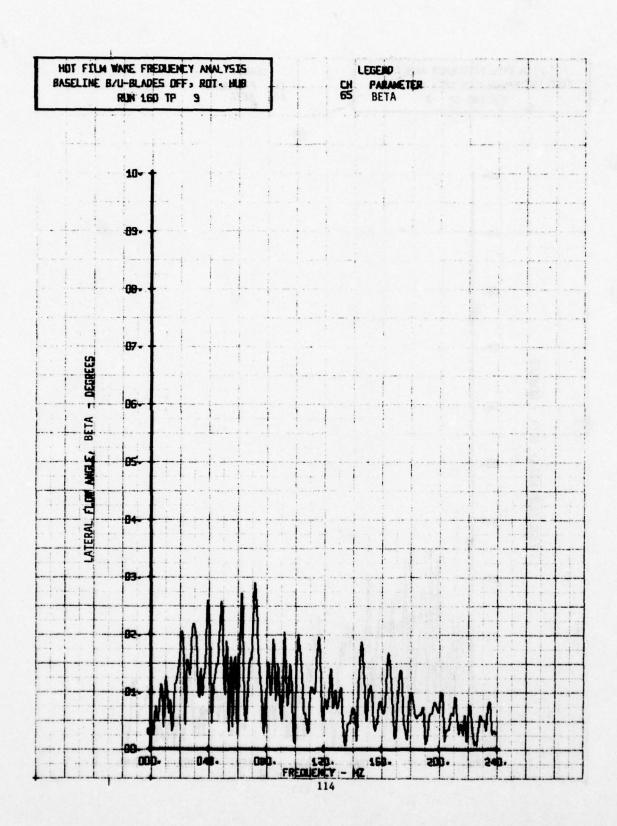


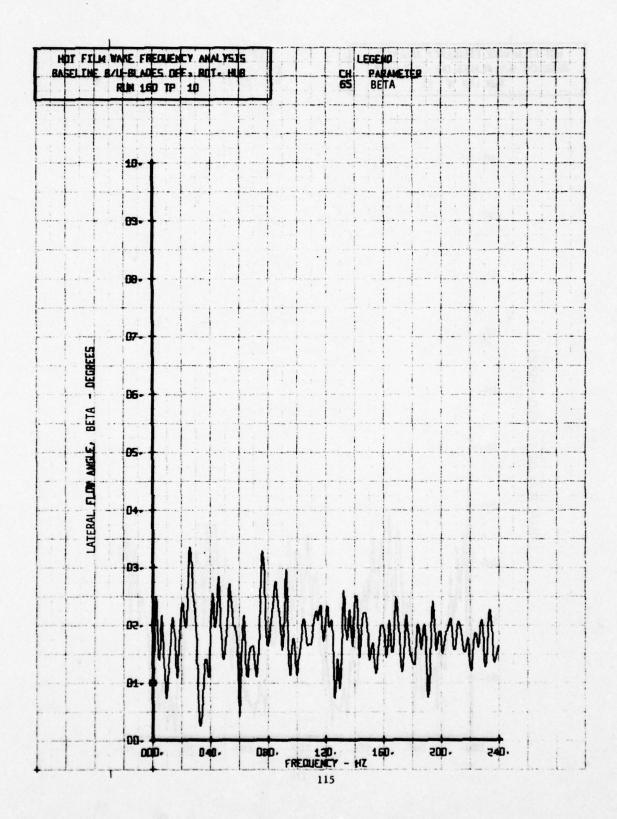


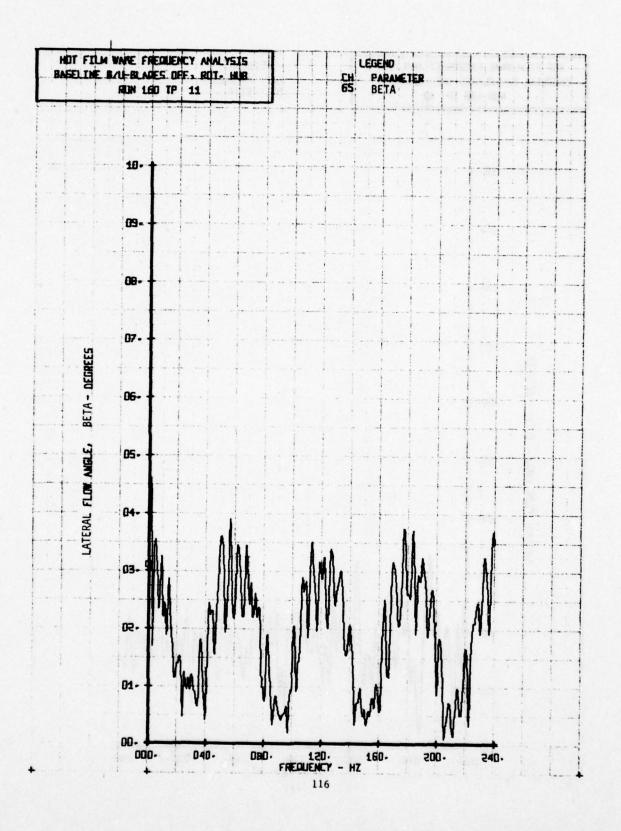


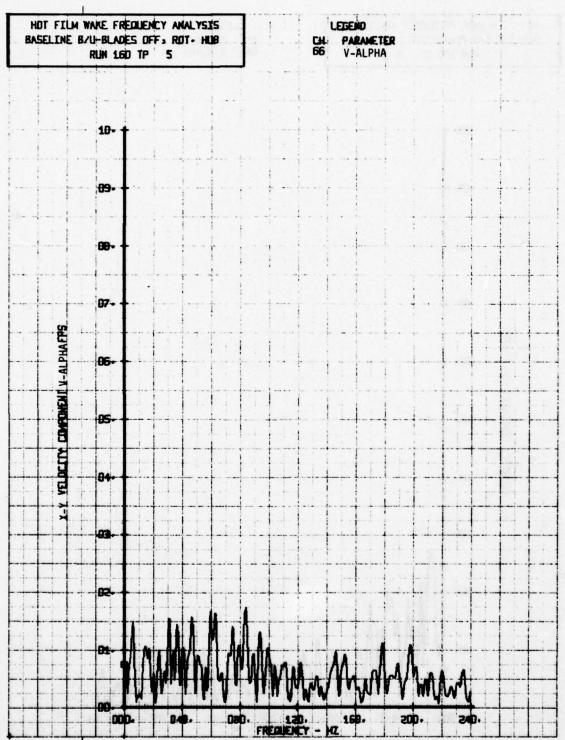


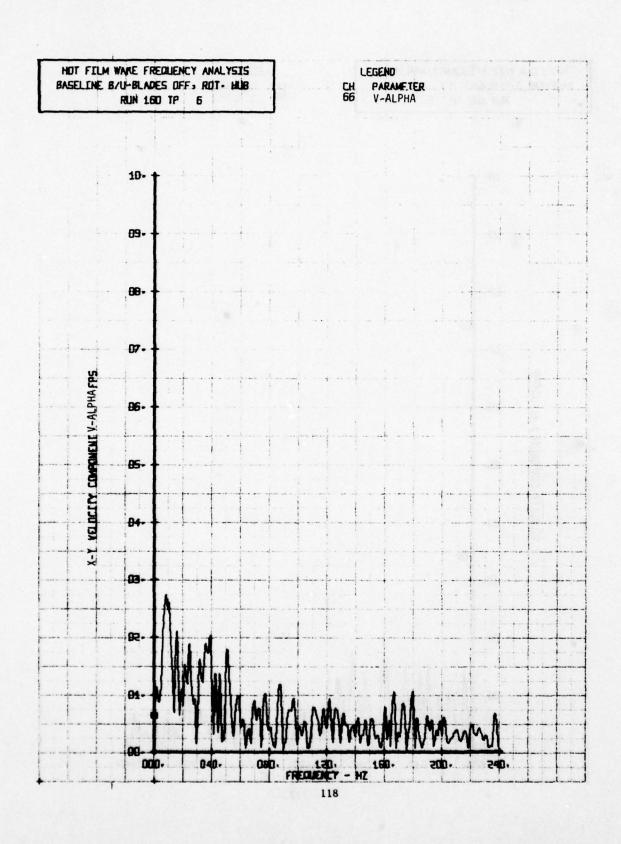


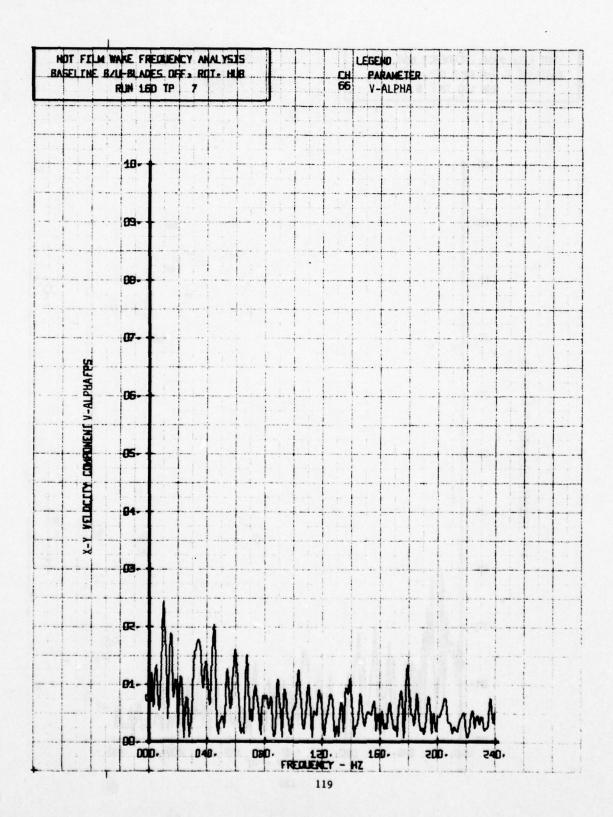


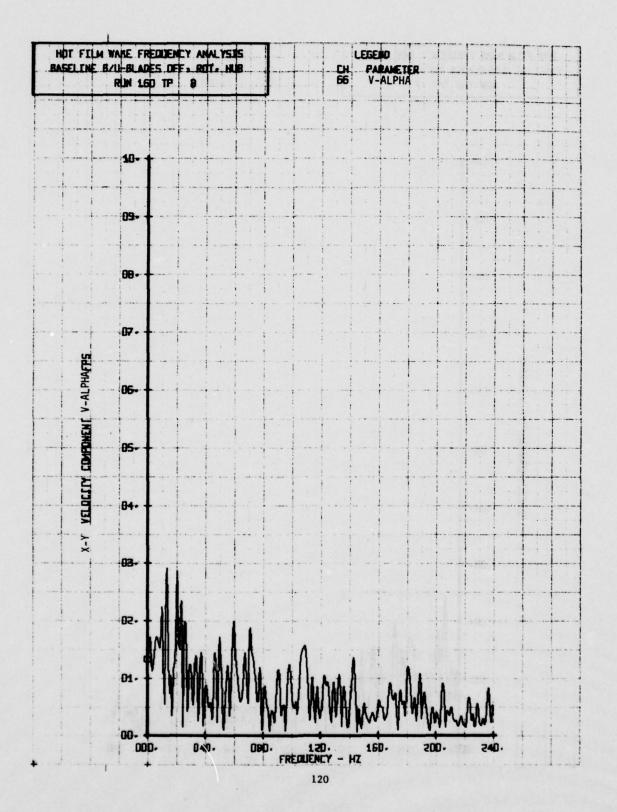


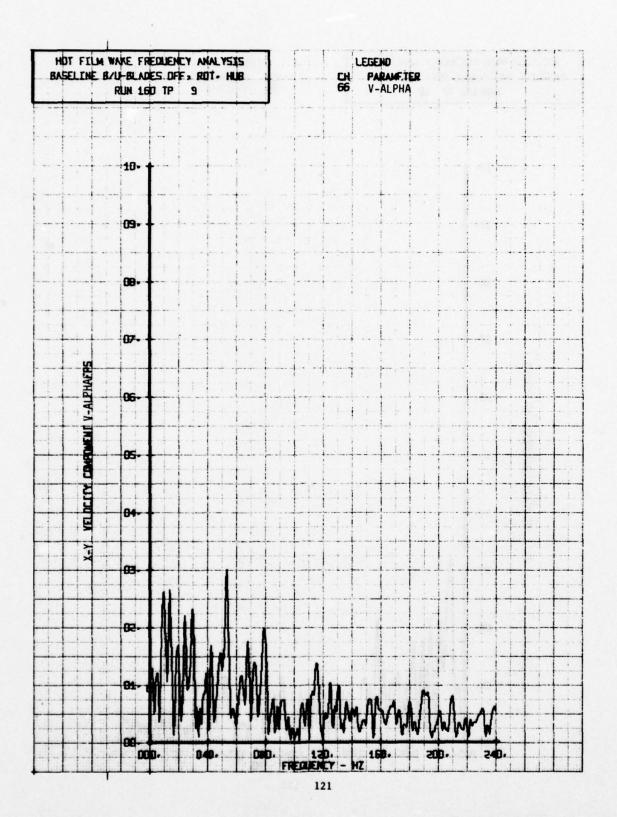


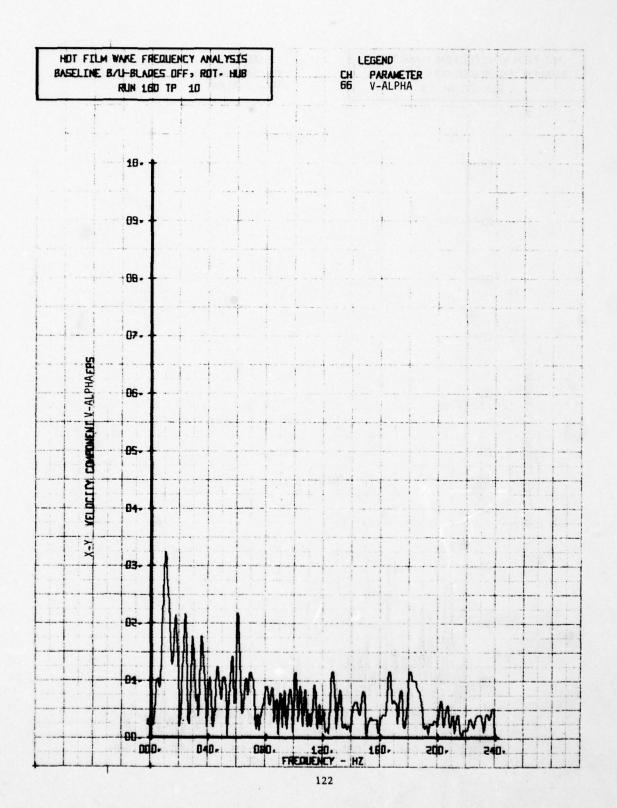


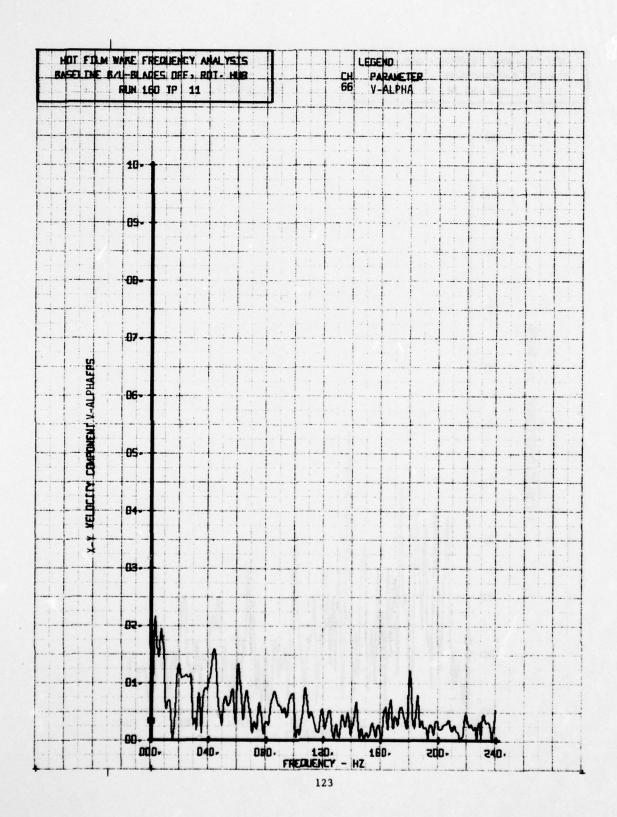


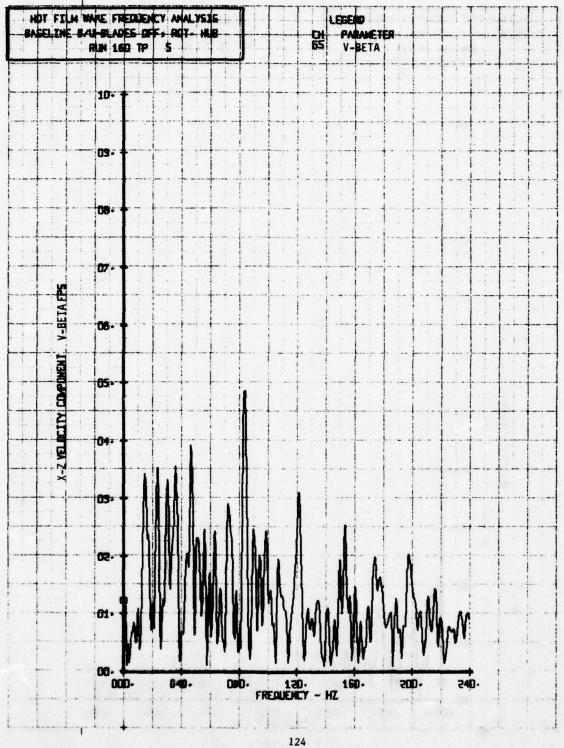


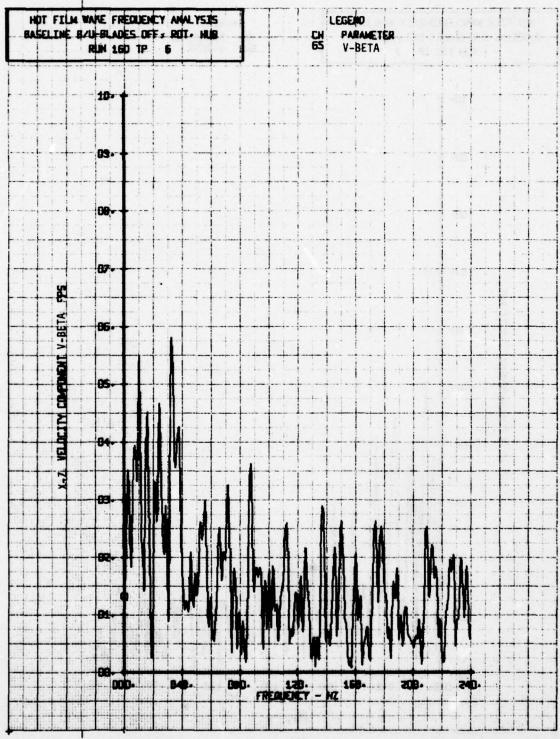


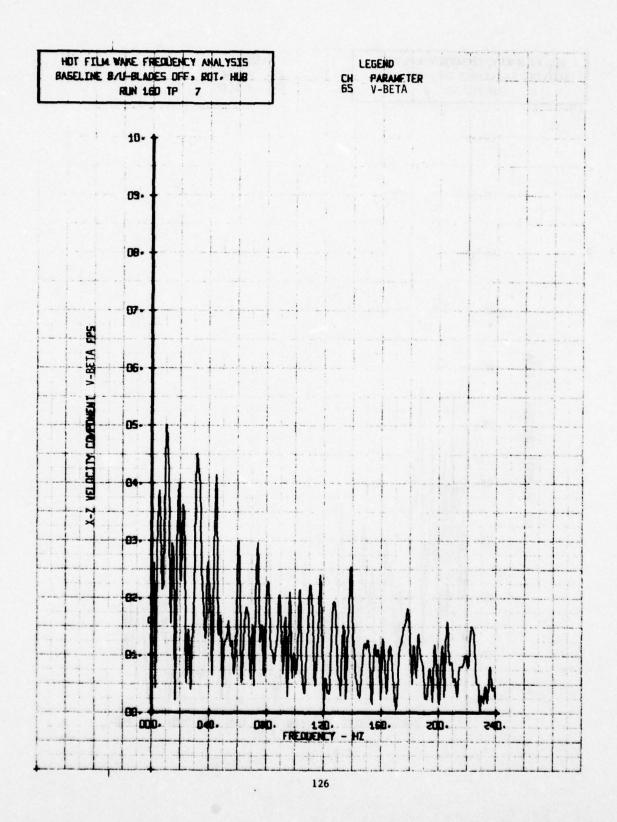


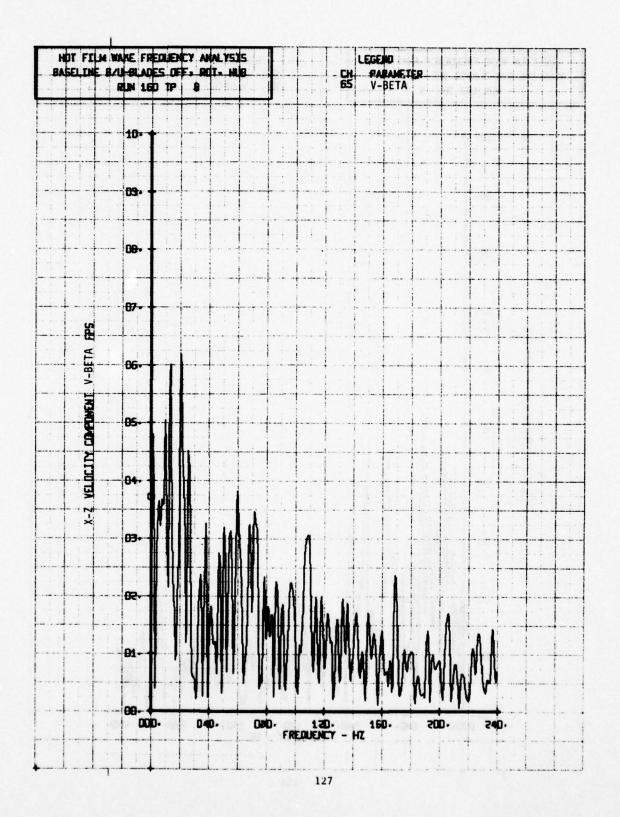


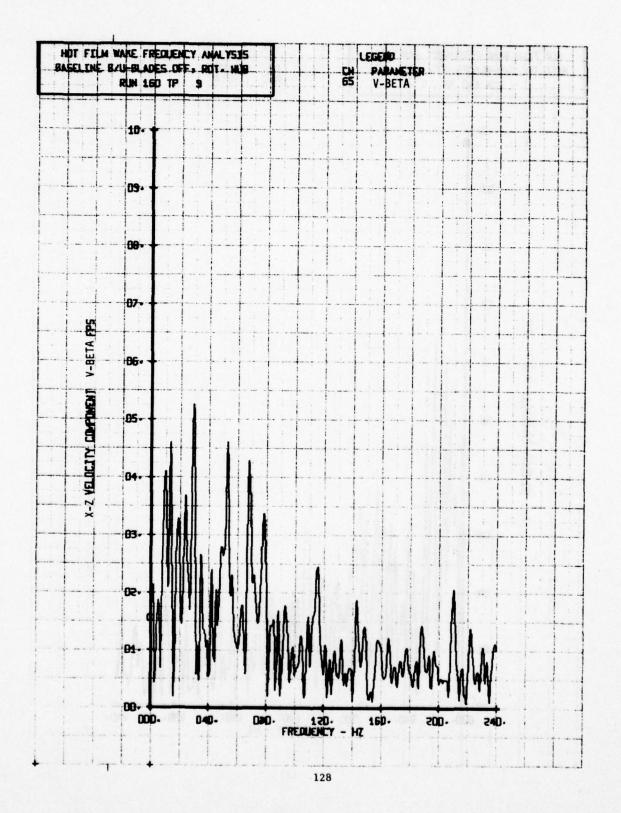


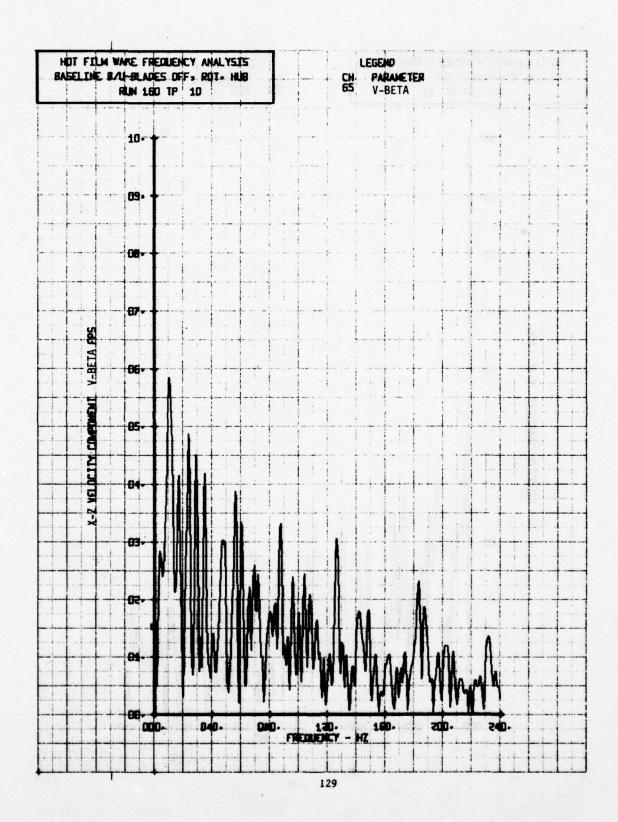


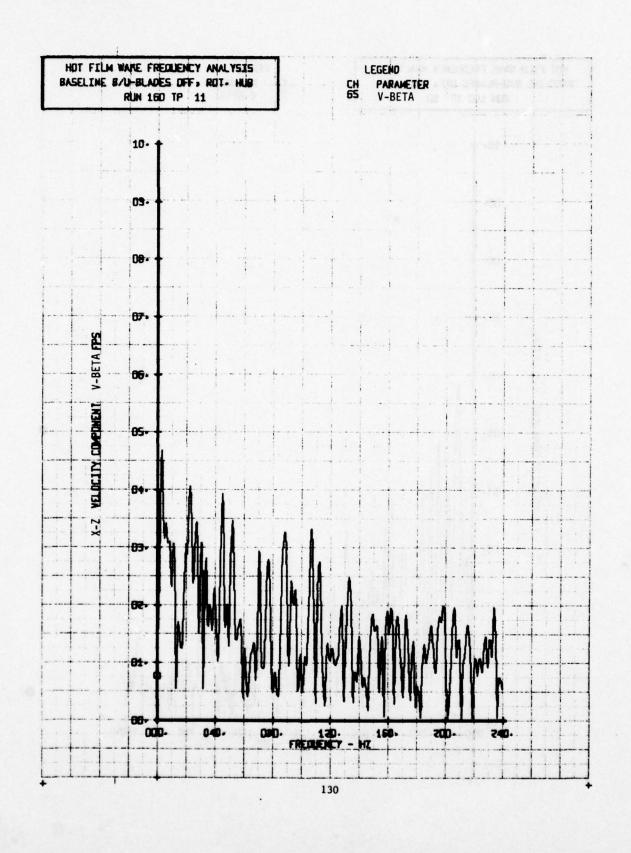




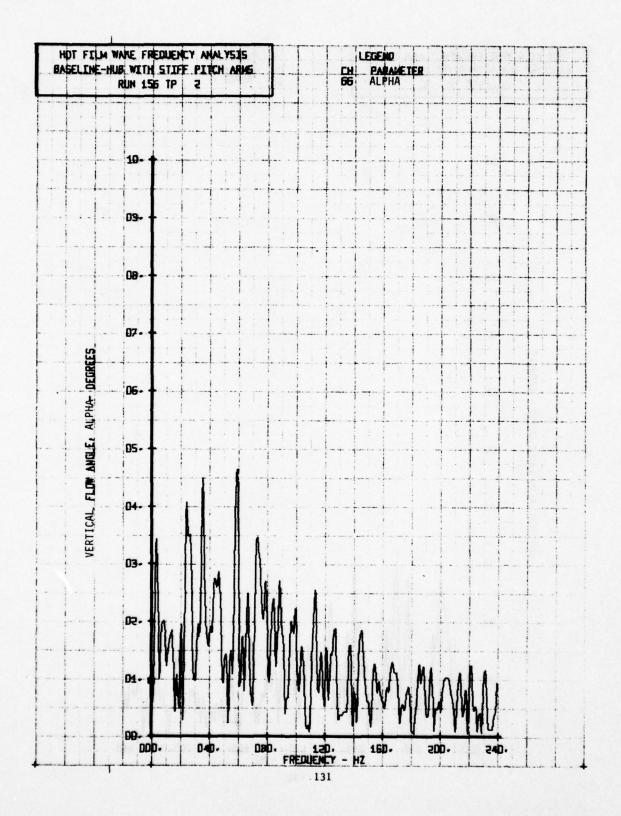


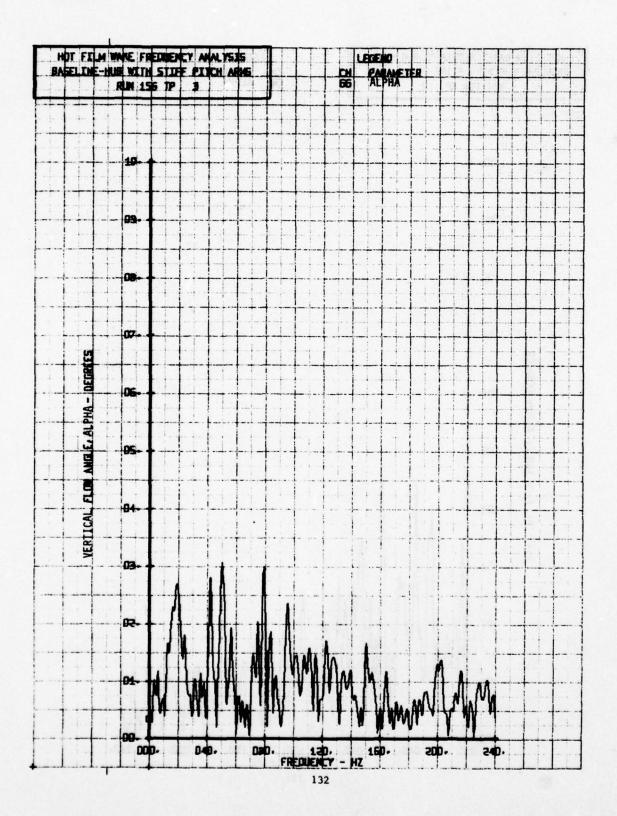


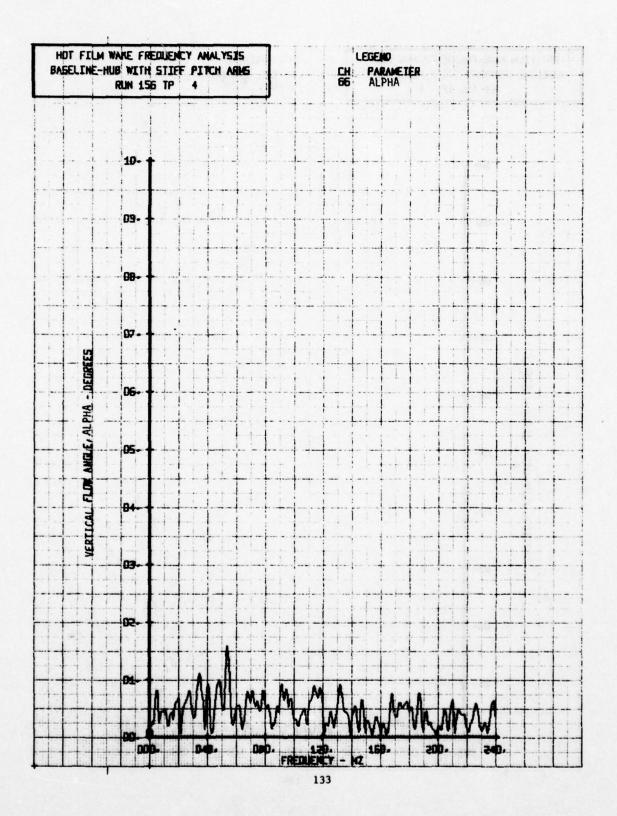


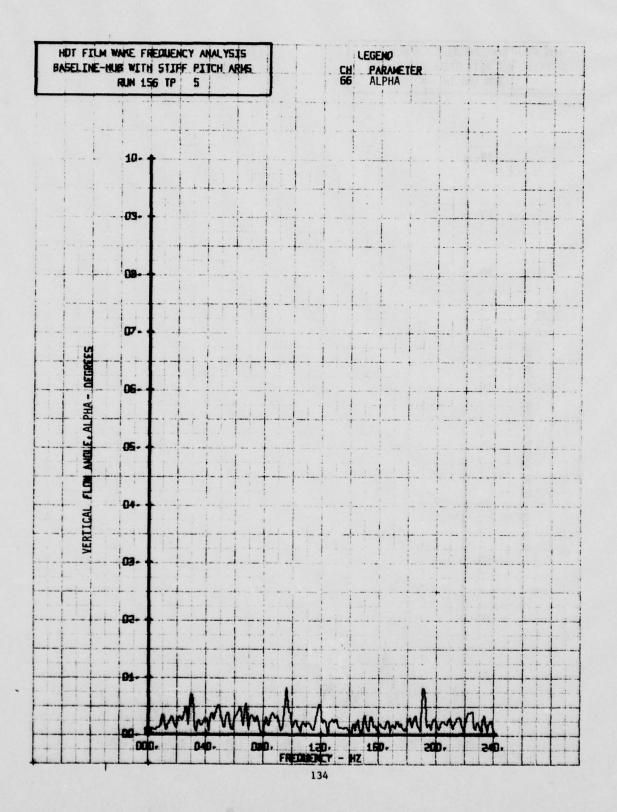


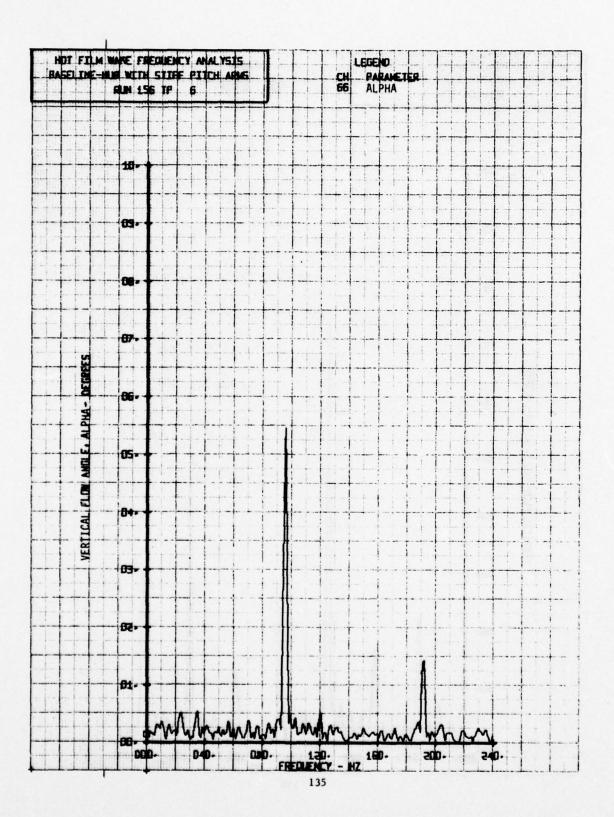
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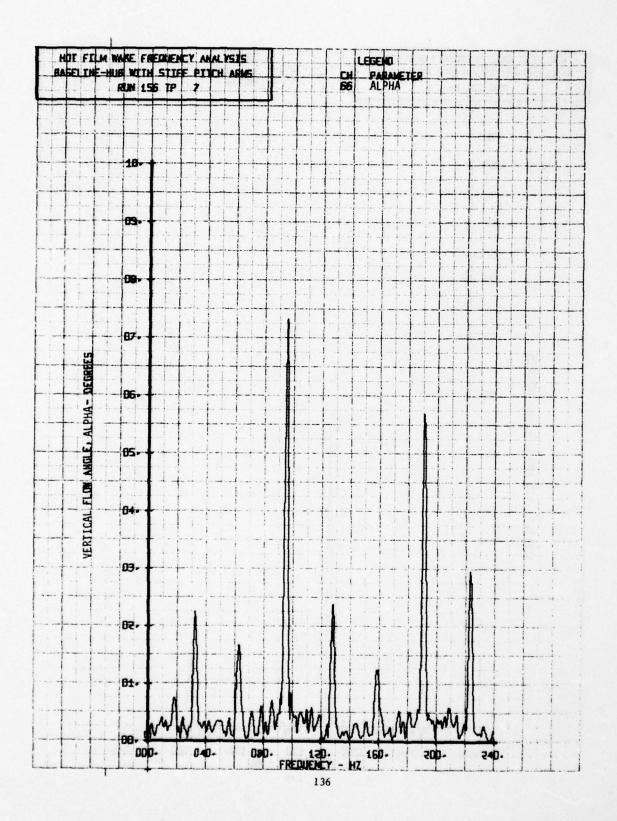


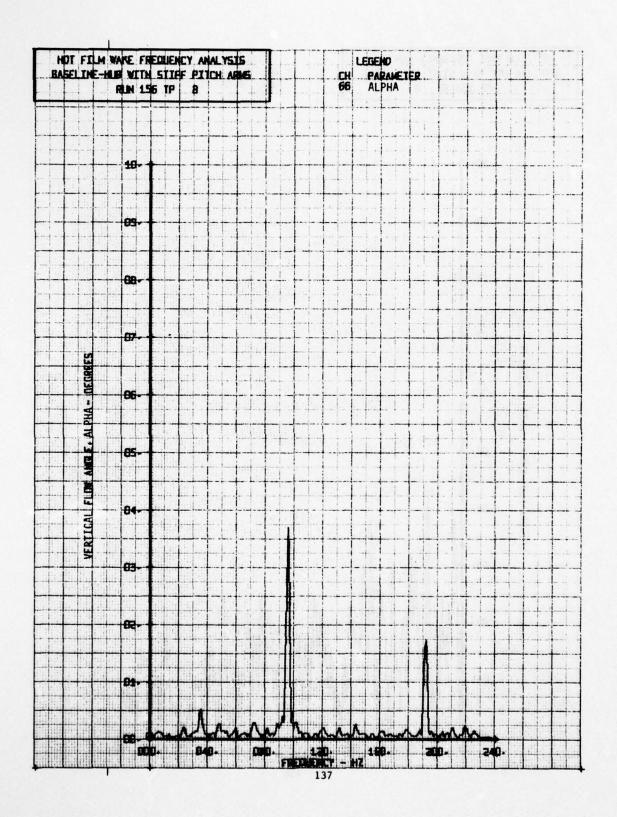


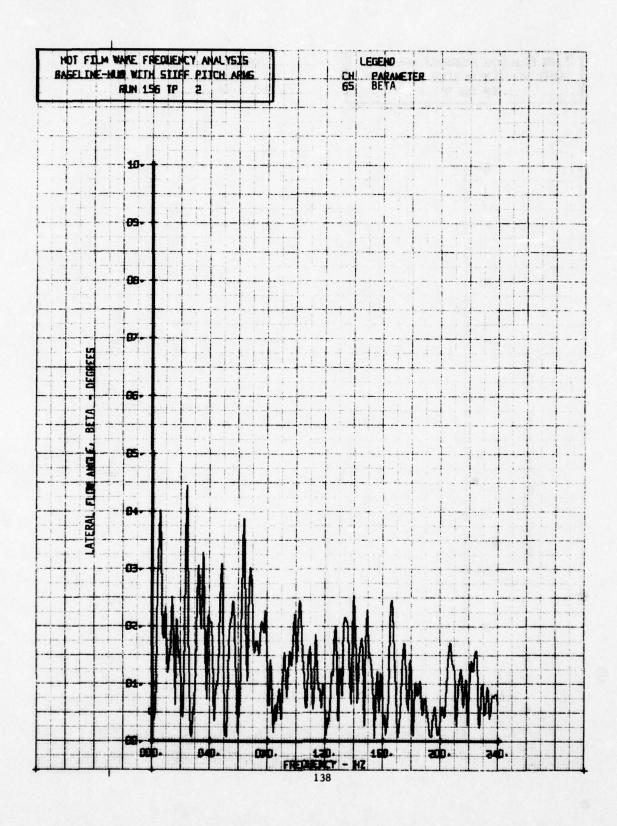


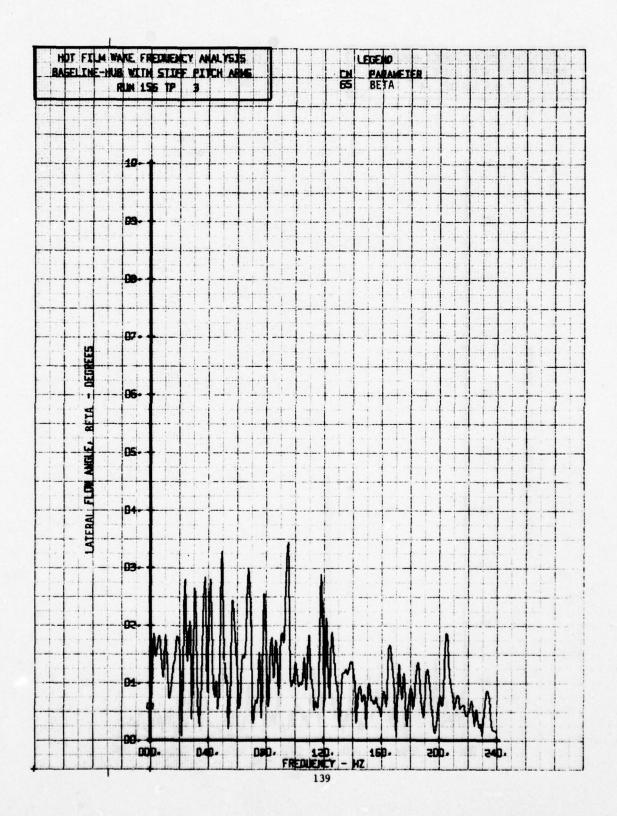


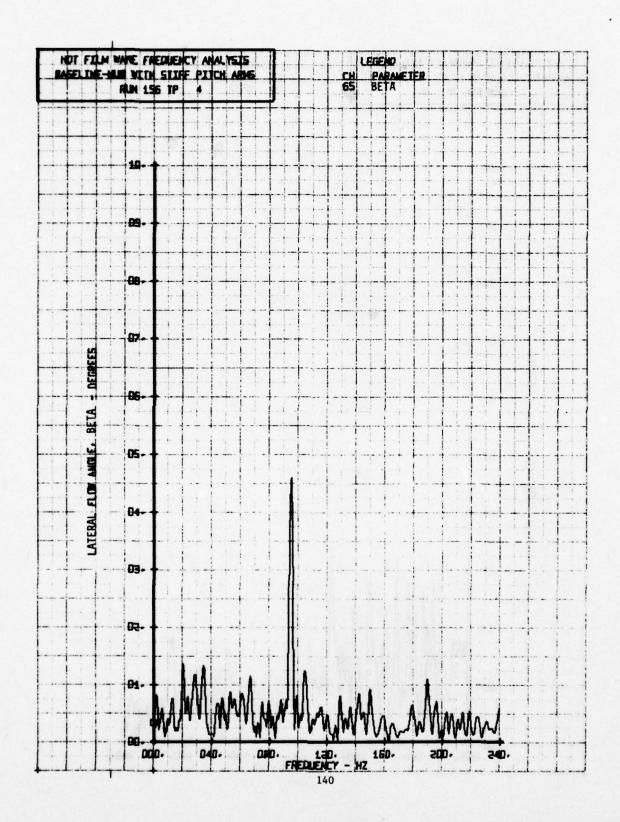


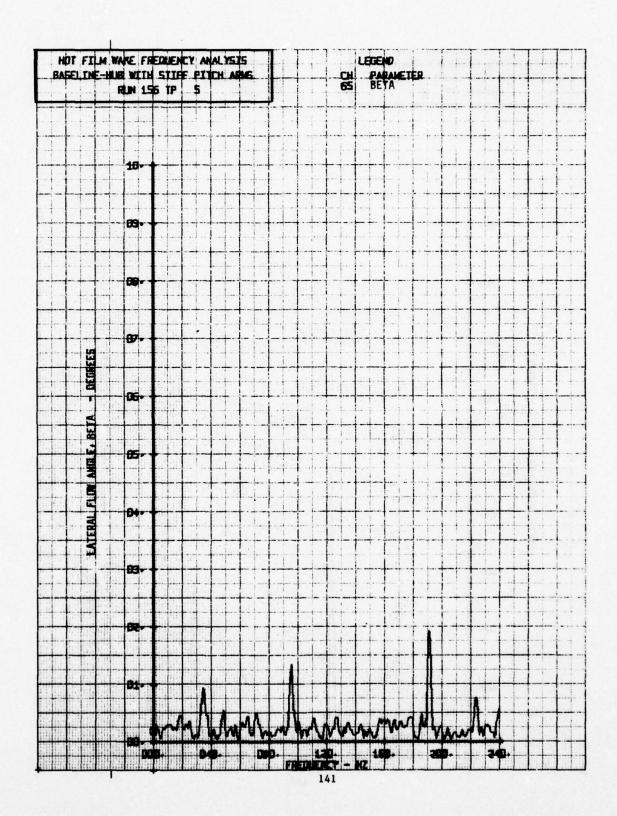


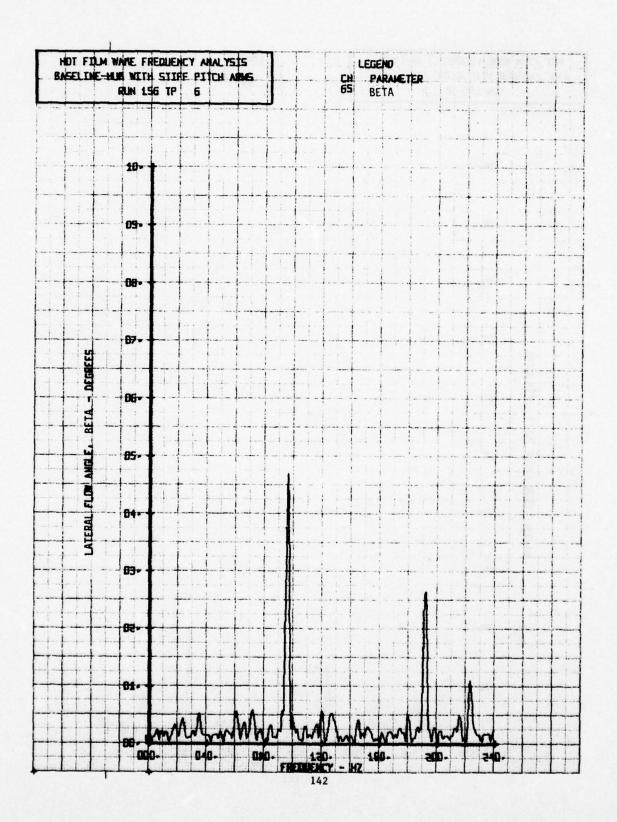


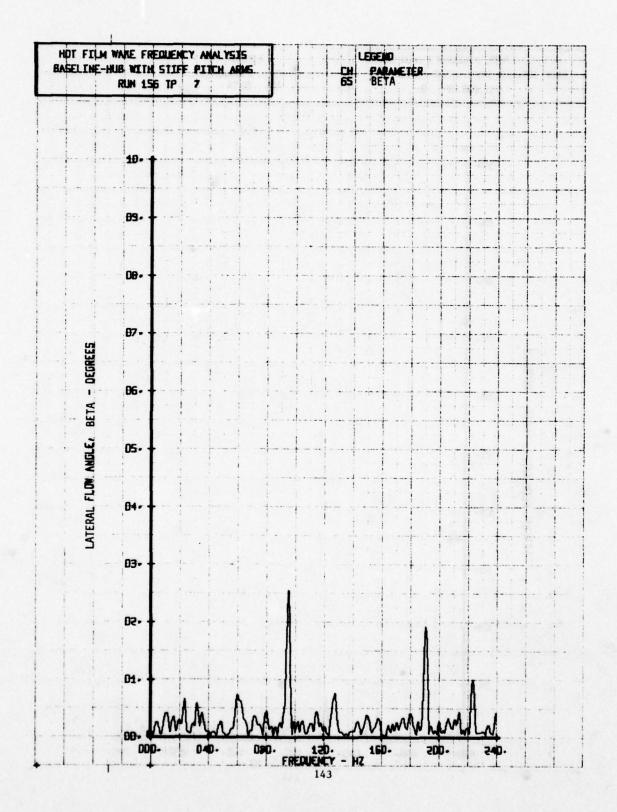


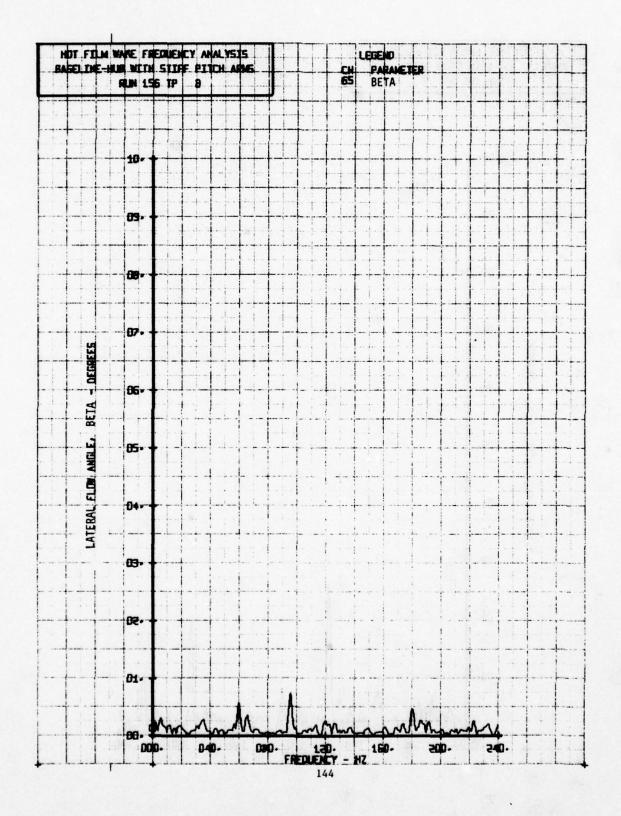


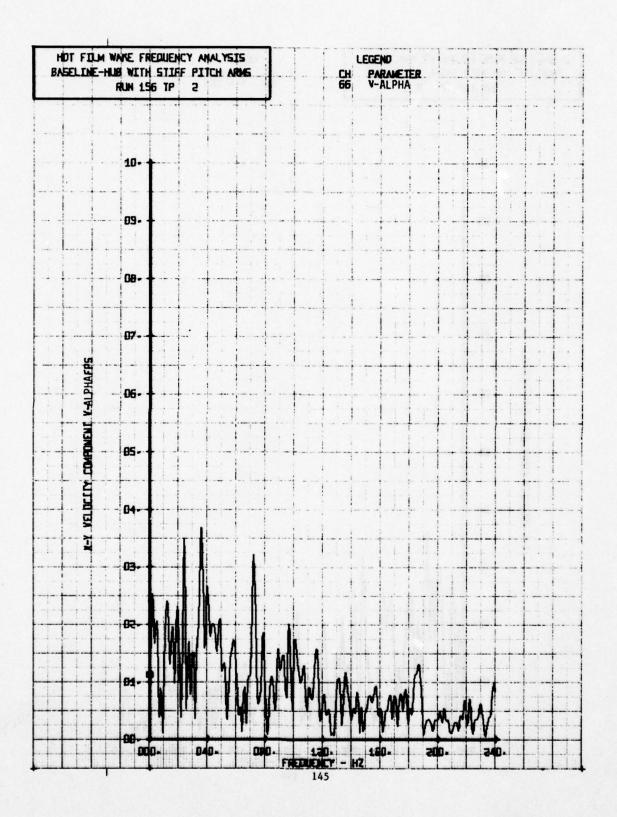


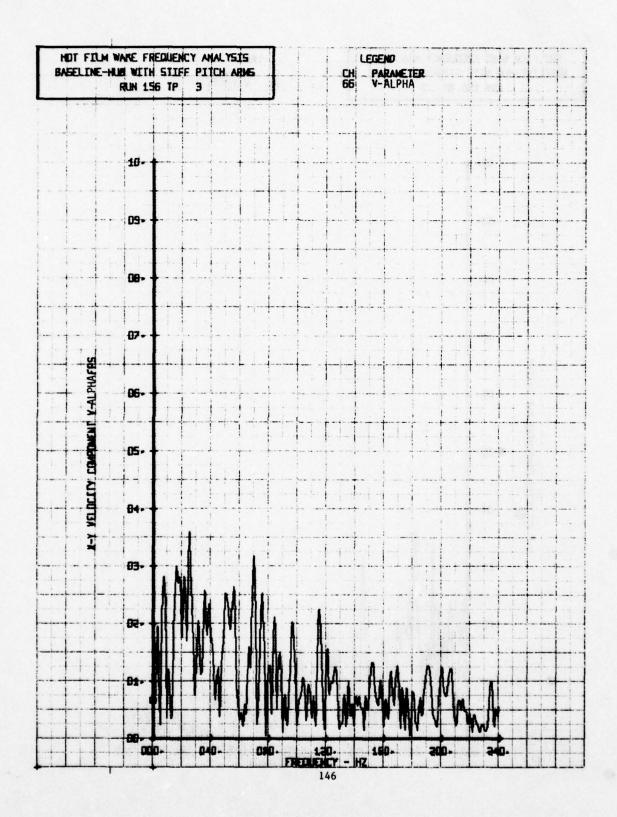


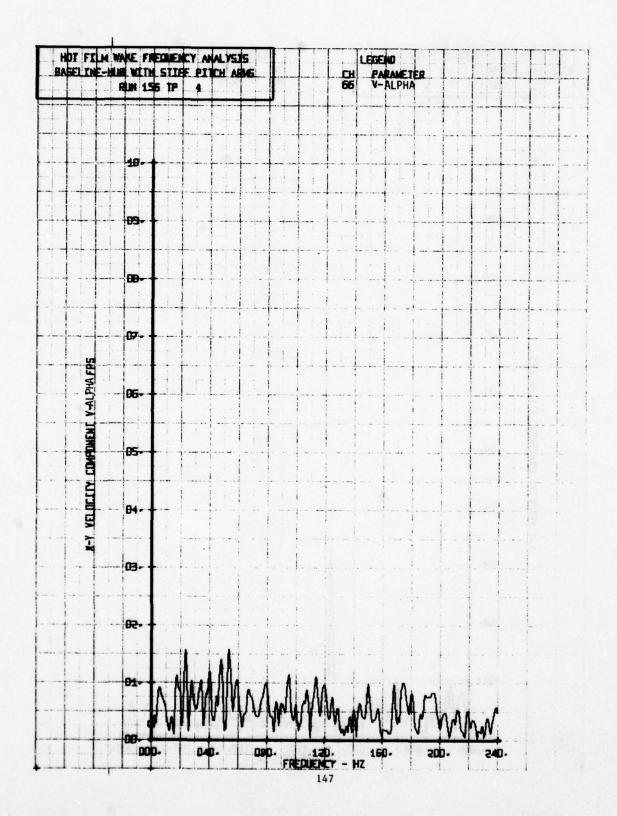


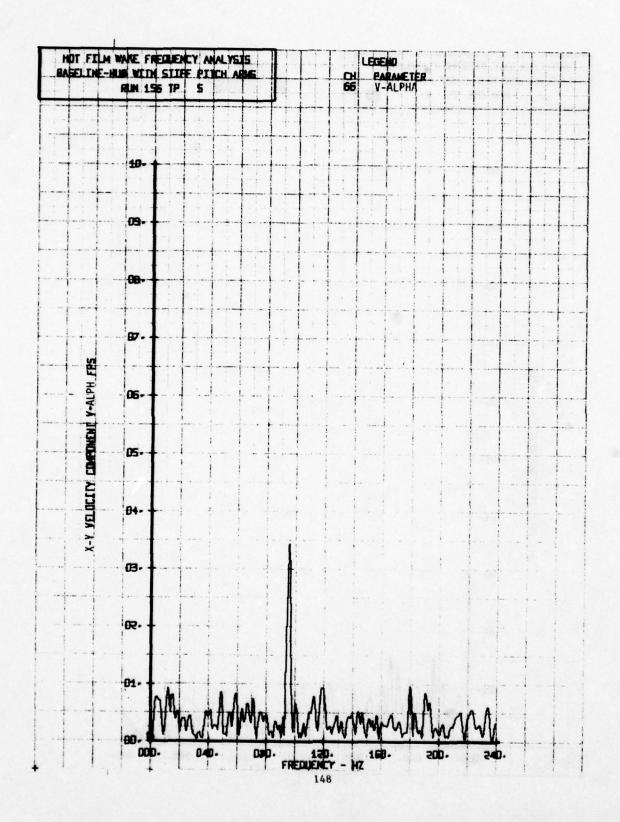


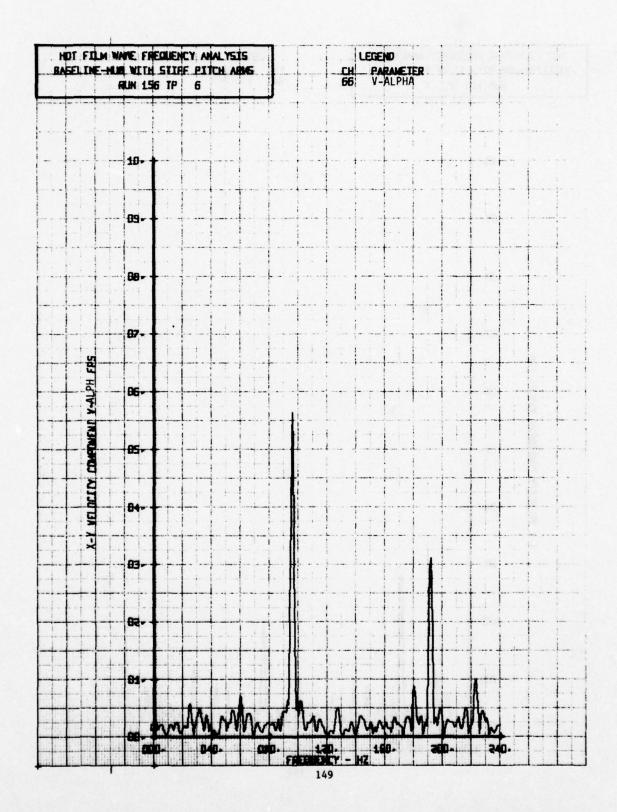


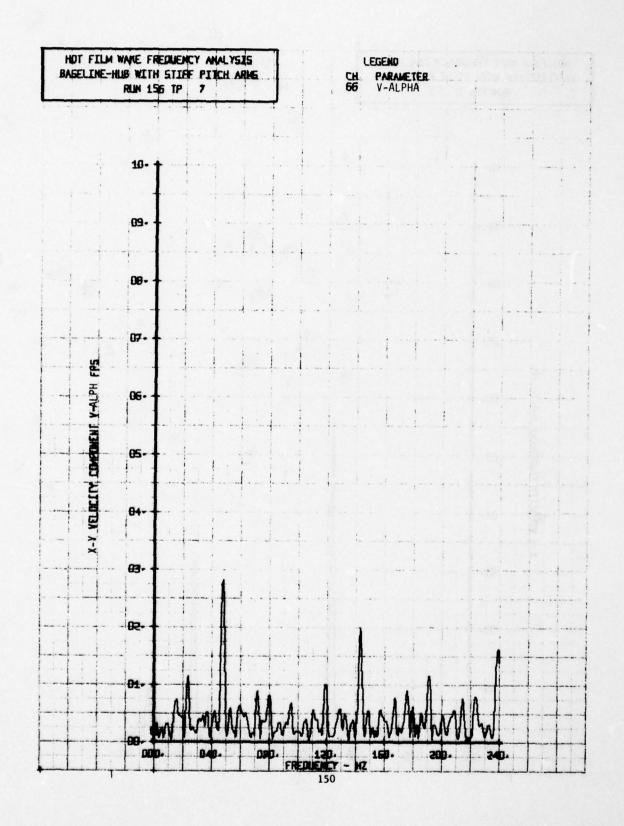


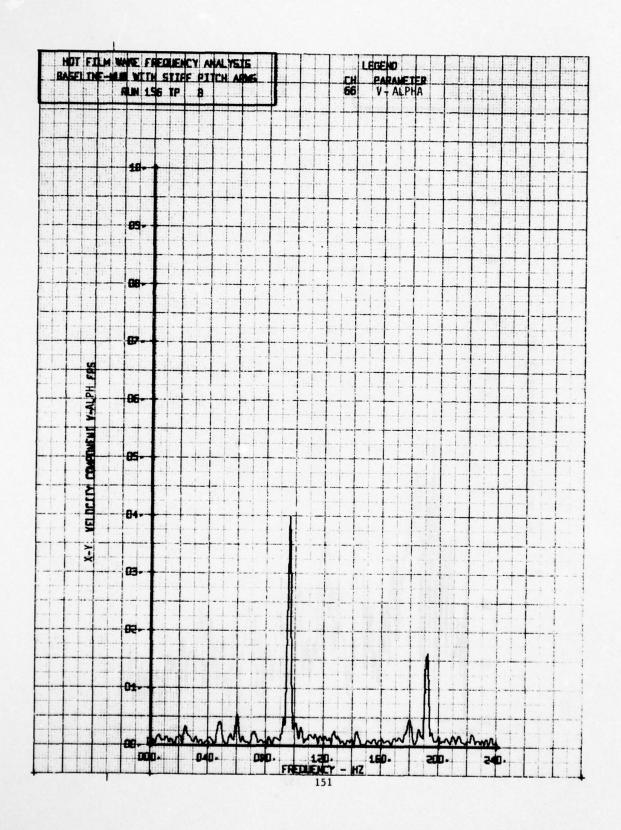


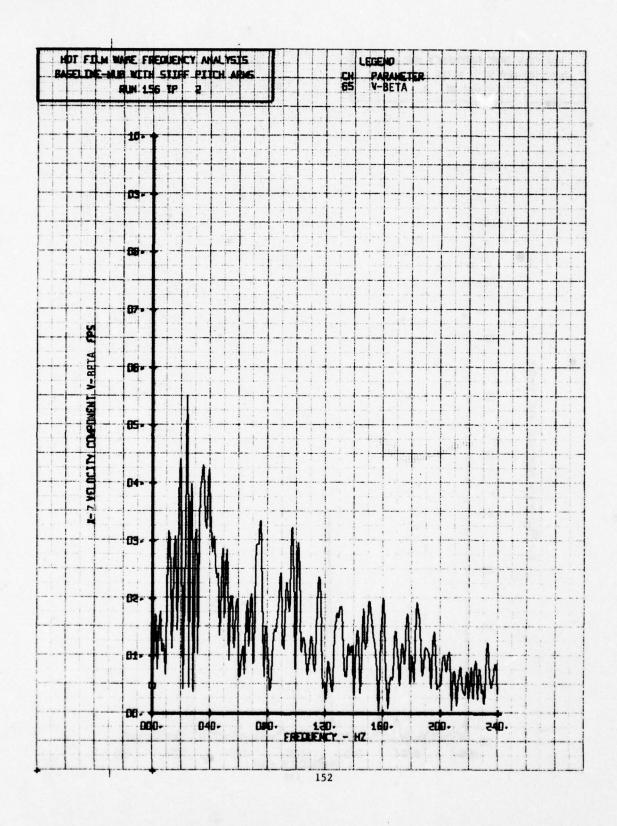


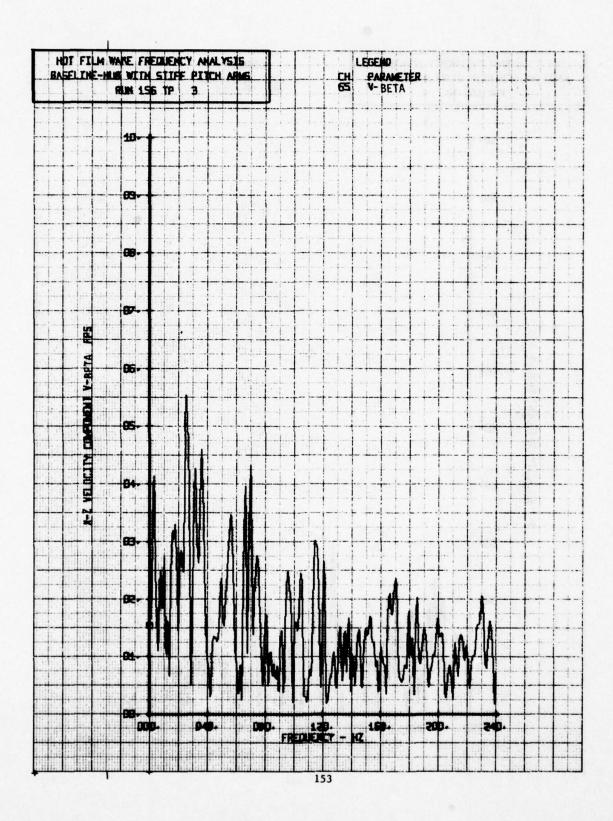


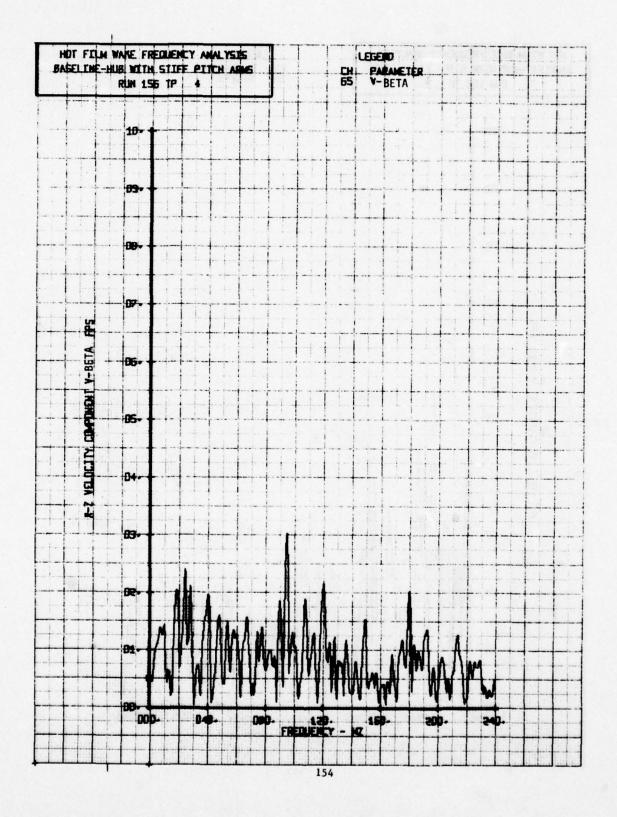


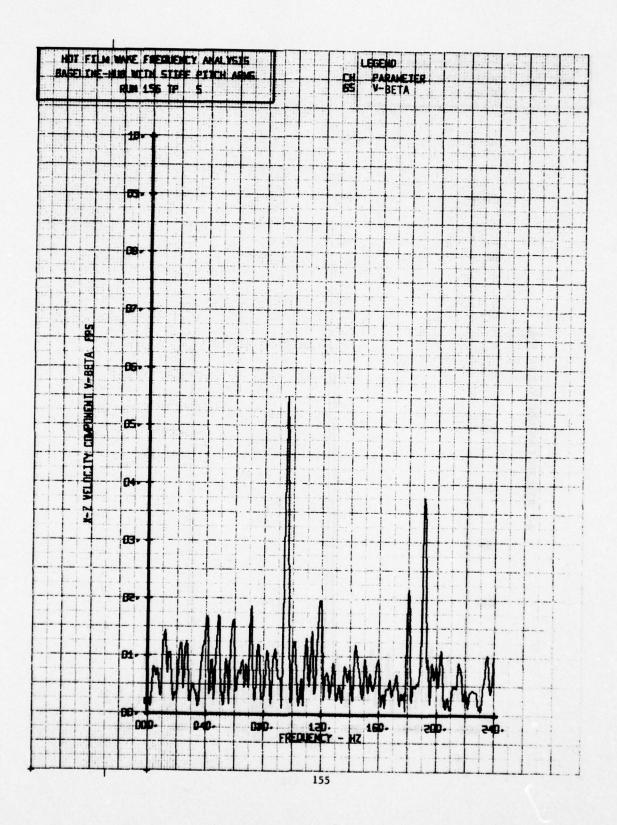


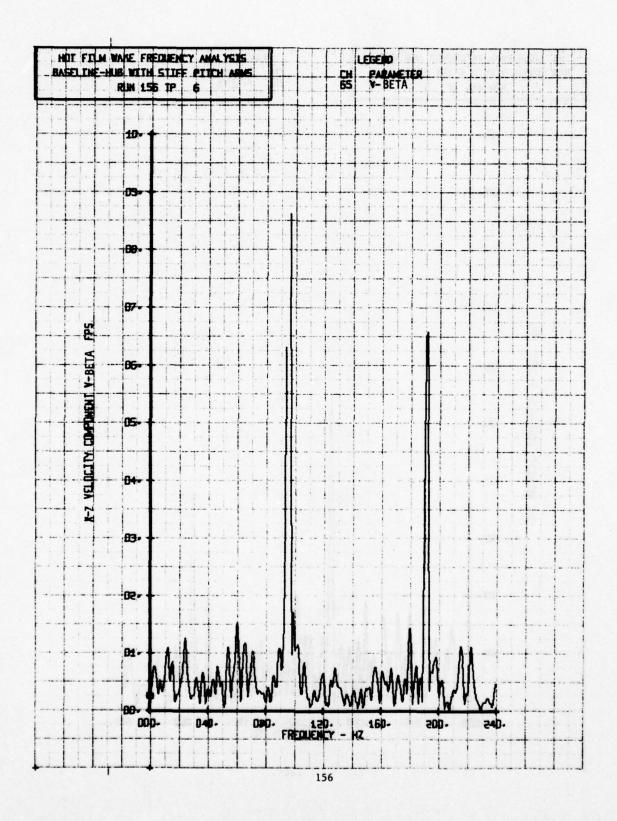




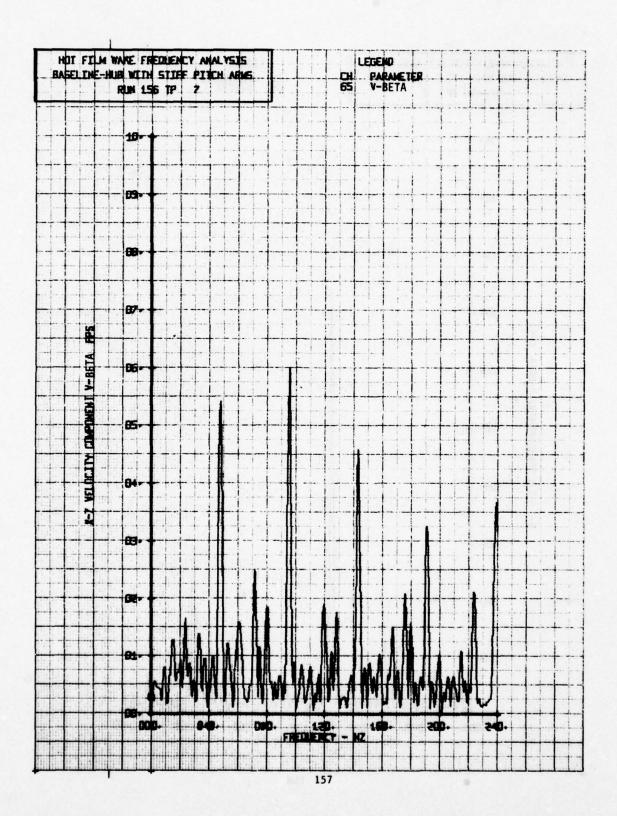


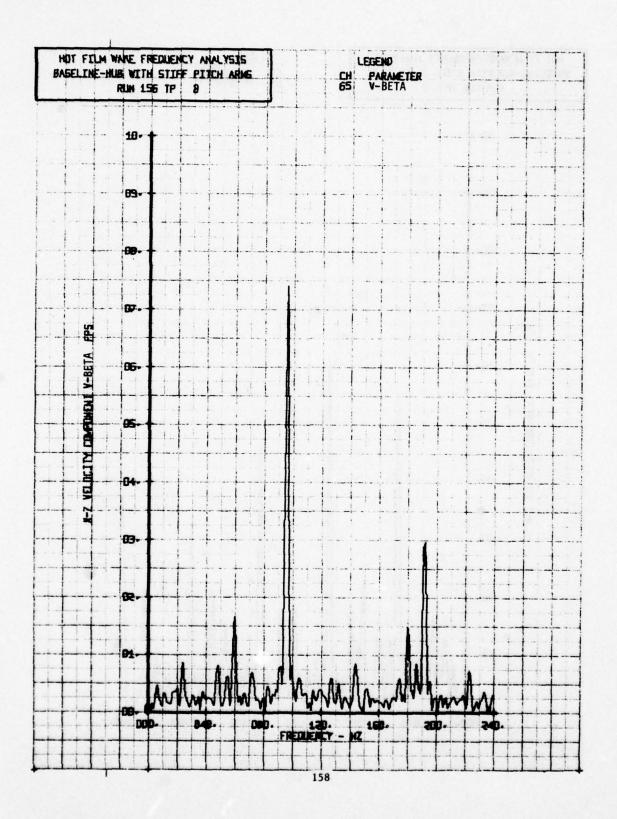


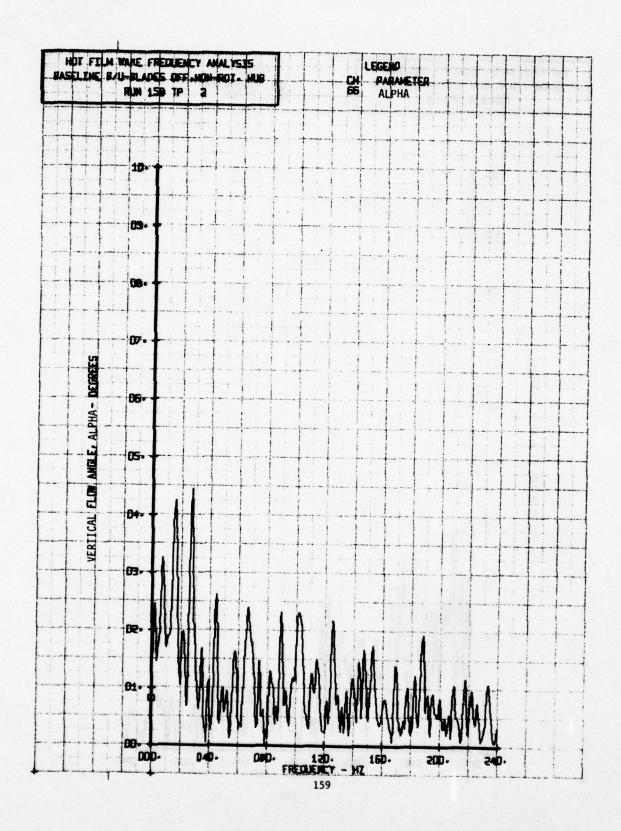




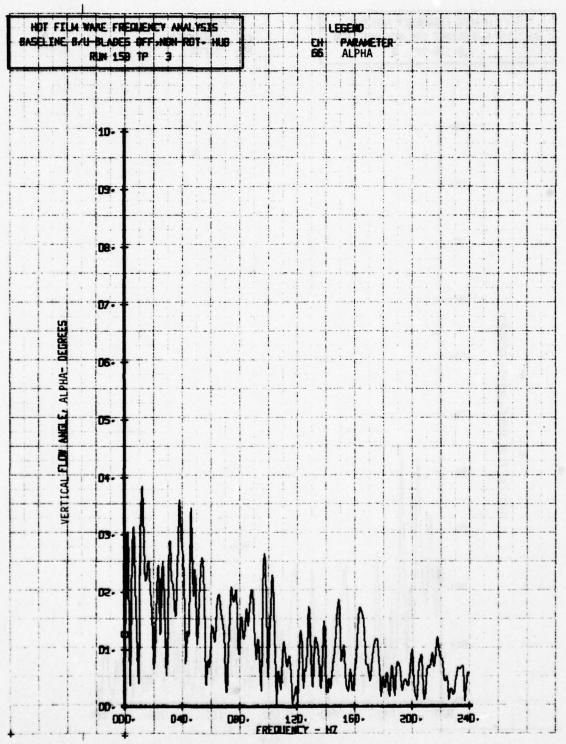
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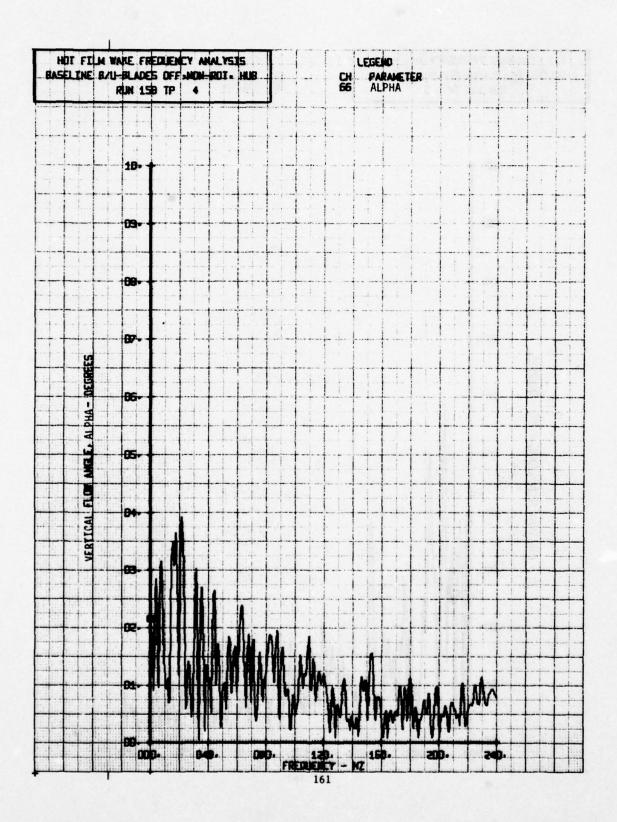


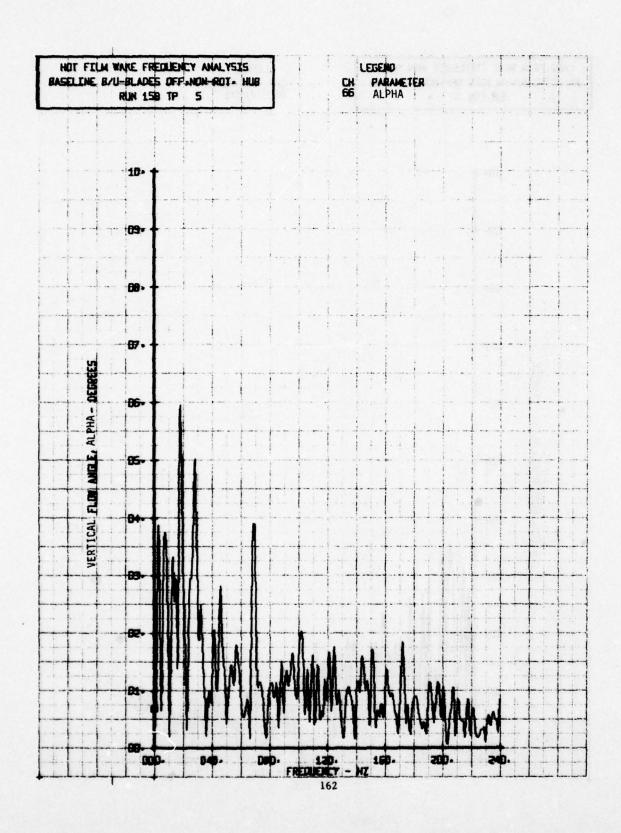


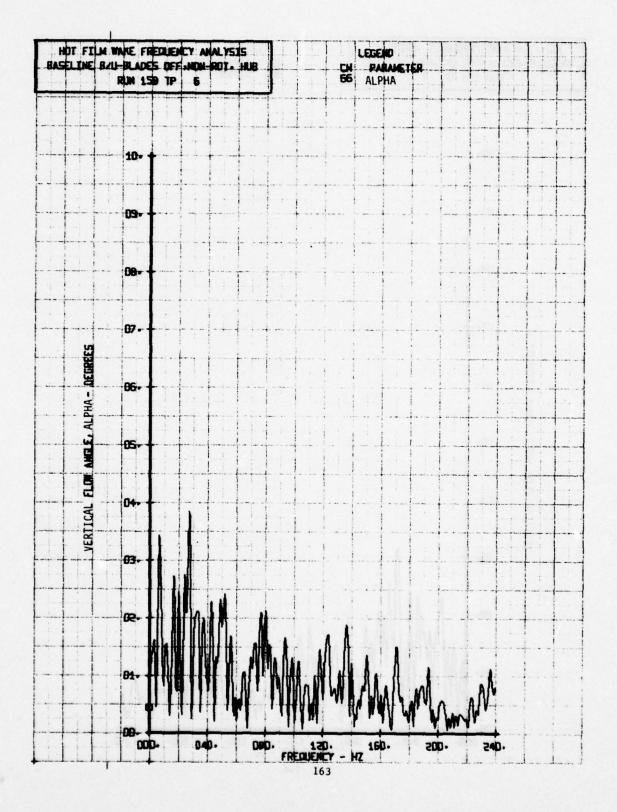


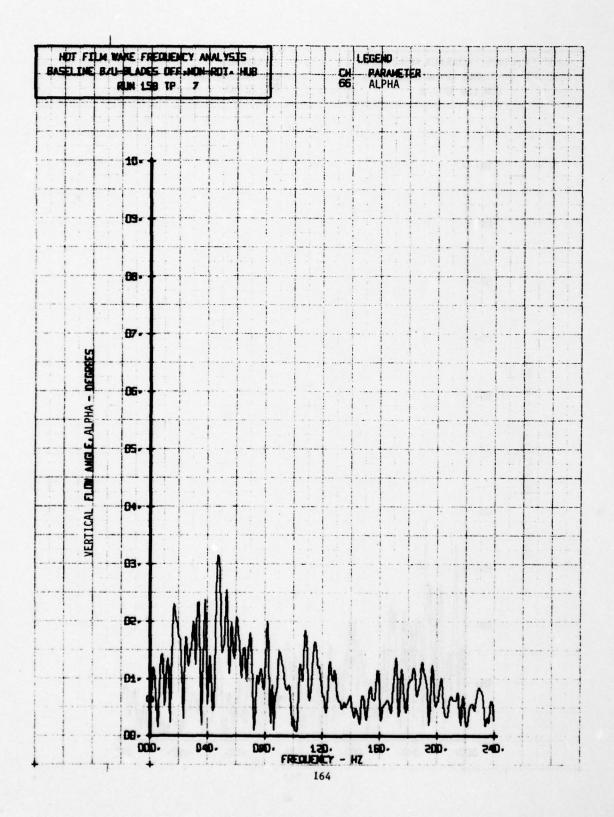
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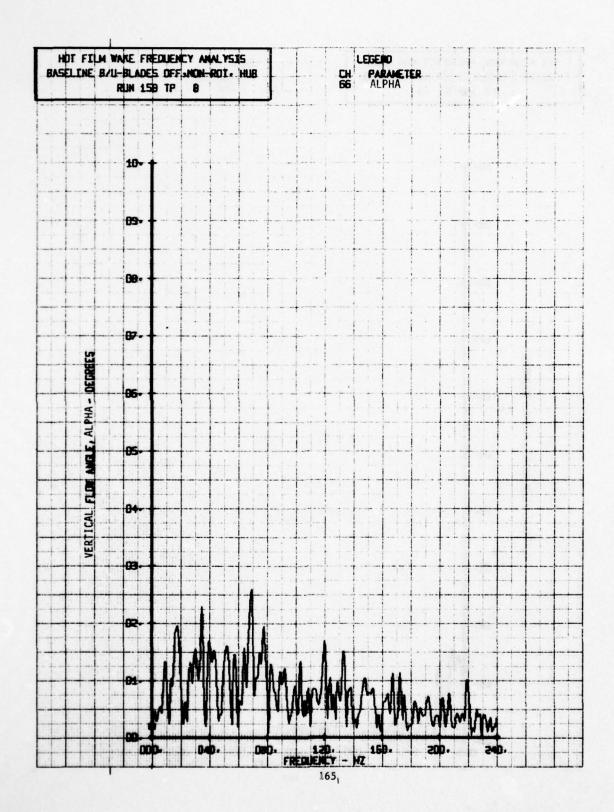


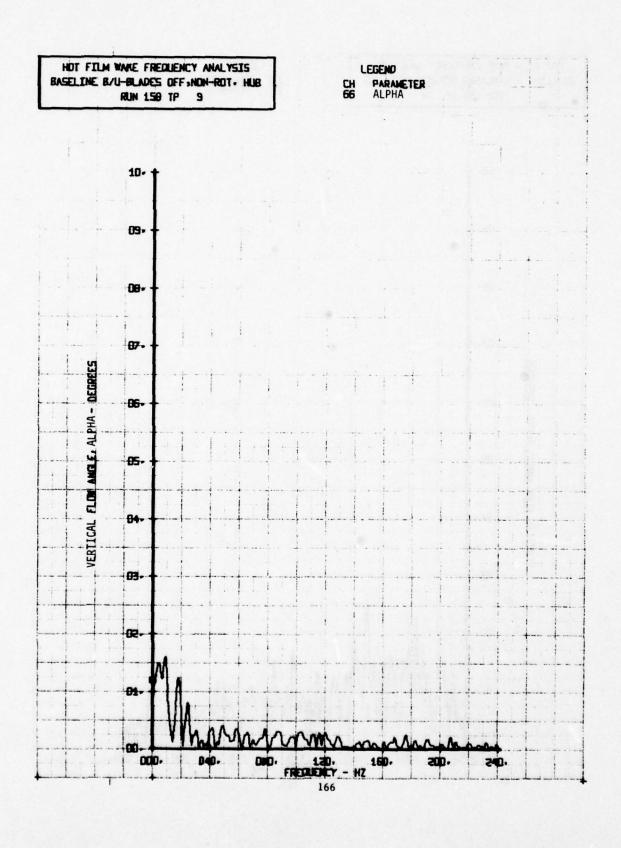


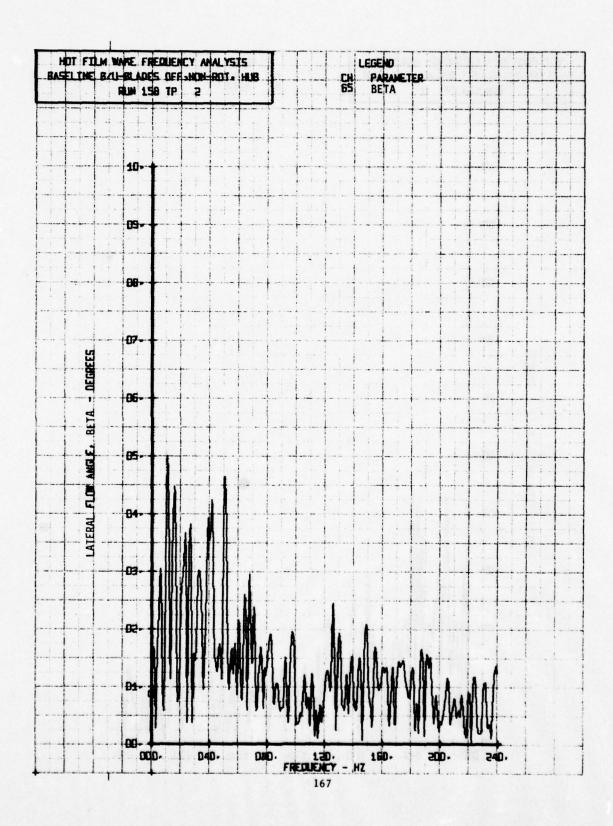


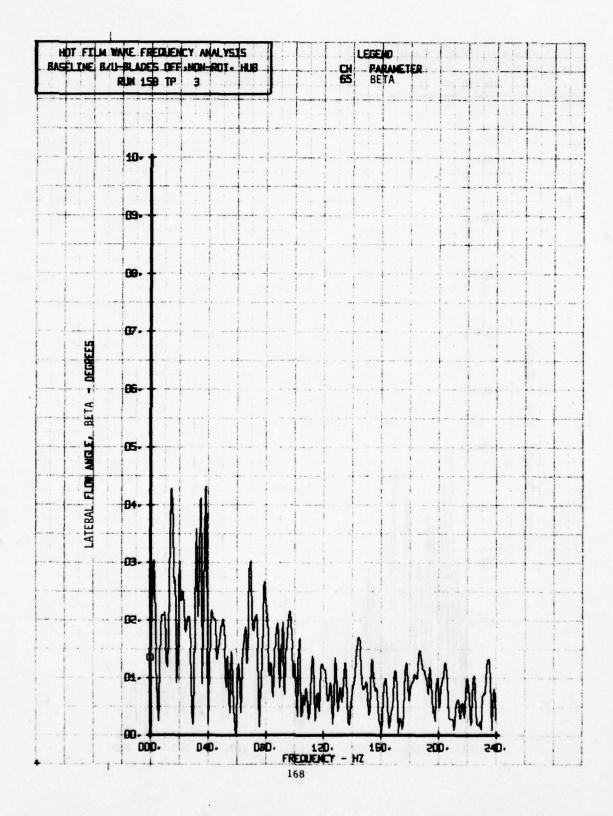


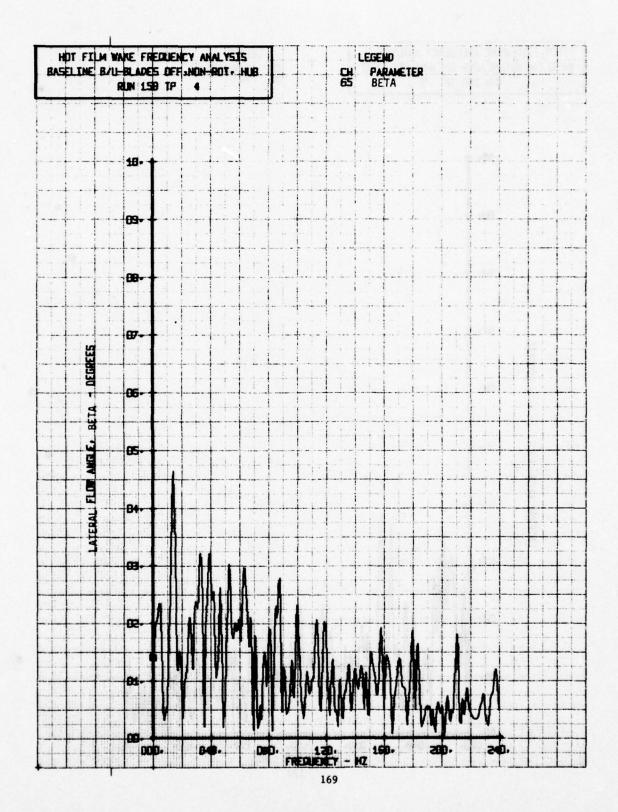


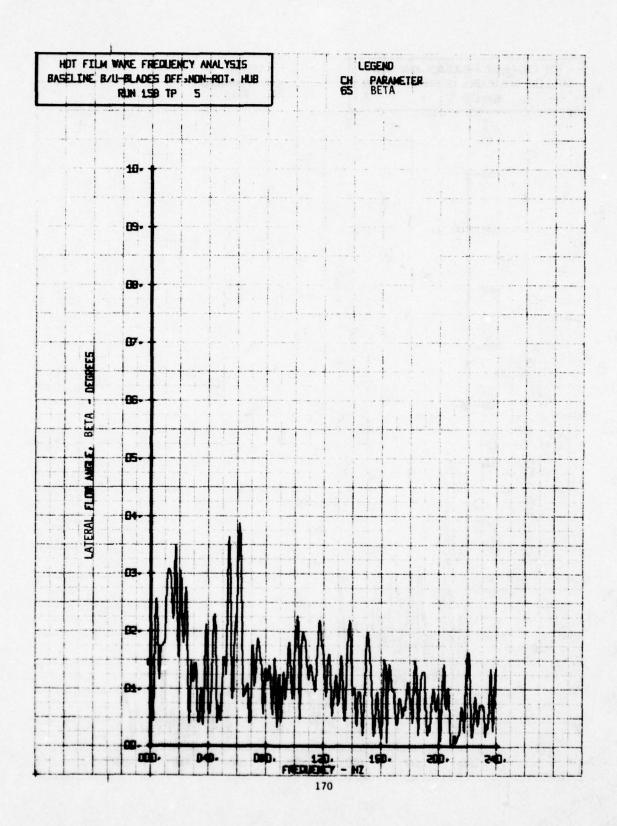


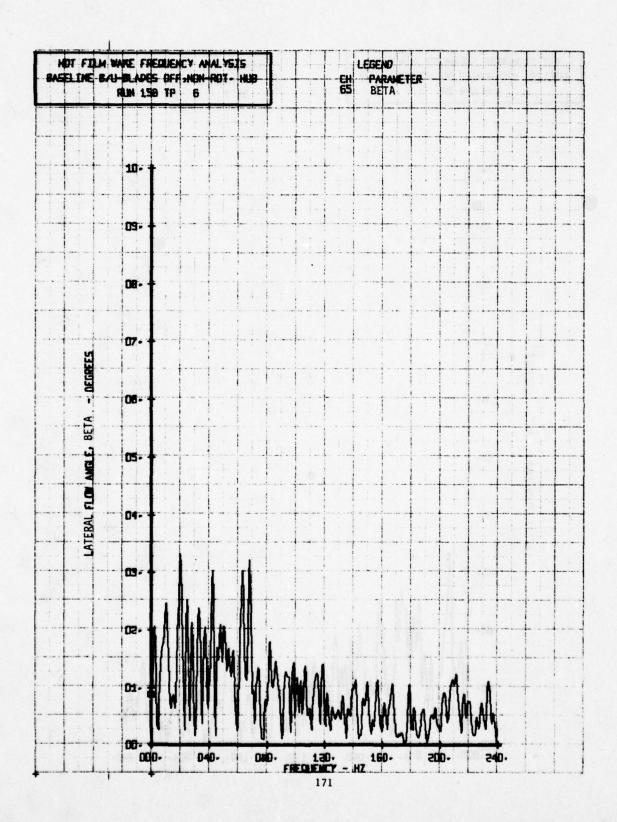


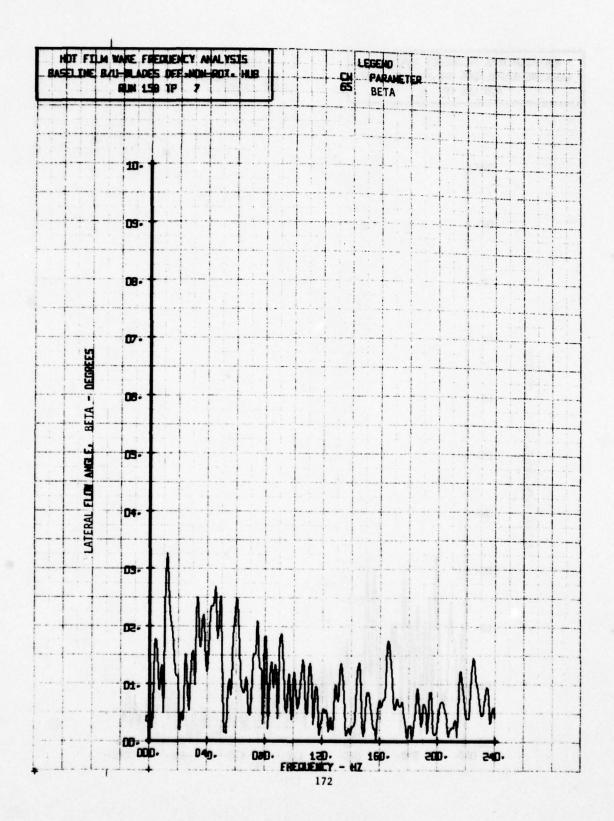


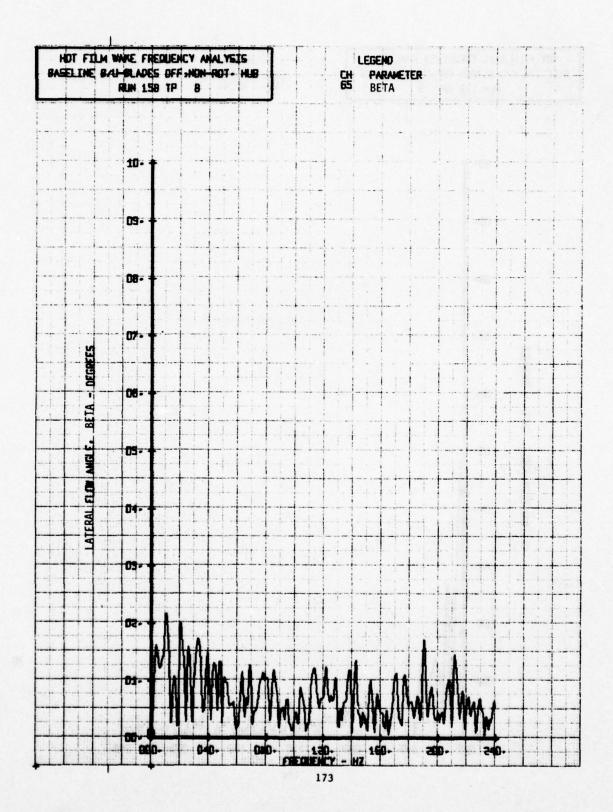


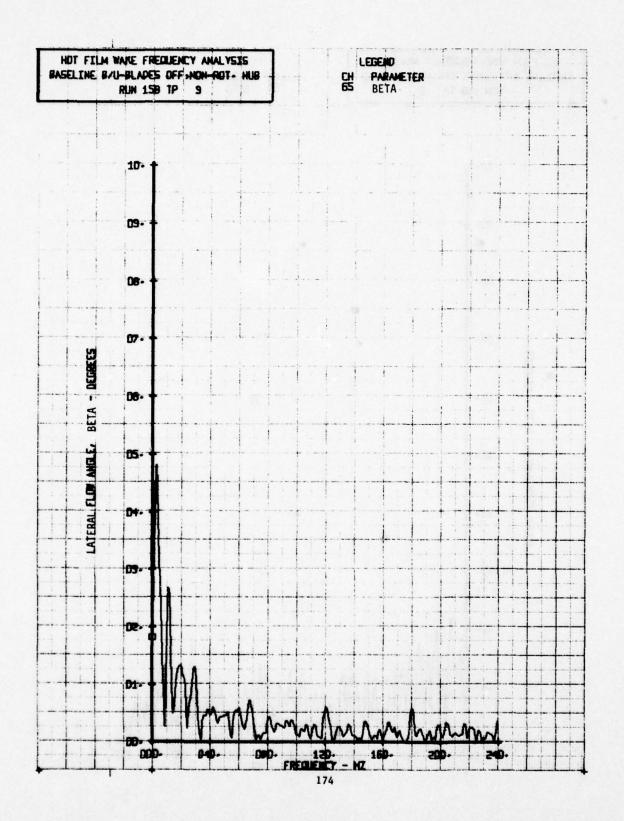




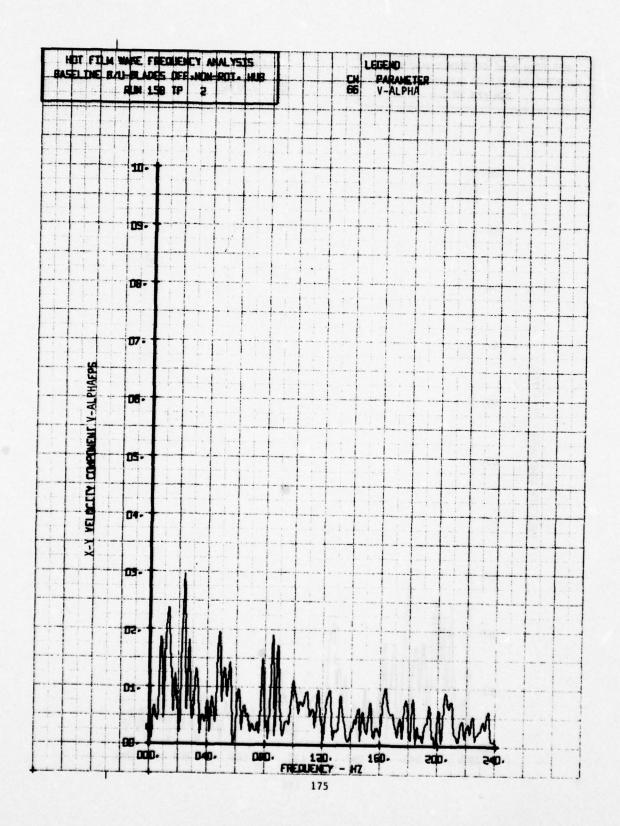


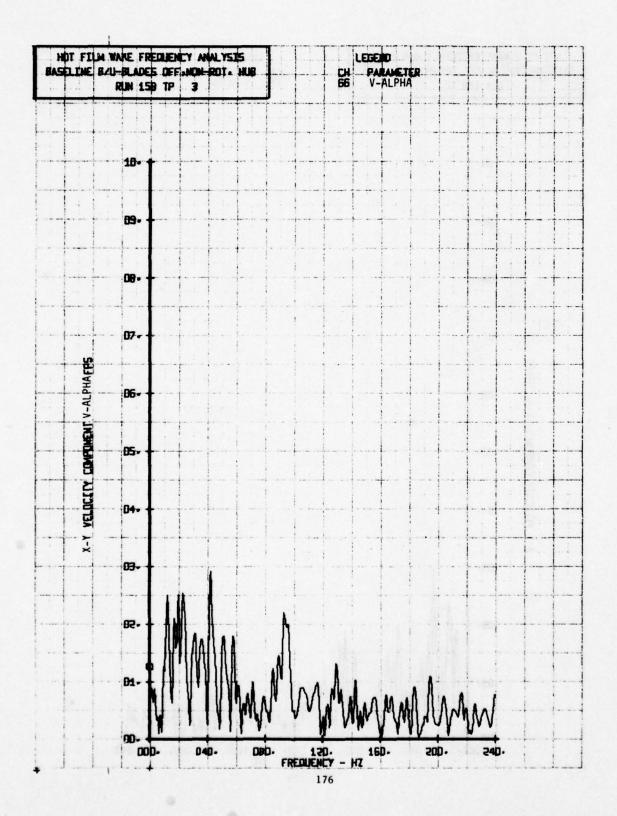


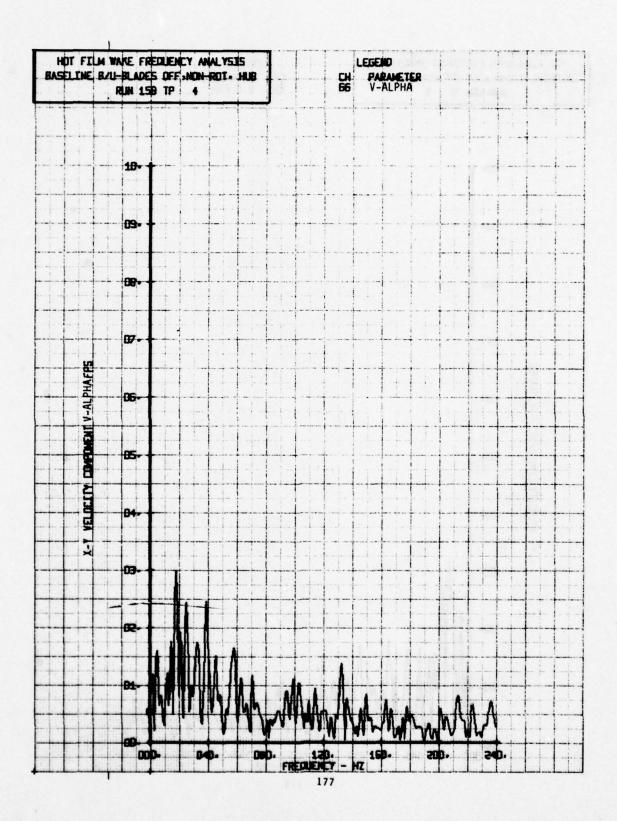


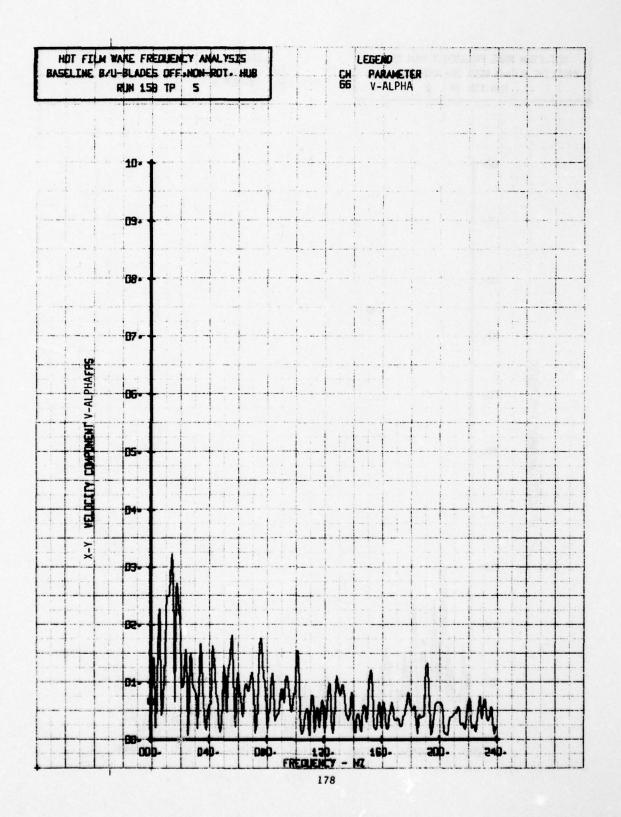


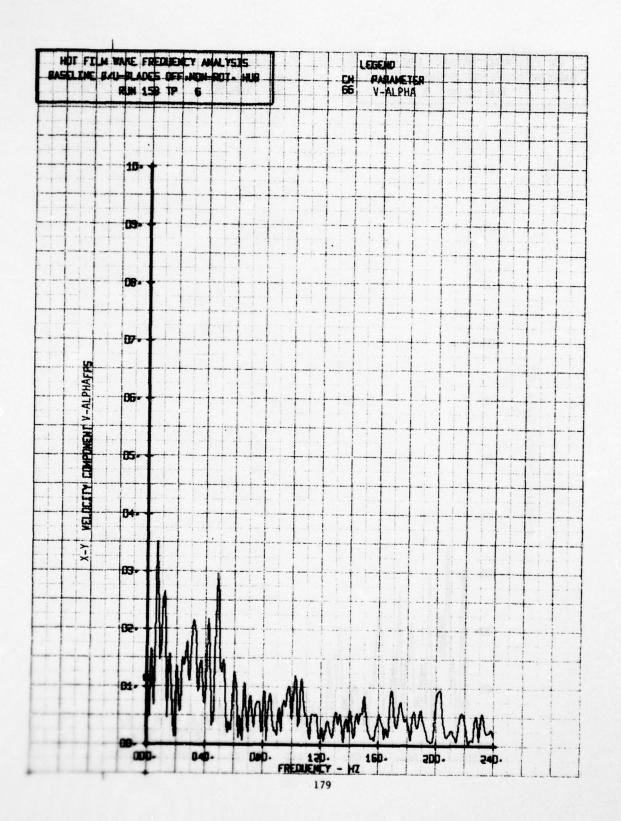
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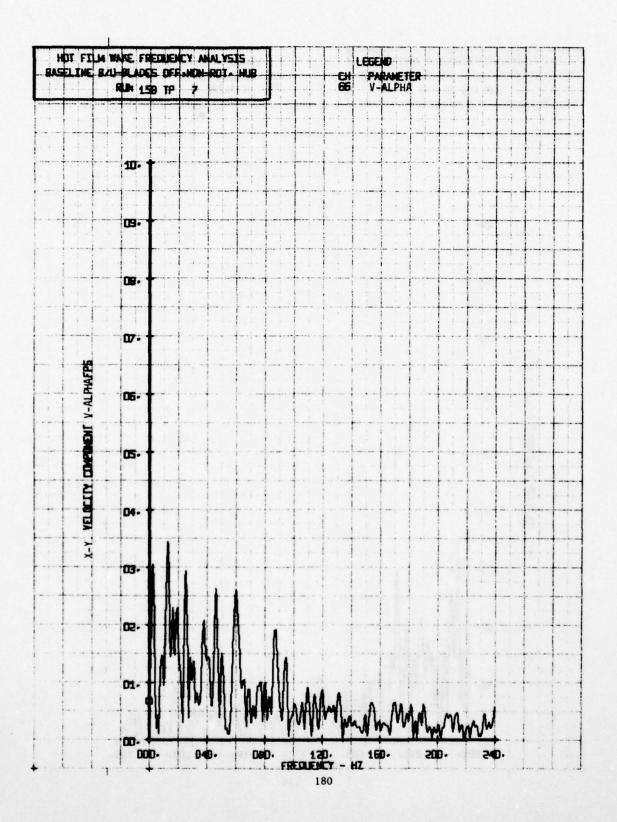


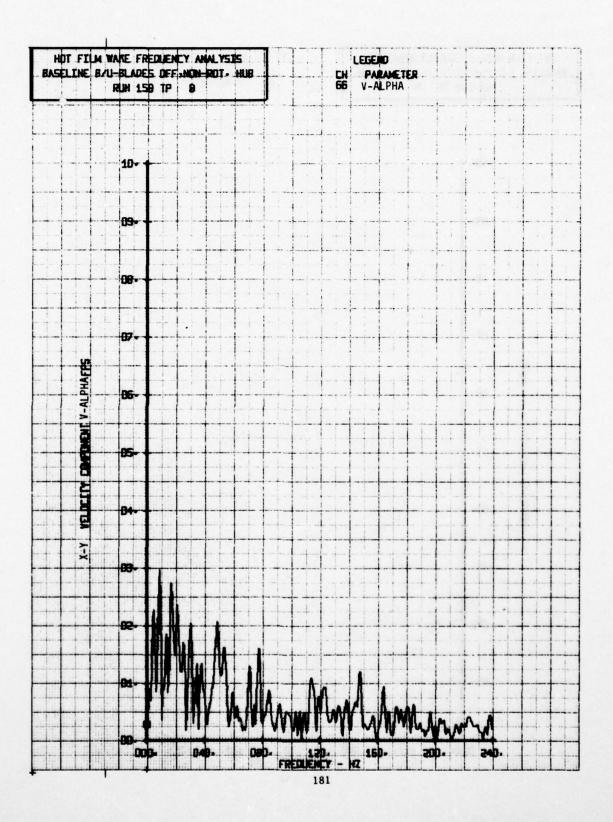


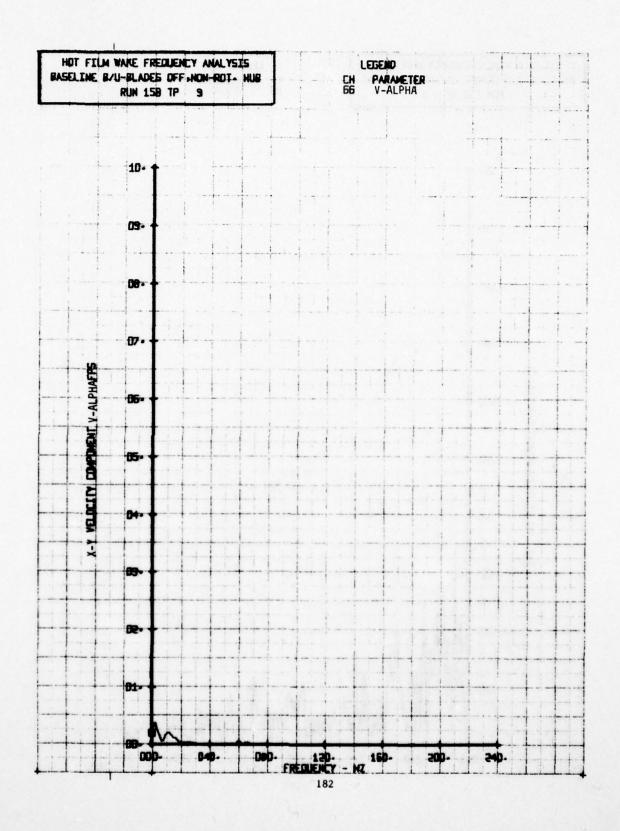


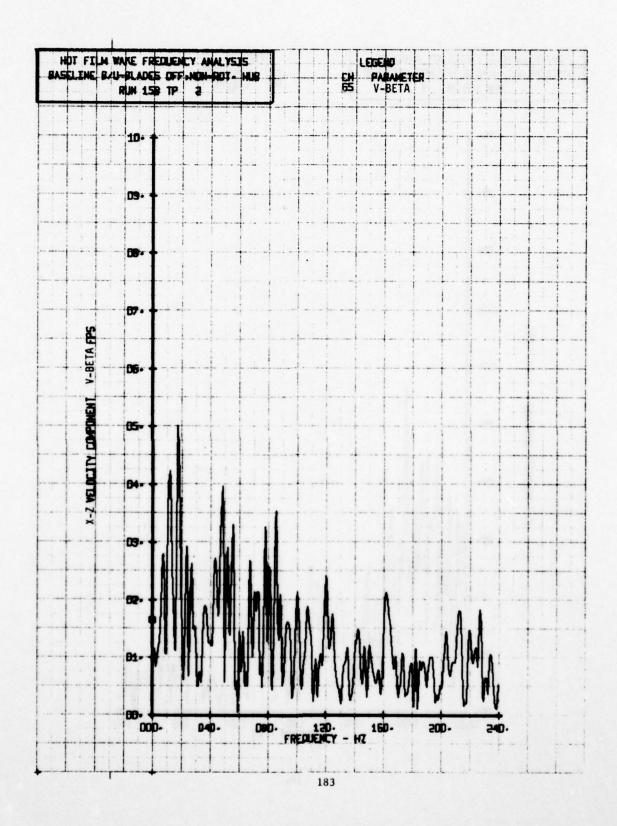


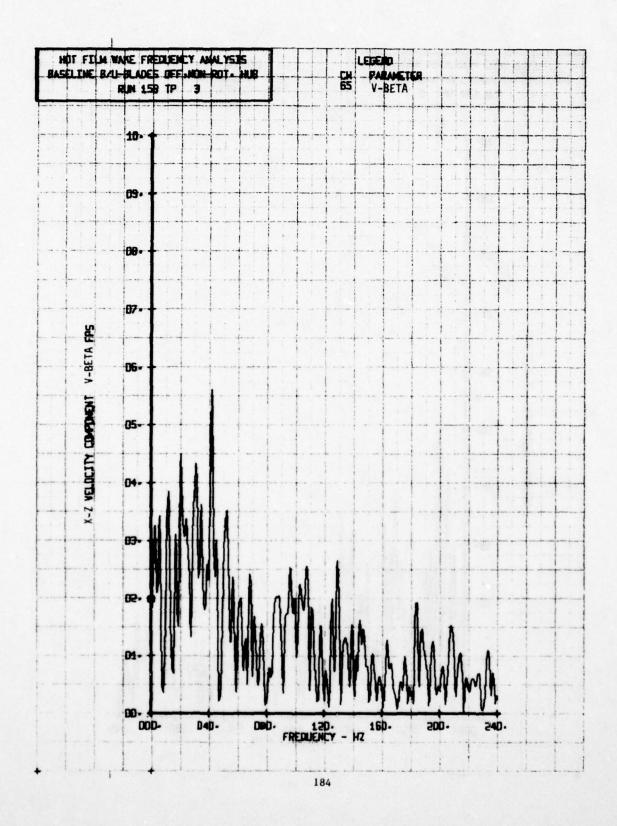


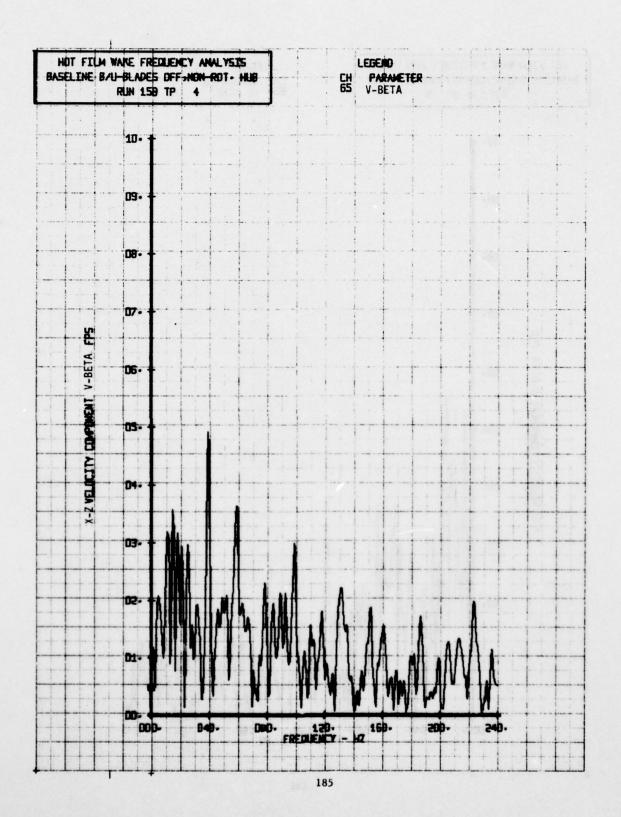


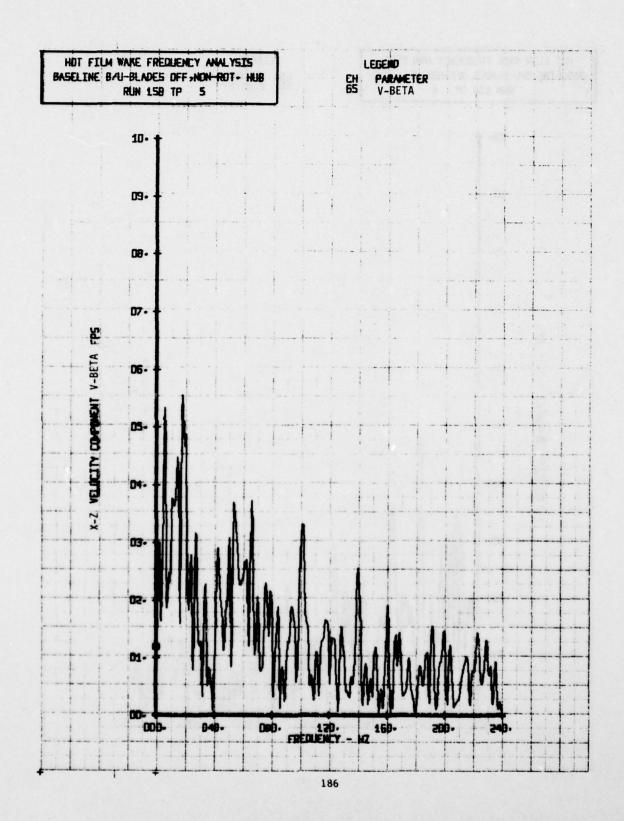


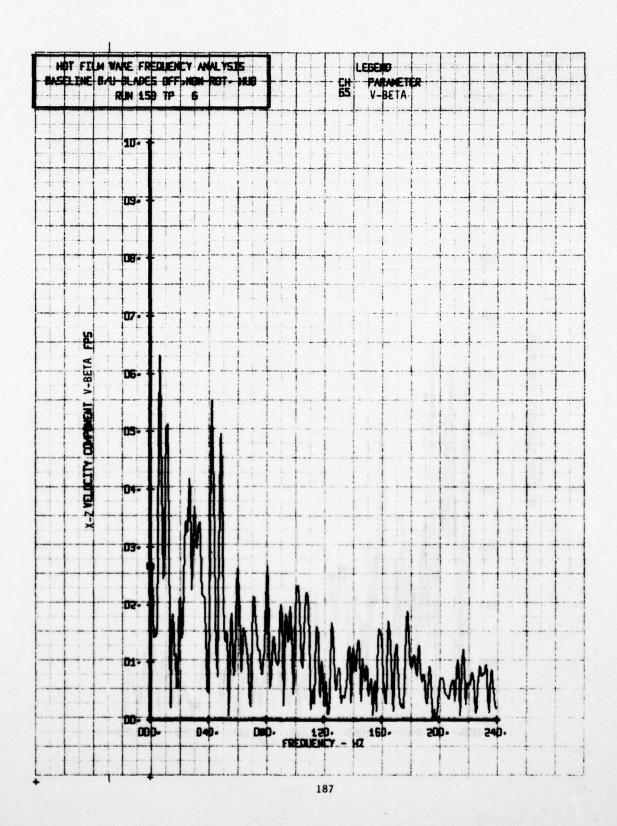


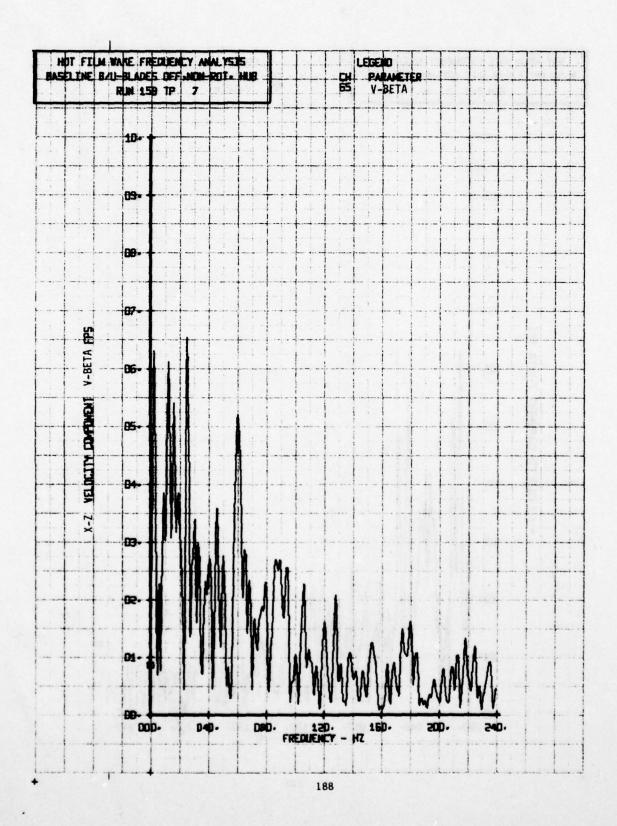


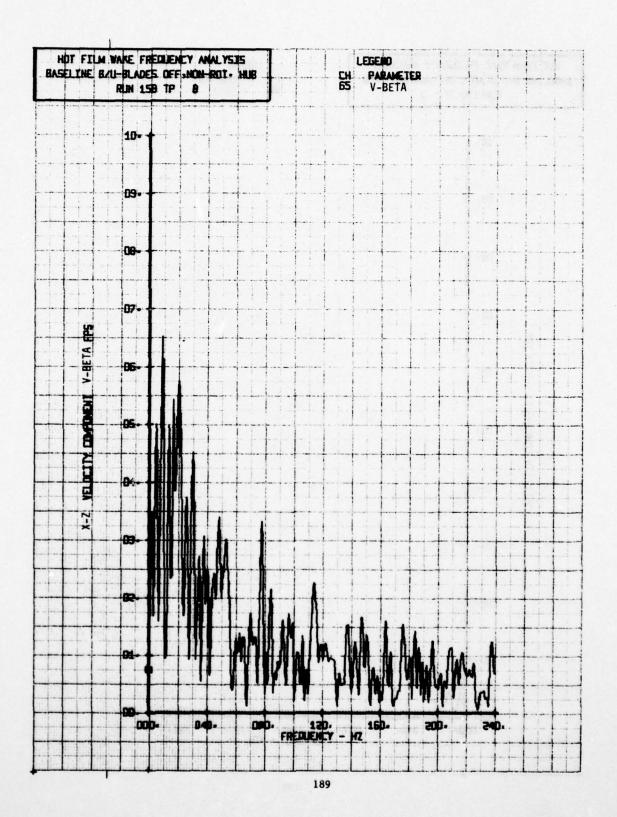


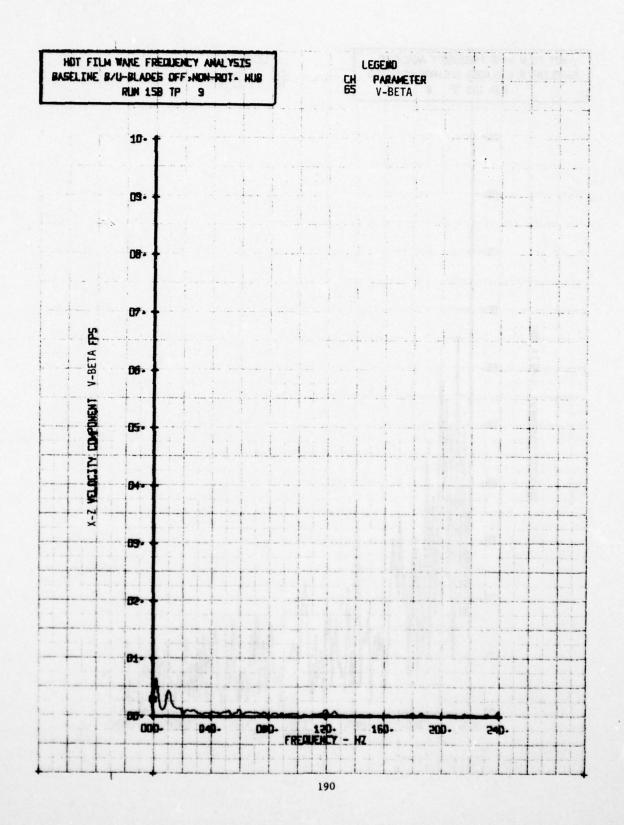


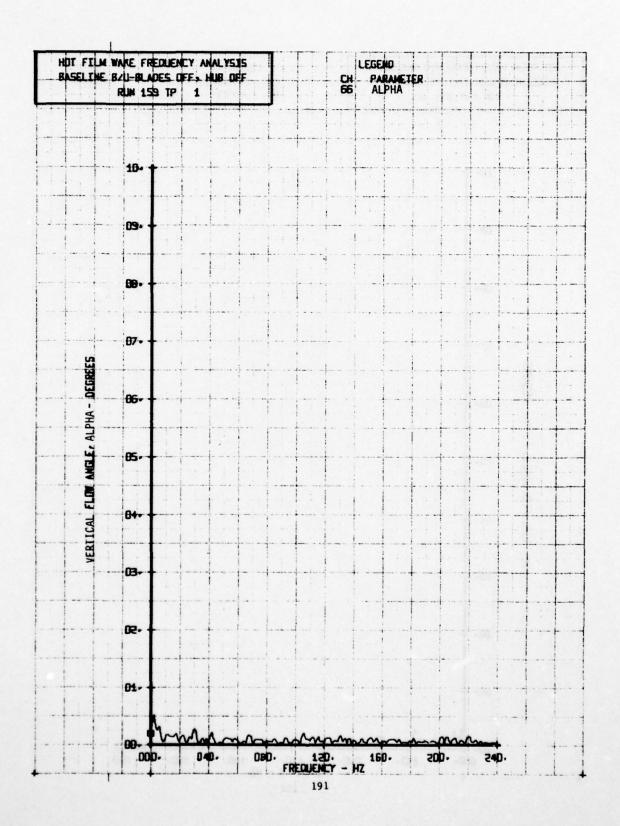


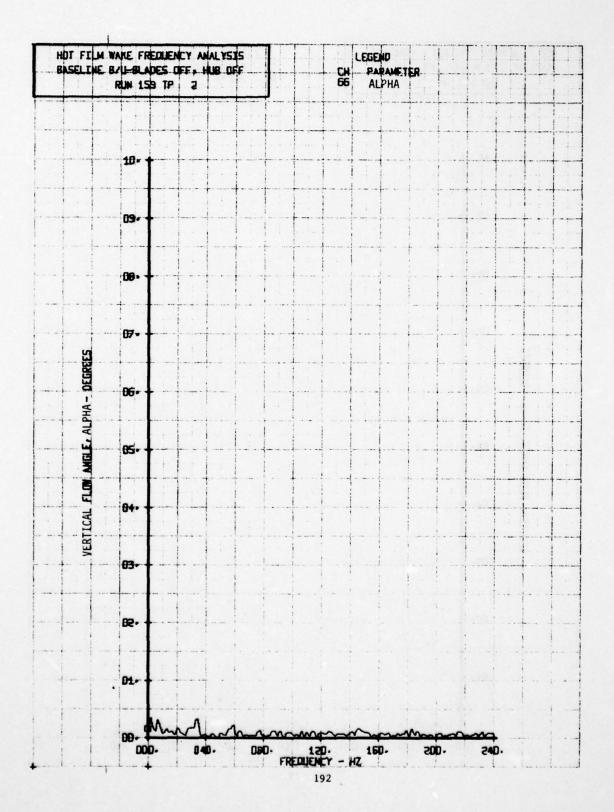


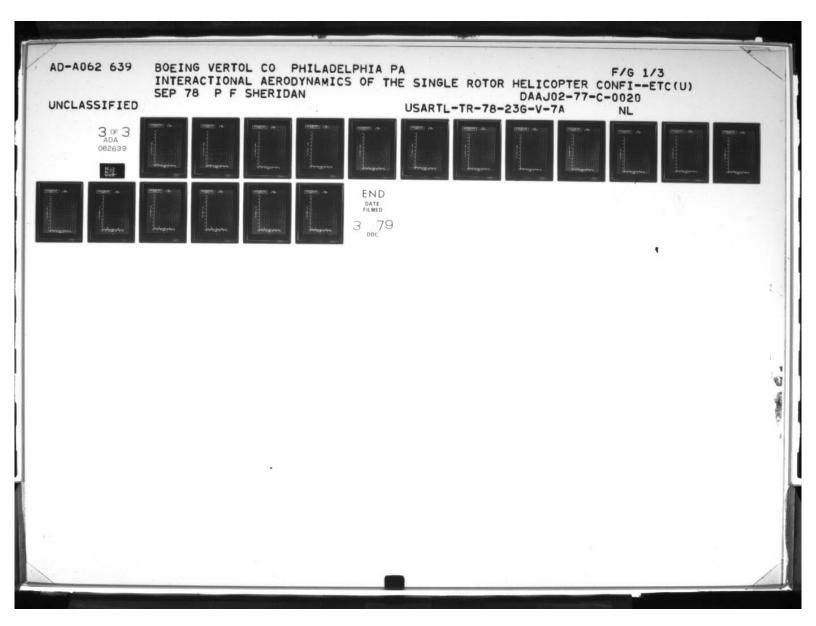


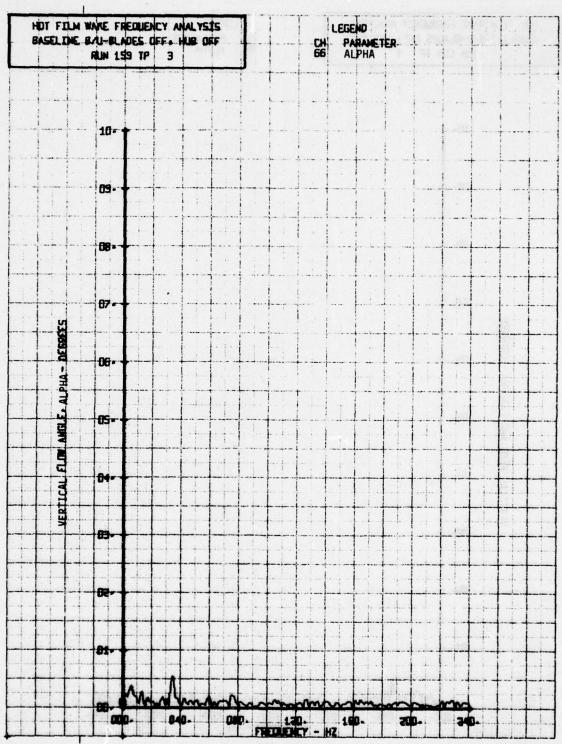


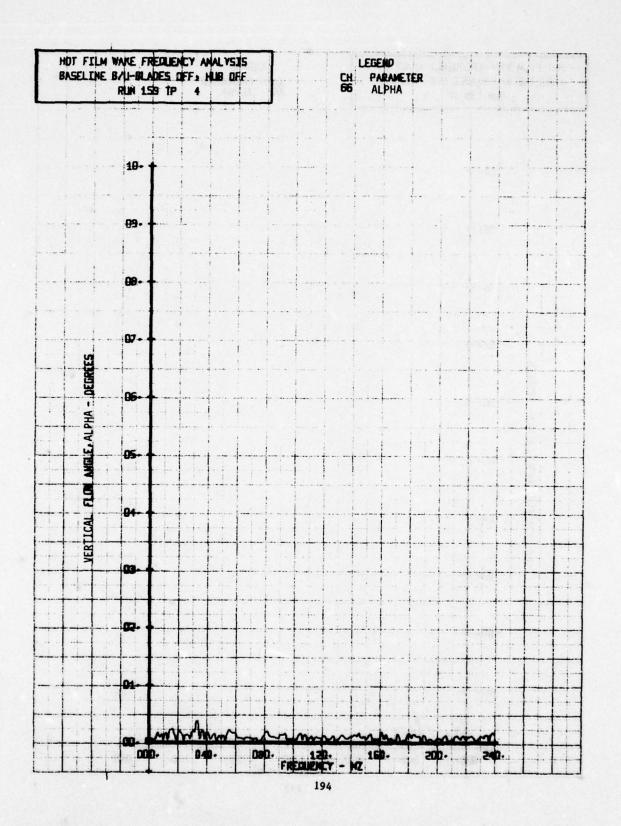


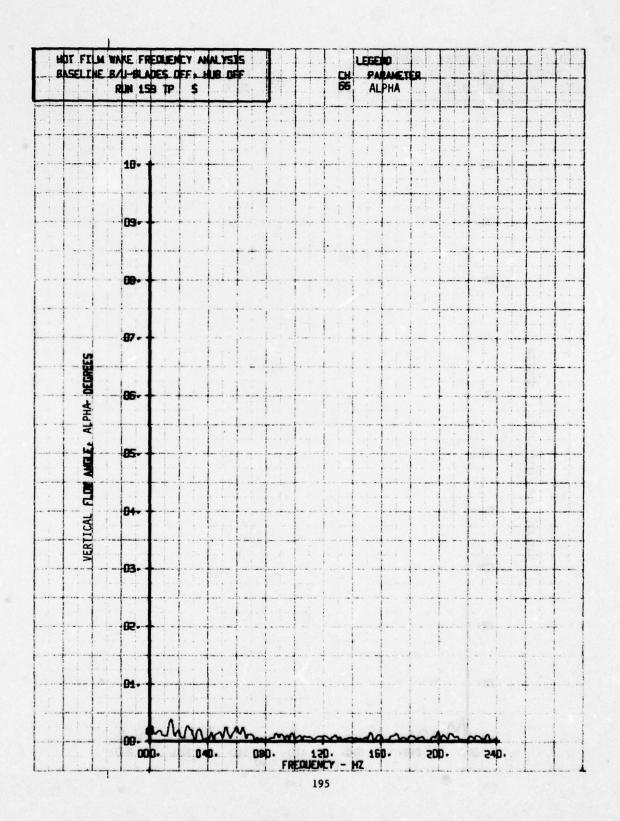


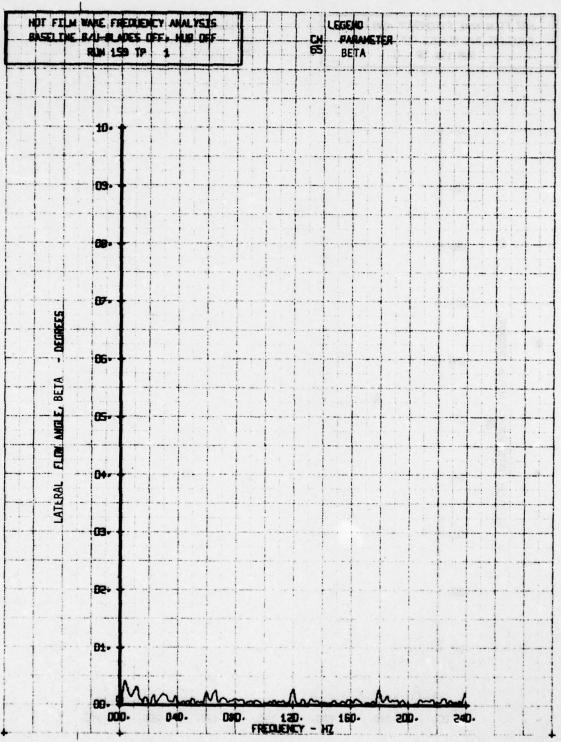


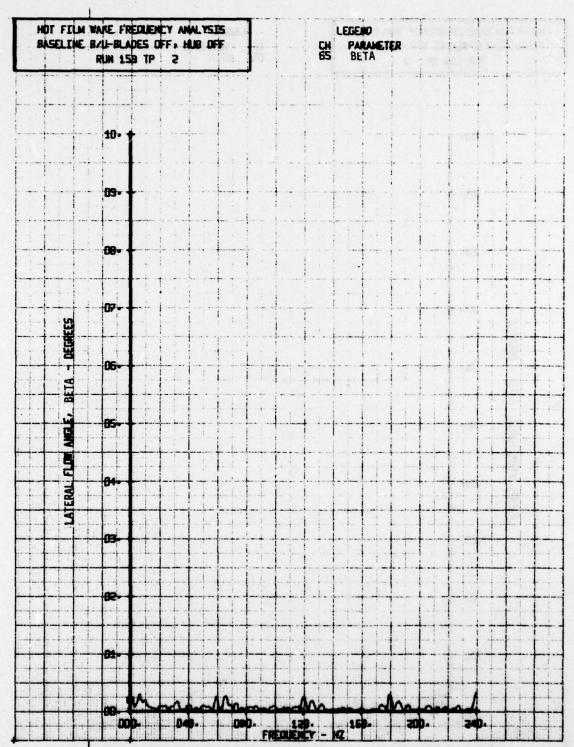


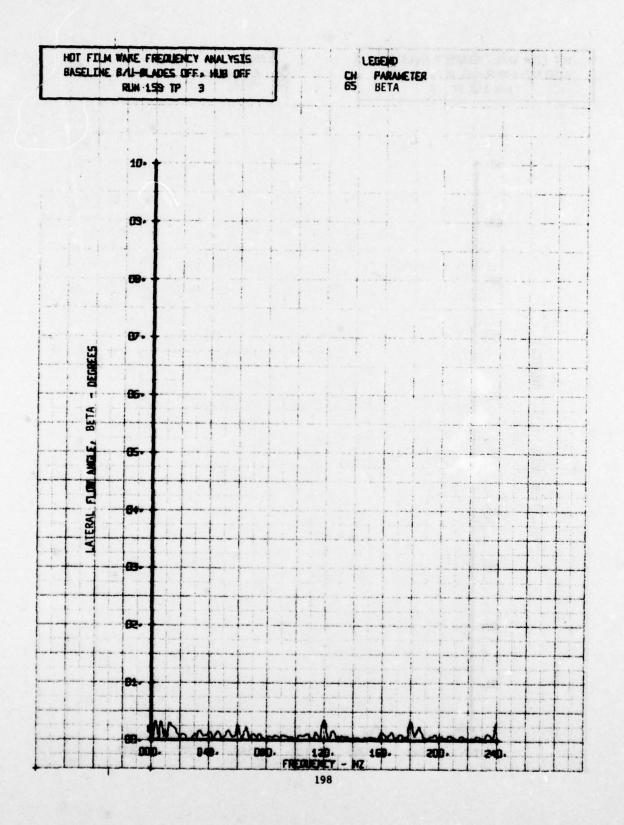




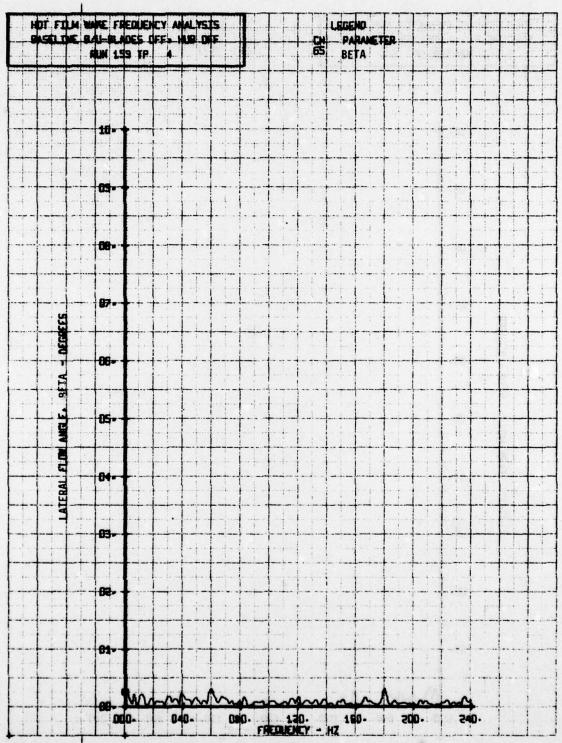


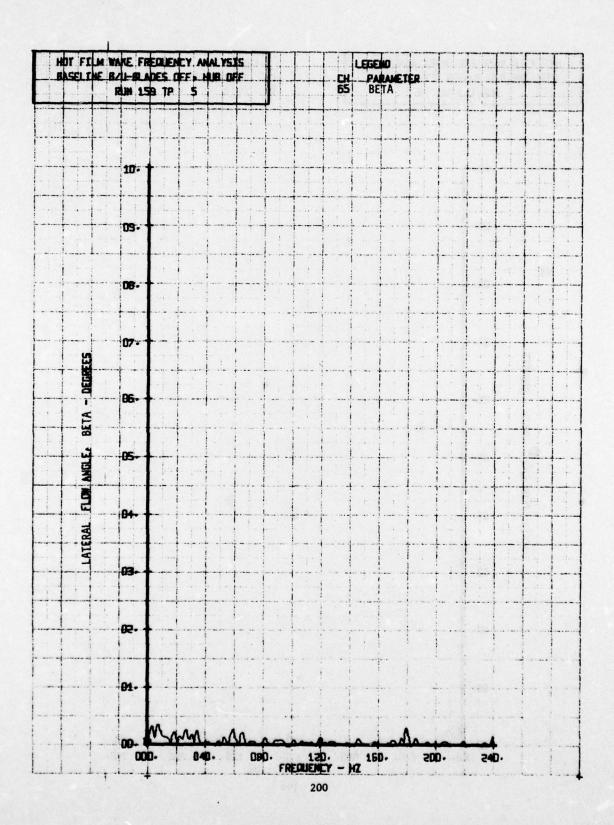






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